

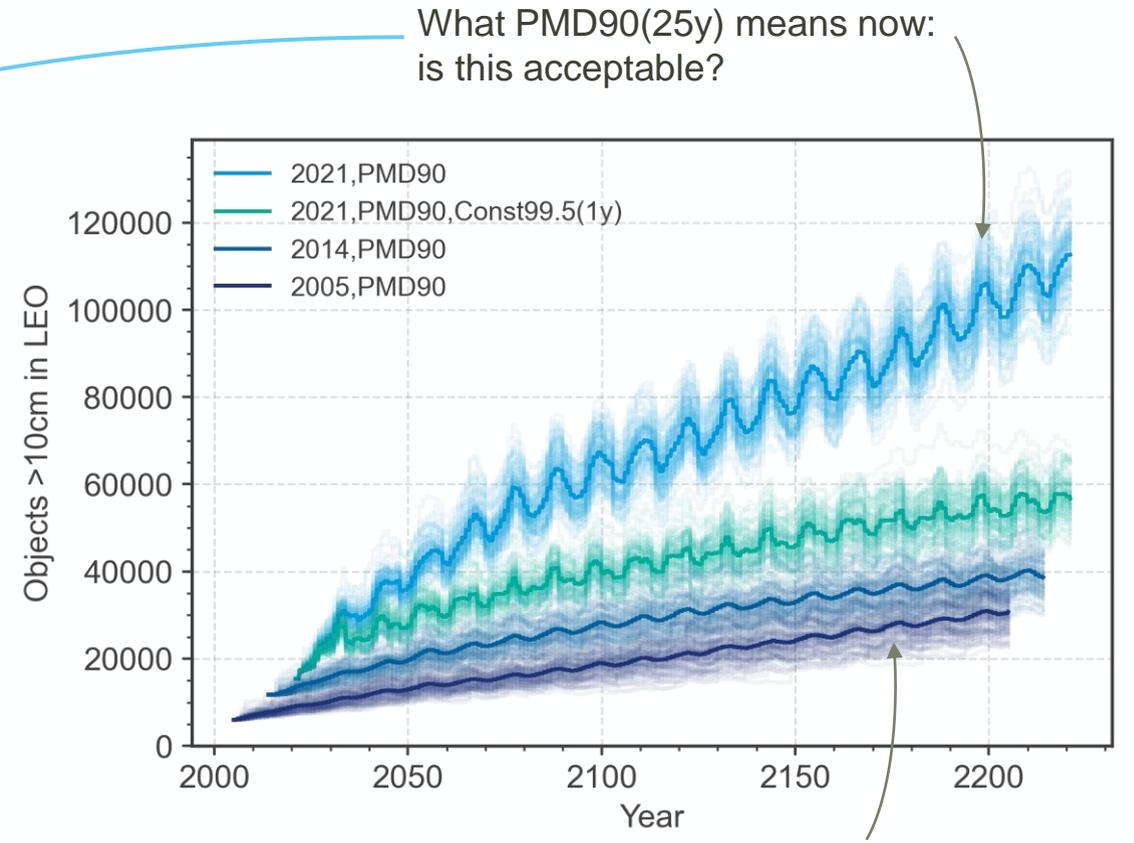
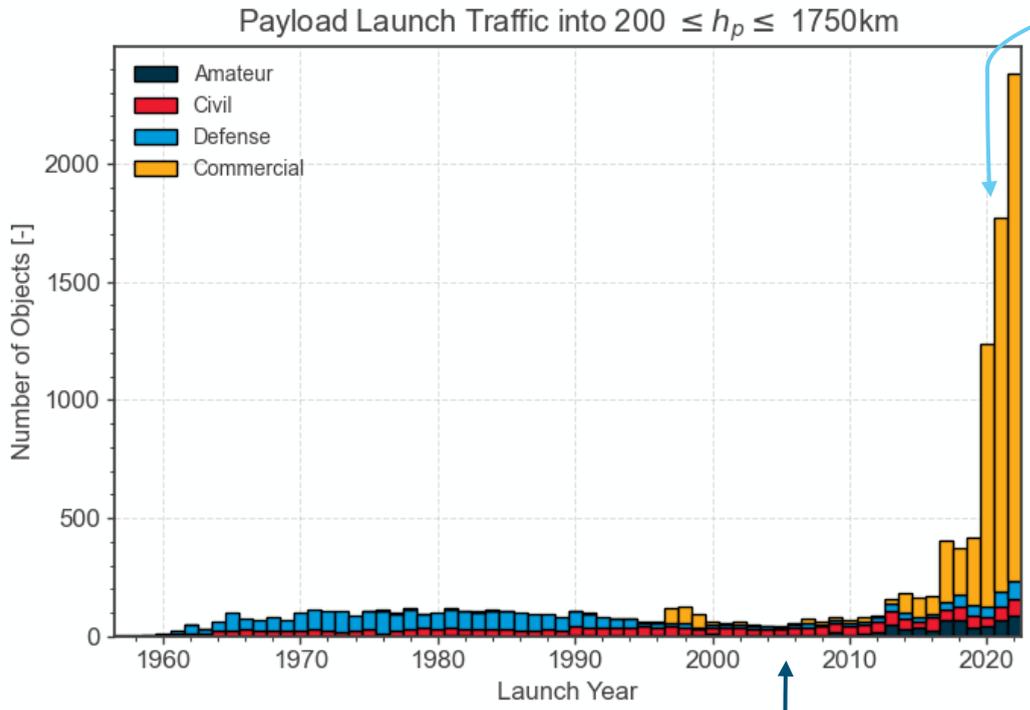
# ESA's Zero Debris Approach: Policy and Requirements

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Paloma Villar

24/06/2024

# Why do we need a new approach to mitigation?



What PMD90(25y) meant when IADC drafted their recommendation

$h_p$ : perigee altitude | PMD: Post-Mission Disposal



# Zero Debris Approach development

“*In ESA we are implementing a policy that, by 2030, we have a ‘**net zero pollution**’ strategy for objects in space, by consistently and reliably removing them from valuable orbits around Earth immediately after they cease operations. We need to **lead** by example here.*”

Josef Aschbacher, ESA Director General

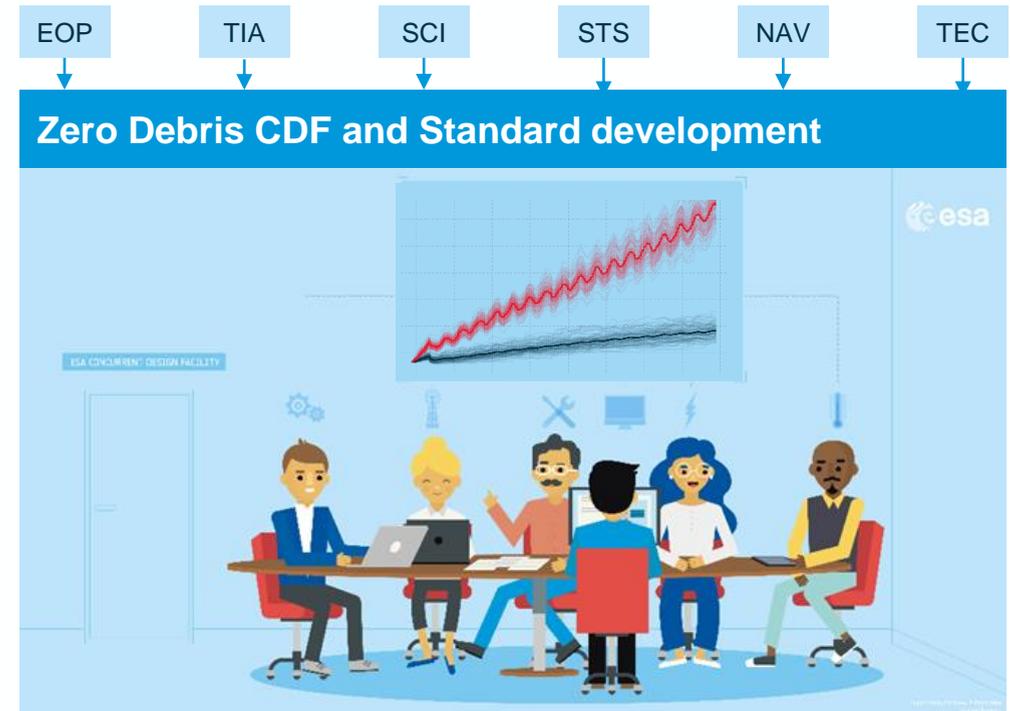


## Policy update recommendations

considering environmental needs and impact on future missions, informed by an extensive **simulation campaign**



**Roadmap** for technical developments & standards, providing an estimation of the **resources** needed and a **phase-in schedule**



Developing ESA Zero Debris approach

Engaging partners, building a community

## ESA SDM Policy & Standard



Technical requirements for ESA missions and contributions

## ESA Technical Developments



ESA support to industry's transition and compliance to SDM standards

## Zero Debris Technology Booklet



Crowd-sourced technical solutions to reach Zero Debris targets by 2030

## Zero Debris Charter



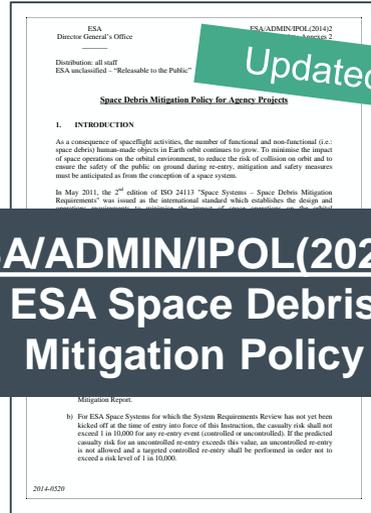
Jointly defined principles and targets for long term space sustainability



# ESA Space Debris Mitigation Regulation status



- Policy
- Standard
- Handbook



Updated in 2023

**ESA/ADMIN/IPOL(2023)1**  
**ESA Space Debris Mitigation Policy**



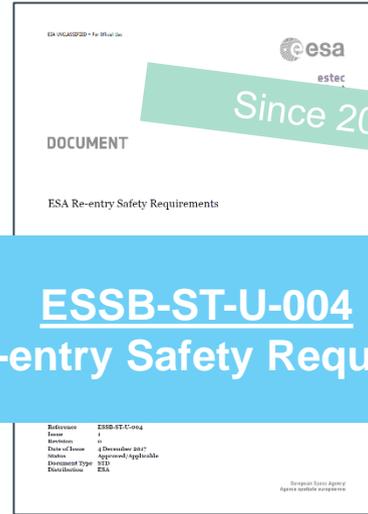
Born in 2023

**ESSB-ST-U-007**  
**ESA Space Debris Mitigation Requirements**



Updated in 2024

**ECSS-U-AS-10C**  
**Space sustainability - Adoption Notice of ISO 24113**



Since 2017

**ESSB-ST-U-004**  
**ESA Re-entry Safety Requirements**



Update in 2024

**ESSB-HB-U-002**  
**ESA Space Debris Mitigation Compliance Verification Guidelines**



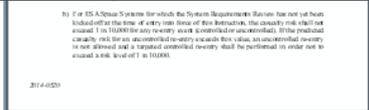
# ESA's Space Debris Mitigation Policy



Who?



**ESA/ADMIN/IPOL(2023)1**  
**ESA Space Debris Mitigation Policy**



Definition of the perimeter of applicability

ESA space systems, operations under ESA's responsibility, contribution to international activities, procurement of launch services



Applicable to all missions regardless of their phase

It provides recommendations in case of Mission Extension Reviews, anomalies affecting space debris mitigation measures, and requests for deviations/waivers



Introduction of the Space Debris Mitigation Assessment Board

<https://technology.esa.int/upload/media/ESA-ADMIN-IPOL-2023-1-Space-Debris-Mitigation-Policy-Final.pdf>

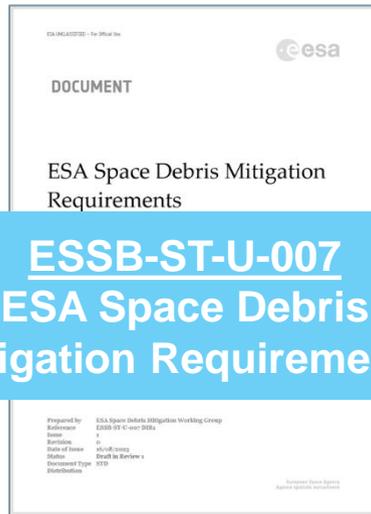


# ESA's Space Debris Mitigation Requirements



What's new

What?



## ESSB-ST-U-007 ESA Space Debris Mitigation Requirements



### Clearance criteria

- + **5 years in LEO**
- + Collision probability threshold
- + Apogee below 375 km for constellations
- + If graveyard, no crossing with known constellations



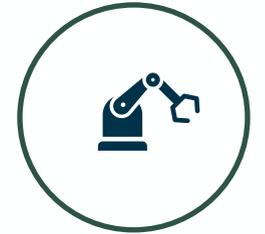
### Probability of successful disposal

- +  $\geq 90\%$  considering both **internal** (reliability) and **external** (impacts) factors
- +  $\geq 95\%$  for large constellations
- + Monitoring and reassessment



### COLA & STM

- + Encoding of current best practices (e.g. data sharing)
- + Recurrent manoeuvre capability in GEO, in LEO for high and very high-risk objects, and for constellations
- + Collision probability threshold for action  $\leq 1:10000$



### Design for removal

- + Preparation for removal for high-risk objects in the protected regions

<https://technology.esa.int/upload/media/ESA-Space-Debris-Mitigation-Requirements-ESSB-ST-U-007-Issue1.pdf>

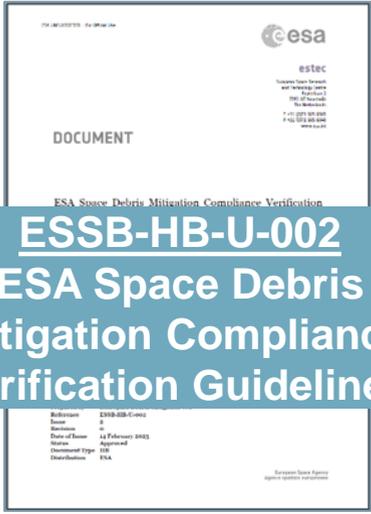
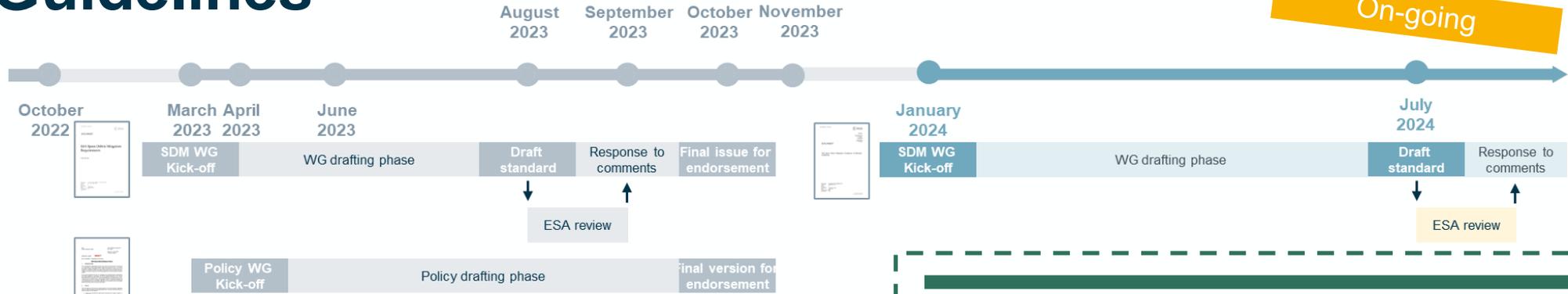
COLA: Collision Avoidance | STM: Space Traffic Management



# ESA's Space Debris Mitigation Compliance Verification Guidelines

On-going

How?



Guidelines on suitable methodologies for verification

Indication of what's expected at the different mission phases

Revision of ESA's available tool for compliance analysis

Update/coordination of/with related documents

**drama**

Update & planned releases to support analyses

Design-for-Demise, Close Proximity Operations

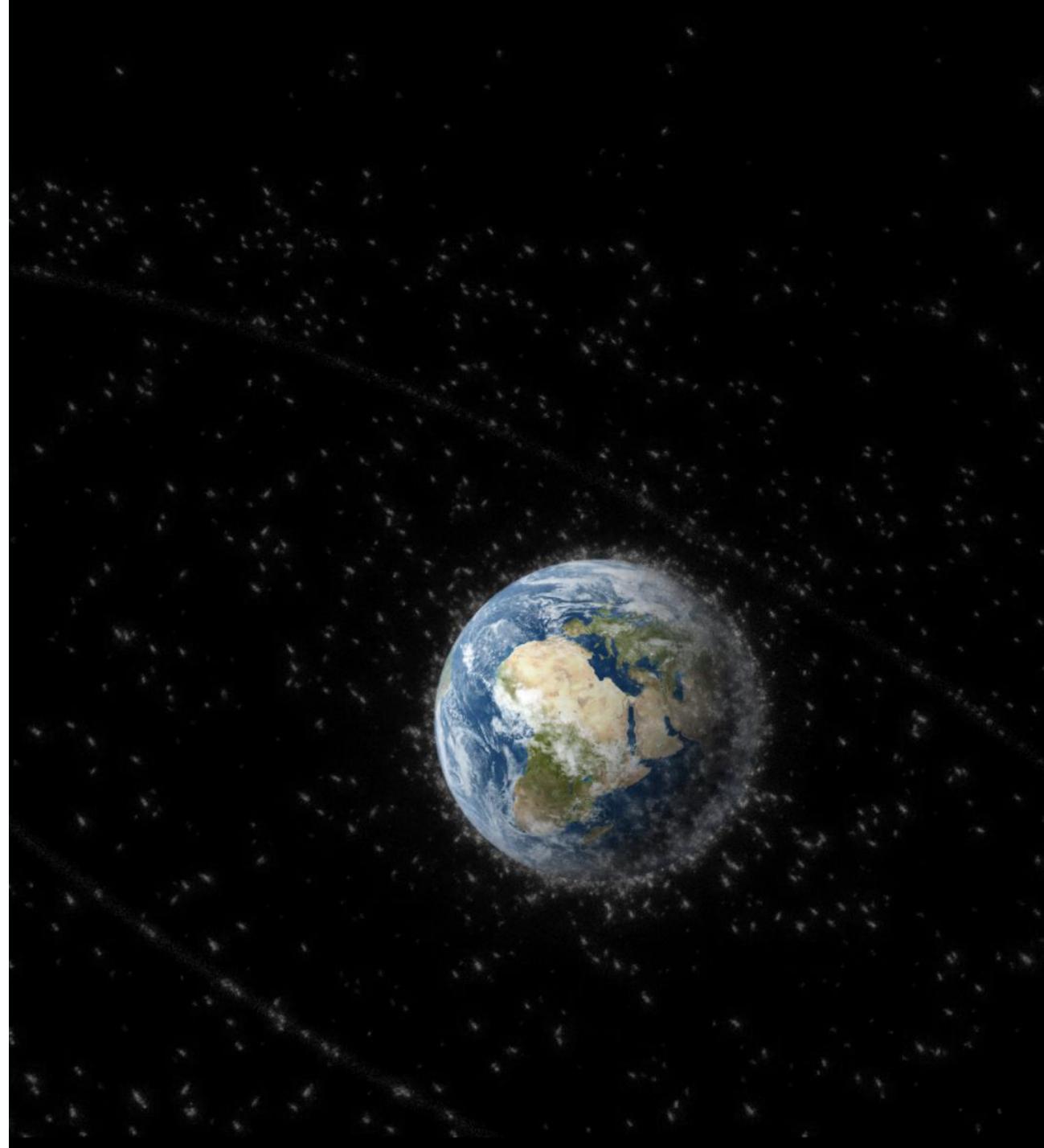
ESA's Space Debris Mitigation Toolkit



# **SPACE DEBRIS MITIGATION REQUIREMENTS**

a bit more in detail

BACK UP SLIDES





## Classical requirements with specified thresholds/**targets**

Pyrotechnics shall be designed not to release space debris larger than 1 mm in their largest dimension into Earth orbit.

Intentional break-up of a spacecraft or launch vehicle orbital element shall not be performed.

A spacecraft or launch vehicle orbital stage operating in Earth orbit shall be designed to guarantee a probability of successful passivation through to the end of life of:

- 1) At least 0,90
- 2) At least 0,95, when operating in the LEO protected region in an orbit with a natural orbital decay duration longer than 25 years
- 3) At least 0,95, when operating in the GEO protected region



## **Seed** requirements i.e. request of quantification/assessment

During the design, the developer of a spacecraft operating in near Earth orbit with a recurrent manoeuvre capability shall quantify the operational impact during normal operations due to conjunctions.

The developer of a spacecraft or launch vehicle orbital element injected in near Earth orbit shall quantify:

- the expected number of conjunctions at  $10^{-4}$  and  $10^{-6}$  collision probability threshold,
- the estimated number of collision avoidance manoeuvres triggered thereby on other spacecraft during normal operations and after end of life until re-entry or up to 100 years.

LIFETIME

## High risk

natural orbital decay duration  
between 5 and 25 years

## Medium risk

natural orbital decay up to 5 years  
and crossing altitudes above 375 km

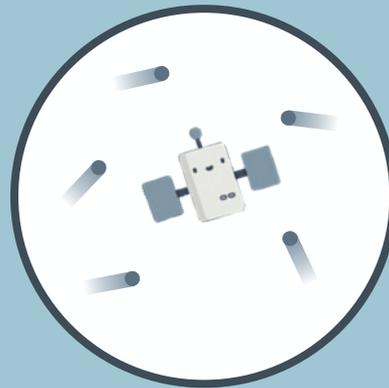


## Very high risk

natural orbital decay duration  
longer than 25 years

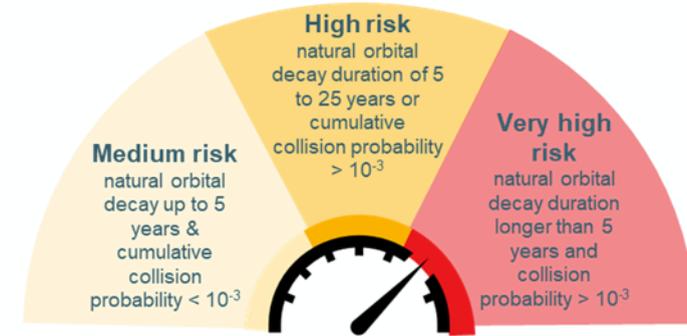
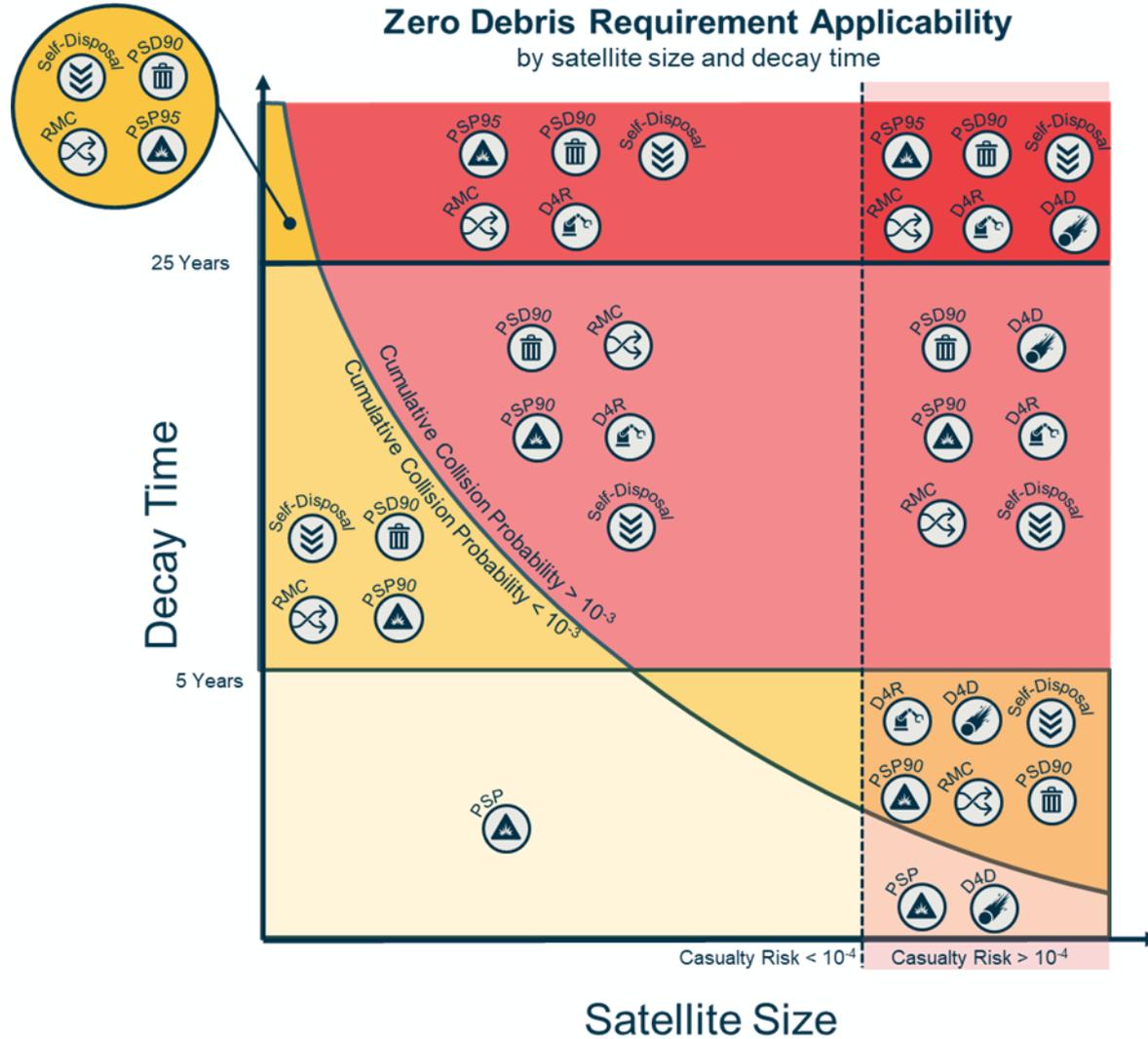
COLLISION PROBABILITY

Collision probability with  
**space debris** objects  
larger than **1 cm**



A space object in Earth orbit  
without capability of performing  
collision avoidance manoeuvres  
and with a cumulative collision  
probability with space objects  
larger than 1 cm above **1 in 1000** is  
considered **environmentally  
hazardous**.

# ESSB-ST-U-007 rationale – example for single satellite



## Key

- PSD90  
Probability of Successful Disposal  $> 90\%$
- DAD  
Design for Demise
- DAR  
Design for Removal
- RMC  
Recurrent Manoeuvre Capability
- PSPX  
Probability of Successful Passivation  $> X\%$
- Self-Disposal  
Self-Disposal Capability

# ESSB-ST-U-007 scope: orbital regions

Protected regions (i.e. LEO and GEO)

Near-Earth orbits (perigee < 100000 km)

Earth orbits (including Libration Point Orbits)

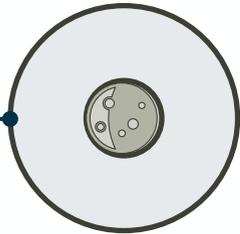
Lunar orbits (including Libration Point Orbits)

examples

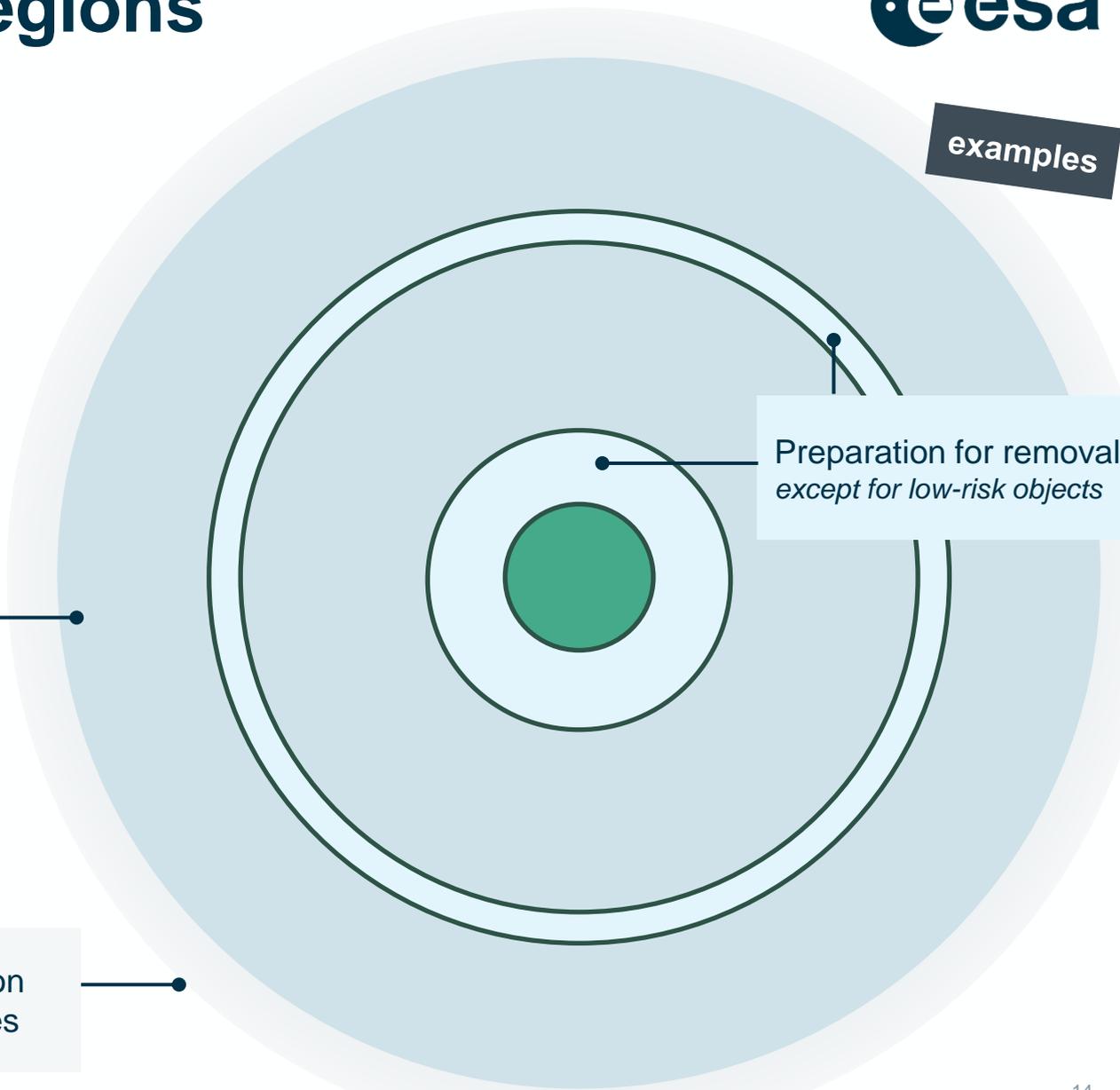
Preparation for removal  
except for low-risk objects

Acceptable collision  
probability per  
conjunction < 1:10000

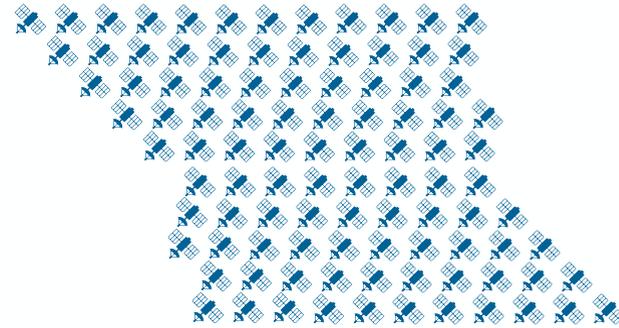
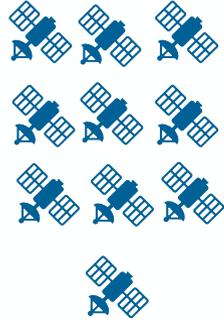
Analysis of  
disposal options



Passivation  
capabilities



# ESSB-ST-U-007 scope: space system type



## Single spacecraft

## Constellation (≥ 10 spacecraft)

## Large constellation (≥ 100 spacecraft)

## Launch vehicle (including elements, and orbital stages)

Request for collision avoidance capability in GEO and LEO if high or very high risk

Request for collision avoidance capability in near-Earth orbit

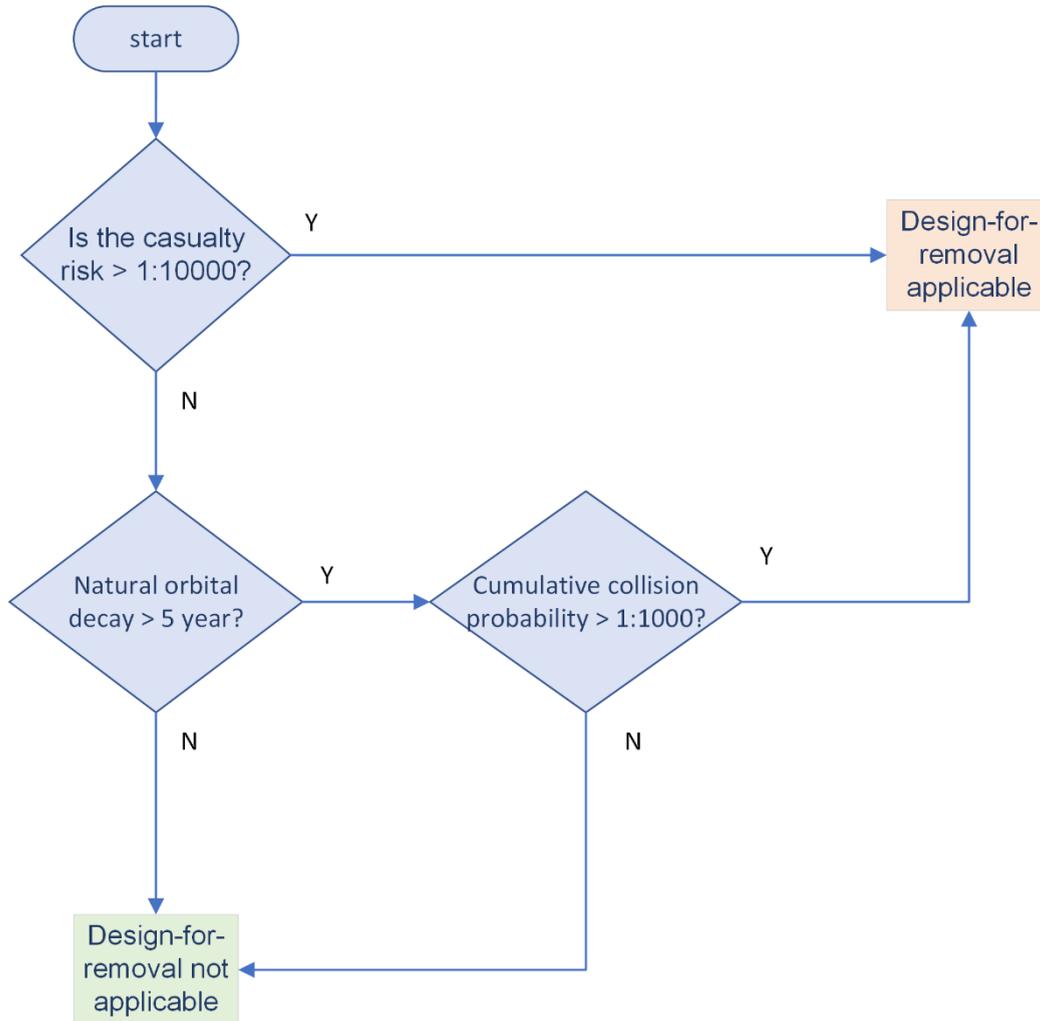
System reliability > 0.95

In LEO, disposal below 375 km and injection orbit with natural decay time < 5 years

Re-entry casualty risk per spacecraft < 1:10<sup>6</sup>

# Preparation for removal

novelty level



The requirements cover several aspects

- Mechanical **interfaces**
- Support to passive **navigation**
- Assessment of long-term **attitude**
- Attitude reconstruction from ground
- Limiting and damping **angular rates**
- **Operations**
- ...



# Re-entry

Note to the novelty indicator  
Novelty level of the requirement. High novelty does not mean that the requirement is difficult to achieve

novelty level

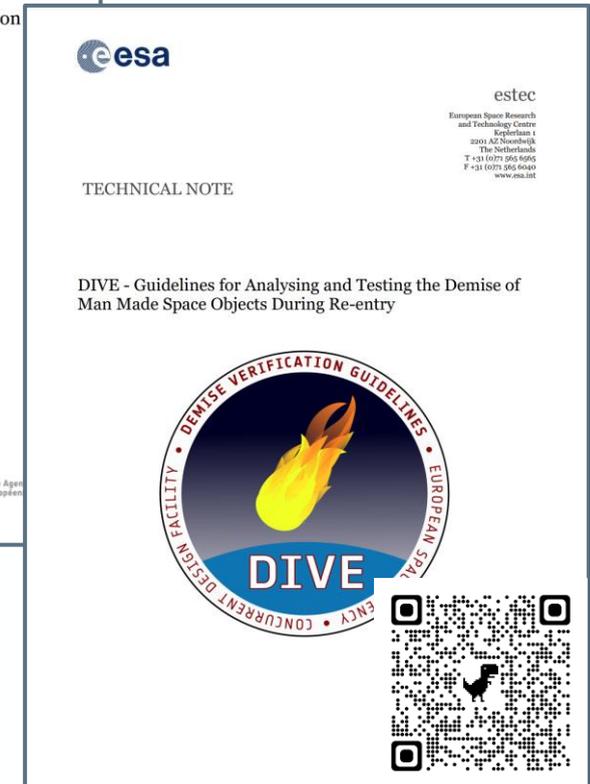


The main requirement has not changed:  
**re-entry casualty risk  $< 10^{-4}$**

## What's new

Explicit request for a **probabilistic assessment** of the casualty risk

- Uncertainty sources to be considered described in ESA Space Debris Mitigation Compliance Verification Guidelines (ESSB-HB-U-002-Issue 2),
- Modelling guidelines in DIVE - Guidelines for Analysing and Testing the Demise of Man-Made Space Objects During Re-entry (ESA-TECSYE-TN-018311)



# Lunar orbits

No intentional **breakup**

No **release** of space debris during normal operations

Space & ground segment designed to have **ephemerides** available for space traffic coordination

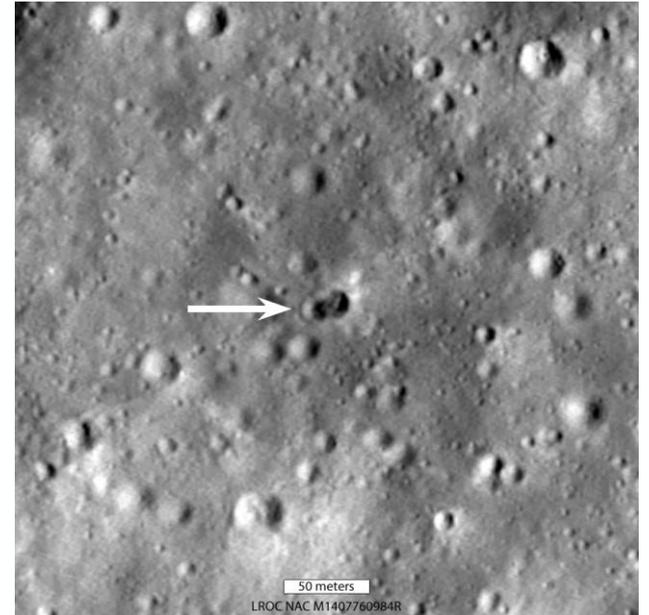
**Disposal** by one of the following means in order of preference:

1. **Heliocentric** orbit
2. Lunar impact, Earth re-entry, or a Lunar graveyard orbit

The free drift trajectories after disposal of a spacecraft or launch vehicle orbital element in lunar orbit shall be analysed for at least 100 years to evaluate:

1. Probability of **Earth re-entry** and its associated impact area
2. Probability of **Lunar impact** and its associated impact area

Double crater created by the impact of a rocket body on the Moon in March 2022.  
Credits: NASA/Goddard/Arizona State University





[space.debris.mitigation@esa.int](mailto:space.debris.mitigation@esa.int)



ESA Space Debris  
Mitigation  
Requirements

<https://technology.esa.int/upload/media/ESA-Space-Debris-Mitigation-Requirements-ESSB-ST-U-007-Issue1.pdf>



ESA Space Debris  
Mitigation Compliance  
Guidelines

<https://sdup.esoc.esa.int/documents/download/ESSB-HB-U-002-Issue214February2023.pdf>



ESA Tools (DRAMA)

<https://sdup.esoc.esa.int/drama/>