



Lessons Learned and Challenges from ORION ESM

TRISMAC

Trilateral Safety and Mission
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Lessons Learned and Challenges from ORION ESM



2014:

- ESA awarded Airbus Defence and Space as prime contractor.
- Scope: Develop & produce the ORION European Service Module (ESM) for ORION spacecraft as part of the ARTEMIS programme.

Challenges & Lessons Learned experienced by Airbus in the last 10 years from a Product Assurance & Safety perspective.

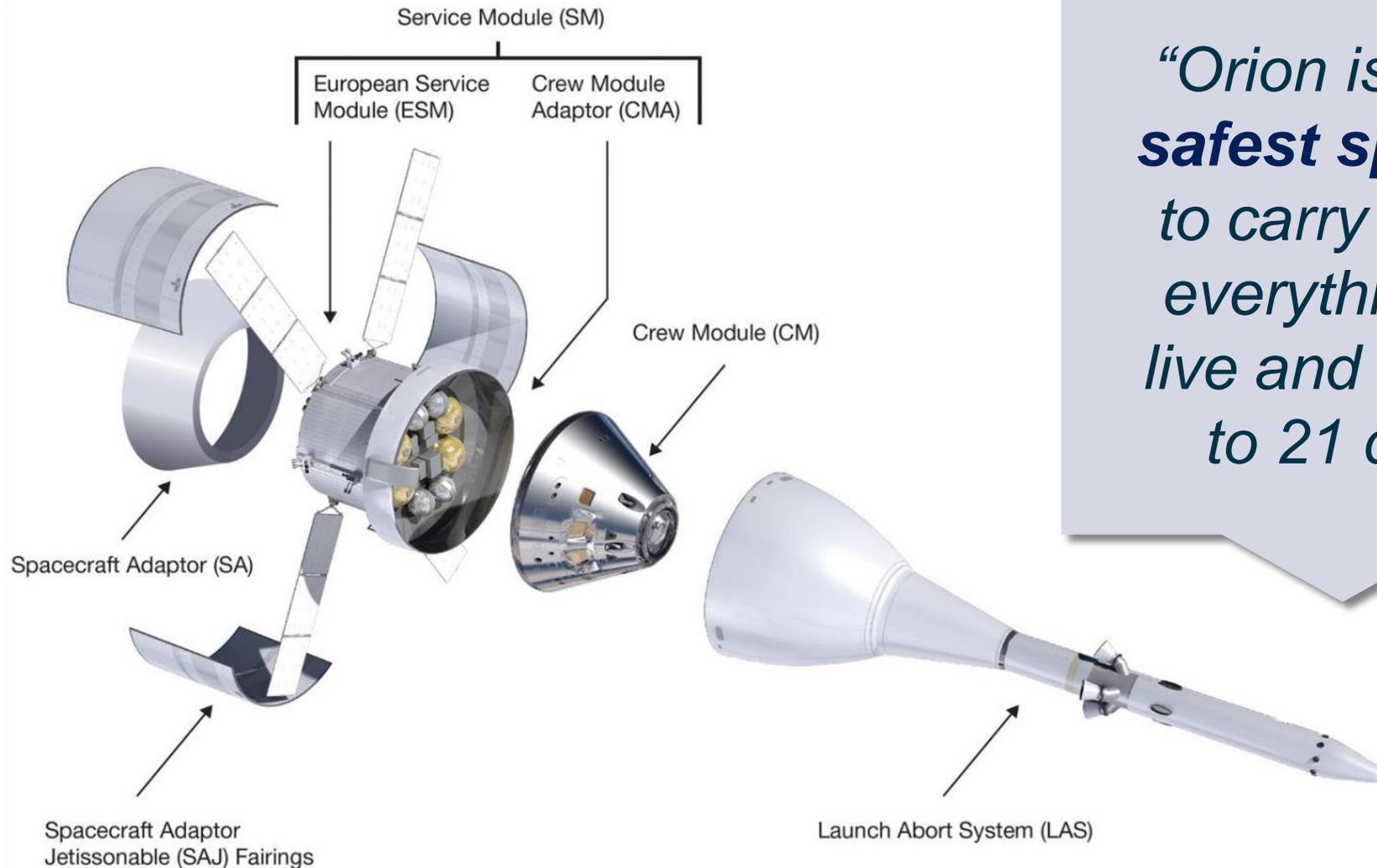
Challenges:

- Multi-purpose mission
- Industrial Setup
- Heritage of involved products & standards

Lessons learnt:

- Hazard Minimisation: DFMR vs. FT
- Design Changes
- Complex Programme Structure; Organisation, Tools and Soft Skills

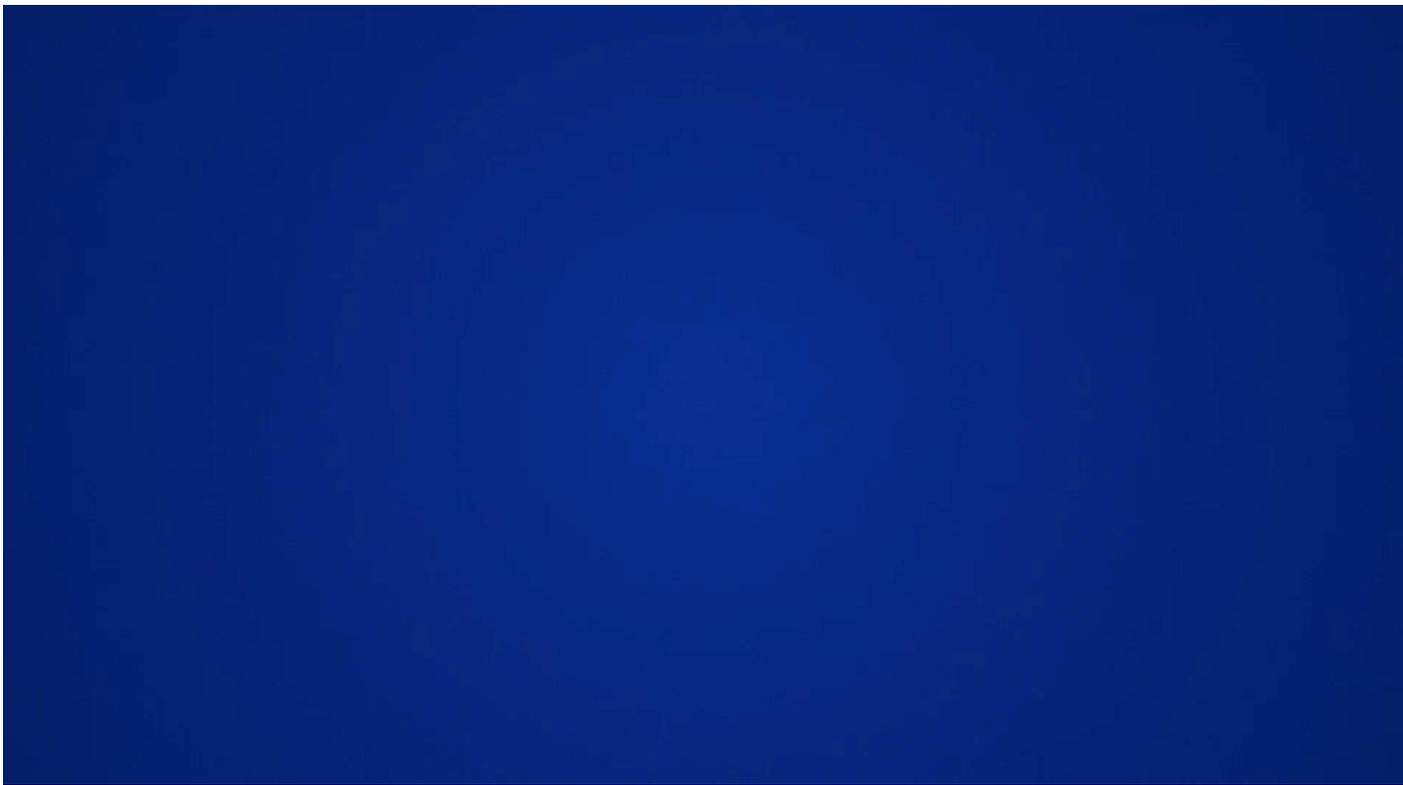
Challenge: Multi-purpose mission ORION Product Assurance & Safety challenge



*“Orion is designed to be the **safest spacecraft ever built** to carry four astronauts and everything they will need to live and work in space for up to 21 days per mission.”*

Challenge: The ESM Industrial Setup

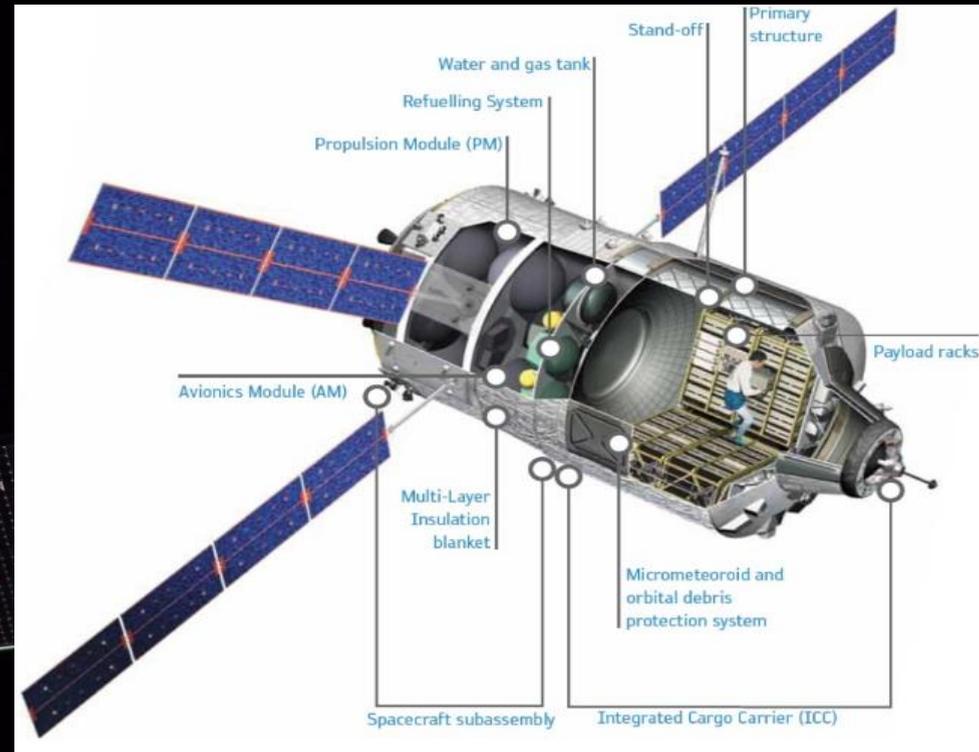
Quality Management & Organisational challenge



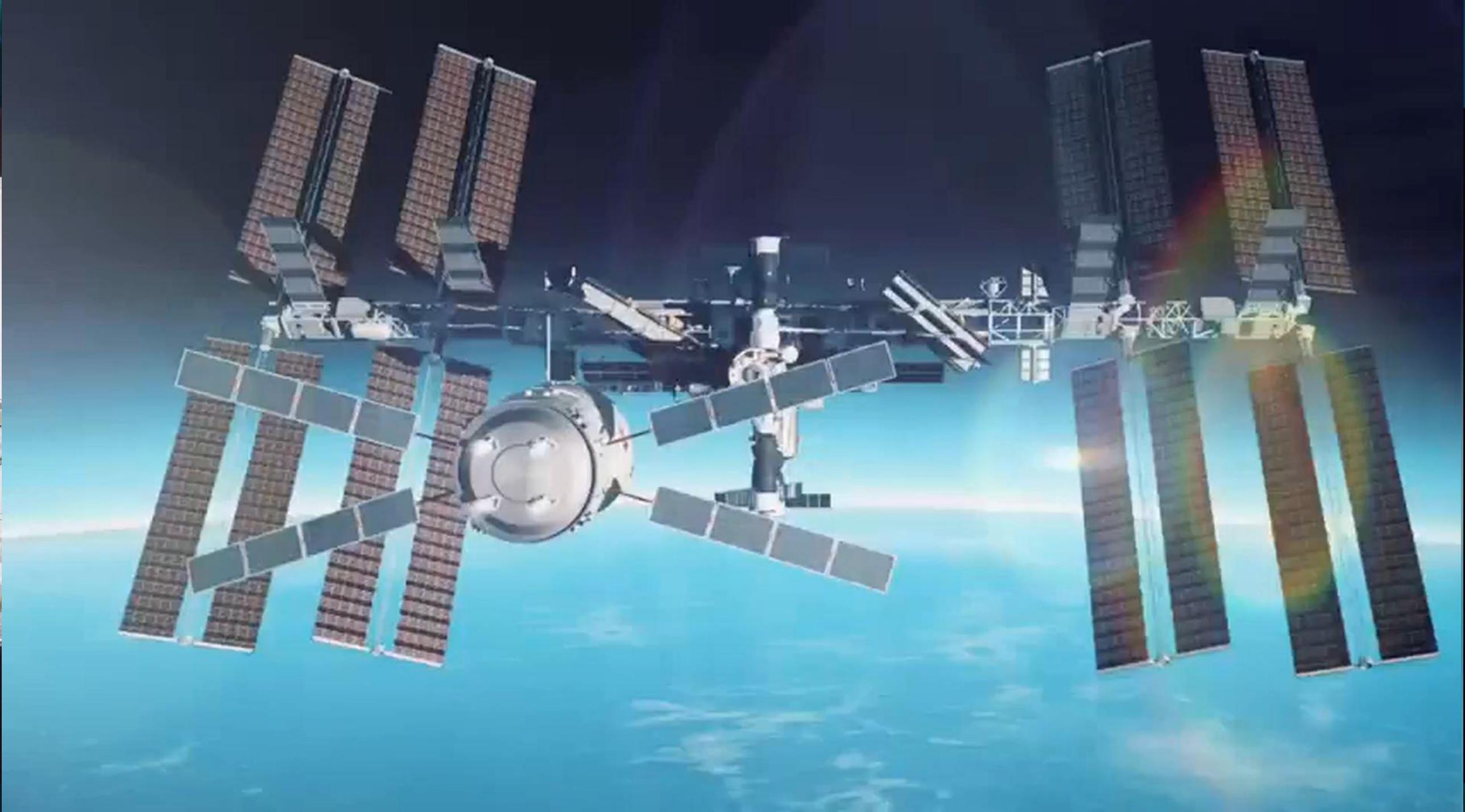
| | | |
|--|--|--|
| Airbus Defence and Space GmbH | | Project management, system engineering, AIT & ground operations management |
| Airbus Defence and Space SAS | | System engineering |
| Airbus Defence and Space Netherlands B.V. | | Solar Array Wings (SAW) |
| Airbus DS Space Systems Inc. | | U.S. interface |
| CompuCrisas, Crisas Ingeniería, S.A.U. (CRISA) | | Thermal Control Unit (TCU) |
| Tesat-Spacecom GmbH & Co. KG | | EEE for PDE |
| Airbus Group SAS | | Engineering support, ESM-QF, software, pressurant tanks |
| Airbus Group S.p.A. | | Engineering support, AIT, PCA, RCS, Fill & Drain Valve, Propellant Management Device, Propellant Management Device, PSS tubing |
| Thales Alenia Space Ltd. | | Structure, CSS subsystem, thermal control |
| Thales Alenia Space Belgium | | Pressure Regulator Unit (PRU) |
| Leonardo S.p.A. | | Power Conditioning and Distribution Unit (PCDU) |
| RUAG Schweiz AG | | Solar Array Drive Assembly (SADA) |
| Lockheed Martin Space | | Pyro Valves |
| Latécoère Interconnection Systems | | DC harness |
| Factor | | DC harness |
| Adcole Maryland AeroSpace | | S/C Sensor |
| VACCO Industries | | Auxiliary propellant filters |
| Clemessy S.A. | | EGSE maintenance |

- ~~Airbus contract under ESA contract~~
- ~~NASA is the end-customer~~
- ~~Airbus as prime contractor~~
- ~~System engineering split between Bremen and Les Mureaux~~
- ~~Lockheed Martin is responsible for the Crew Module and the end-to-end final integration~~
- ~~Prime contractors selection must fulfill ESA return requirements~~

Challenge: The ESM Heritage Programme ATV (Automated Transfer Vehicle)

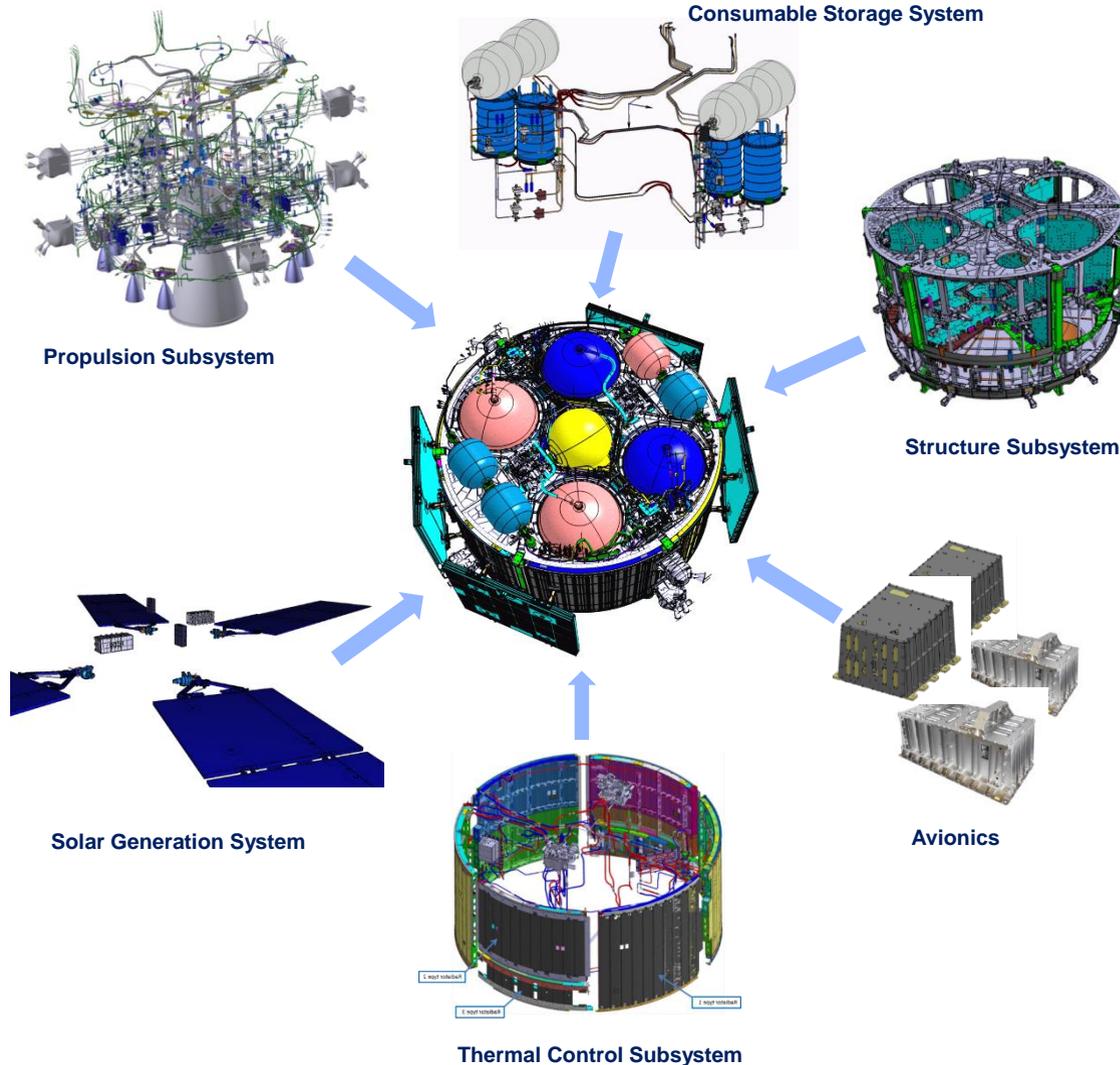


- Developed by Airbus / ESA 1995 – 2007
- 5 vehicles flown to ISS between 2008 & 2014
- 20 ton cargo (8 ton payload) vehicle to supply the ISS
 - Gas, water, refueling, propulsive support, payloads
- Autonomous ISS docking capability
- Designed against
 - ECSS tailoring
 - Compatibility with Ariane 5



The ESM Design

Various General Challenges compared to heritage



- “Multi-Purpose” vehicle
 - Lunar flyby mission / Crewed lunar rendezvous / ISS supply missions
- Industrial set-up
 - ESA / NASA / Lockheed Martin
 - ESA geo-return
- Shuttle heritage equipment
 - OMS-E and TVC
- Requirements changing during the project
 - New launcher upper stage, varying mission scenarios, NASA developing vehicle operations
- European versus U.S. standards
- Very tight budget & schedule

Lessons Learned: Hazard minimisation Design For Minimum Risk (DFMR) vs. Failure Tolerance (FT)



Late agreements on philosophy & requirements for minimising hazards



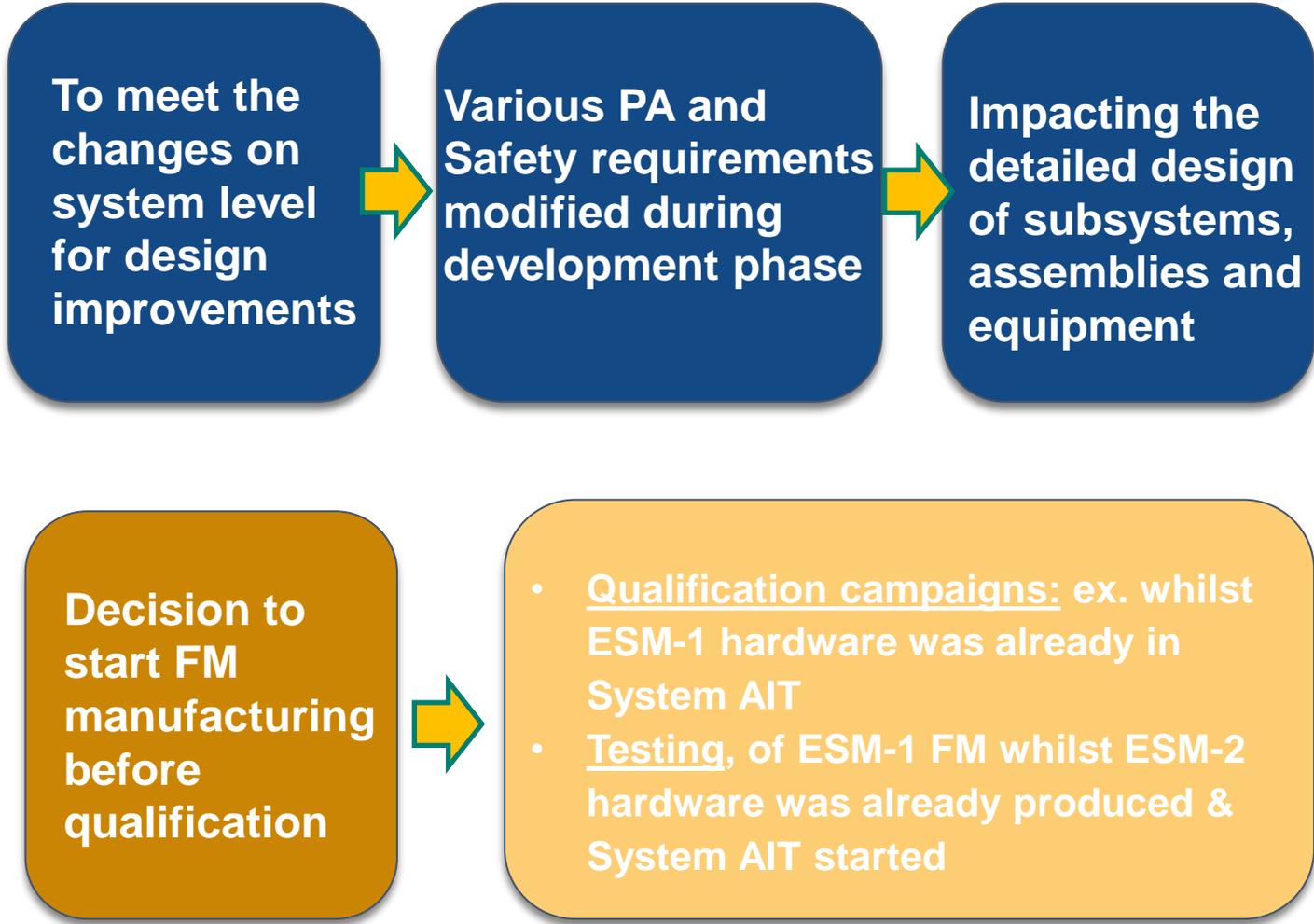
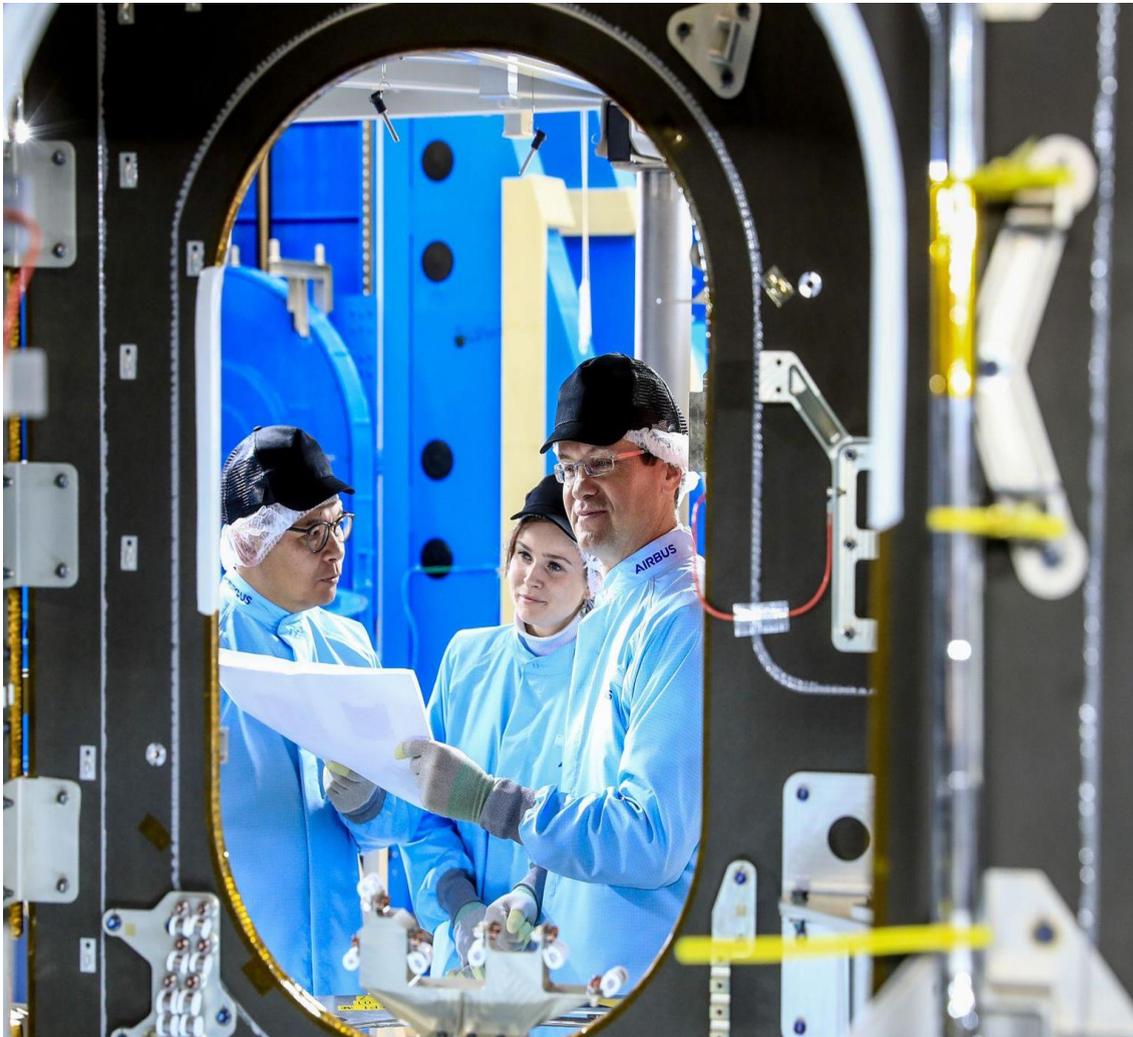
- DFMR only accepted for certain predefined items:**
- ESM Primary structure
 - Structural failure of pressure vessel walls
 - Structural failure of pressurised lines & welded joints

- Items declared “0-FT treated as DFMR”:**
- Secondary structure & Cold plates
 - Radiator & SAW hinges
 - tank bearings

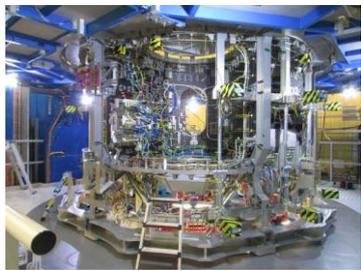
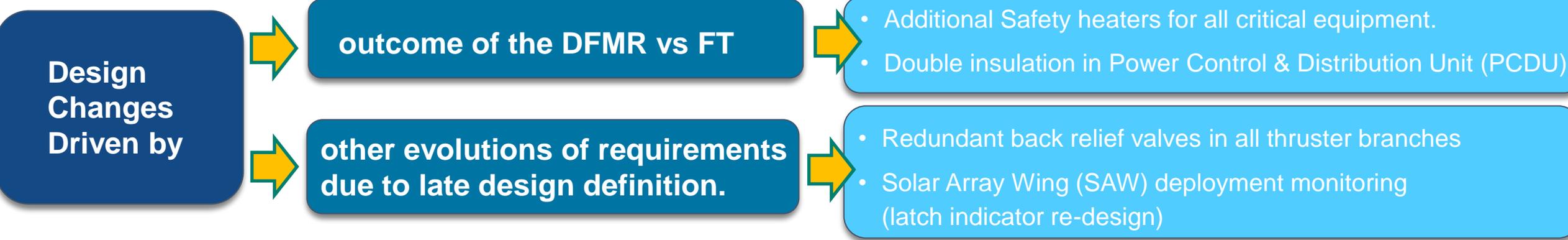
- Items not accepted as DFMR:**
- Single seals & valve bellows
 - Pressure Transducer diaphragm & SADM mechanism
 - SAW panel hinges and release & deployment mechanisms

- **Zero-FT exceptions with significant verification effort**
Additional qualification & acceptance requirements for pressurised valve bellows
- **Partial re-design of heritage equipment**
Secondary containment for valves & pressure transducers

Lessons Learned: Design Changes



Lessons Learned: Design Changes



Bremen, G
Airbus
System and AIT



White Sands, USA
NASA
PQM hot firing tests



Plum Brook Station, USA
NASA
Environmental and structural test



Les Mureaux, F
Airbus / AGG
Functional V&V



Denver, USA
LMCO
ORION S/C Simulation



KSC, USA
NASA
ORION S/C Integration

Lessons Learned Complex Programme Structure



Organisation, Tools and Soft skills



The complex industrial setup leads to a high number of stakeholders

- Programme is largely managed **online** with only few physical meetings (such as QPM's)
- **Powerful tools** for online meetings needed (certified for export control requirements)
- Attention: There are different **IT security requirements** between Europe and the U.S.
- **Distribution of information**, including **export control** compliant data management.
- Dedicated certified software platforms and secure servers needed.

→ **Investment in IT infrastructure and Tools, including personal equipment of the team members**

For standard PA&S processes, high number of stakeholders

- Number of NRB participants >50 persons online.
- Demanding PA Manager role as Chairperson.

→ **Training of the PA&S team members on Softskills and Communication**

Summary Lessons Learned and Challenges from ORION ESM



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THANK YOU

