





NOVEL COMPOSITE SANDWICH STRUCTURES WITH INTEGRATED VACUUM INSULATION PANELS

23rd International Conference on Composite Materials (ICCM23)

30 July - 4 August 2023

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1. DIEHL AVIATION

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ICCM 23

BELFAST 2023 30 JULY – 4 AUGUST



DIEHL

Aviation

Cabin Interiors:

- Floor-to-Floor (sidewalls, ceilings, overhead stowage etc.)
- Crew-Rest Compartments
- Air Distribution
- Galley
- Lavatory
- General Cabin Equipment





Passengers of commercial aircrafts are spending an increasing amount of time in the aircraft cabin during travel. Therefore, they pay more attention to the **comfort of the cabin**. In order to achieve a satisfied passenger experience, the aircraft industry is seeking to improve the cabin comfort continuously. Some of the main factors that affect passenger comfort are **cabin climate, noise and space**. With the intention of **space increase** and simultaneously **improvement of thermal and acoustic situation** in the aircraft cabin, the search for better insulation materials led to **Vacuum Insulation Panels (VIP)**.



Figure 1 – Example of a sidewall with a conventional secondary insulation

Figure 2 - Schematic representation of two types of structure proposals for new interior panels

Figure 3 – Sidewall with attached Vacuum Insulation Panel 4





Figure 4 – Potentials of the new interior panels with integrated VIP





Figure 5 – Potentials of the new interior panels with integrated VIP



Application in aircraft cabin:

- Lining panels (sidewall, door frame lining, door, ceiling)
- Crew Rest Compartments
- Monuments (galley, lavatory)
- Air ducting and others



Figure 6 – Sidewall panels, door frame lining and door, galley, Source: www.diehl.com/aviation



3. STATE OF THE ART – AIRCRAFT INSULATION

The **conventional insulation** in aircraft (primary and secondary) fills the space between the interior panels and fuselage of the aircraft [1]. Glass wool and thermoplastic foams (e.g. PI, PE) are commonly used as insulation material.

The thermal conductivity of such insulation is in the range of 30 - 50 mW/m·K.

Problem of conventional insulation:

Water accumulation in glass wool blankets, which causes following issues:

- "rain in the plain" effect
- increase in the total weight of the aircraft
- degradation of the thermal and acoustic properties of the glass wool packages
- corrosion of the primary structure
- growth of mold and bacteria





Frame insulation Figure 7 – Primary and frame insulation [1]

Figure 8 – Sidewall with secondary insulation on test rig [1]



Figure 9 – Water accumulation in glass wool blankets [1]



3. STATE OF THE ART – INTERIOR PANEL



Figure 10 – Sandwich structure with honeycomb core

Conventional interior panels have a **honeycomb sandwich structure** covered by face sheets from both sides. This **structure** fully **meets requirements** regarding weight, mechanical loads, environmental conditions and flammability.

Acoustic and thermal insulation blankets are **attached** to the **rear side** of interior panels as secondary insulation.

The **production** of the insulation blankets and its **fixation** mean additional steps in the production chain, which is **time consuming** and requires an **enormous amount of manual work**.



Figure 11 – Conventional interior panel on example of sidewall



3. STATE OF THE ART – VACUUM INSULATION PANEL

Vacuum Insulation Panel consists of:

- Fumed silica powder (core)
- High Barrier Foil (MF2)

Thermal conductivity = $4 \text{ mW/m} \cdot \text{K}$

Applications:

- Fridges and freezers;
- Transport containers;
- Building industry.



Figure 12 – Vacuum insulation panel



Figure 13 – Various shapes of Vacuum insulation panel, Source: va-Q-tec AG



Figure 14 – Comparison of VIP to conventional materials (glass wool and foams), Source: ta-netzsch.com



4. STRUCTURE DEFINITION

Four structure for new sandwich panels with integrated VIP were **proposed** based on theoretical analysis and analysis of requirements for interior panels and VIP.





5. MANUFACTURING – FEASIBILITY STUDY

Feasibility study tests were performed for all four structure solutions in order to produce flat panels for thermal conductivity tests.





4. DOUBLE SANDWICH (VIP BONDED ON HONEYCOMB CORE)

Result:

- evacuation of proposals 1 and 3 is possible;
- good surface quality;
- good bonding between core and face sheets;
- · structure solutions have robust structure;
- all solutions are feasible.



6. THERMAL CONDUCTIVITY

All proposals were tested¹ for thermal conductivity. Following test results were obtained:

Proposal	Description	Thermal conductivity, mW/m·K	W/m-K	70 60
Conventional panel 1	Honeycomb panel w/o insulation	65	ivity, m	50
Conventional panel 2	Honeycomb with insulation	42	nducti	40
Solution 1	Honeycomb + Silica Powder (evacuated)	24,5	mal co	30
Solution 2a	VIP + PES Foam	10,3	Ther	20
Solution 2b	VIP + PVC Foam	11,5		10
Solution 2c	VIP + PMI Foam	11,2		C
Solution 3	PU Aerogel foam (evacuated)	11,5		
Solution 4	Double sandwich (Honeycomb + VIP)	12,6		
VIP	Sub component for proposals 2 and 4	6,7		



Results: the new solutions showed a big difference in thermal properties in comparison to conventional structures. The thermal conductivity of **new structures** is **lower by the factor of 3-6**. Even lower values can be achieved with the relevant modifications.



7. CONCLUSION AND OUTLOOK

- In this research project the **integration** of **highly efficient insulation VIP** into cabin interior panels was **investigated**;
- **Four structure solutions** were proposed for further feasibility study tests;
- The experiments demonstrated that all new solutions can be manufactured;
- The thermal conductivity of new structures is lower by the factor of 3-6 in comparison to conventional solutions;



- **Further investigations** for vacuum maintaining, mechanical properties, acoustics, fire properties, manufacturing of curved panels and others are **on going**.
- Besides the cabin interiors for aircrafts, the new sandwich structures are also **highly attractive** for application such as **urban air mobility** (UAM), **trains**, **recreational vehicles** (RV), **caravans** and **transport containers**.



ACKNOWLEDGEMENTS

The authors thank following persons for their support, valuable suggestions and interesting discussions:

- Pfetscher, J.B. Miné, C. Hornemann, M. Steinmayer, H. Jehle from company Diehl Aviation Laupheim GmbH;
- Dr. F. Almeida, H. Feuerstein, N. Brzoska-Steinhaus and T. Bock from company va-Q-tec AG;
- N. Andresen from Fraunhofer Institute for Building Physics IBP;
- and other persons.



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Thank you!



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