



**Jet Propulsion Laboratory**  
California Institute of Technology

# A New Method for Asteroid Impact Monitoring and Hazard Assessment

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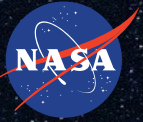
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# Introduction

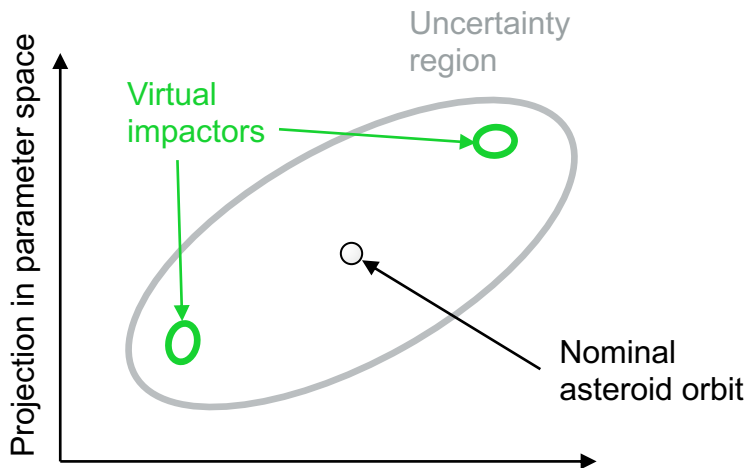
## Motivation for a new impact monitoring algorithm



### Goal of Impact Monitoring

Identify and characterize all virtual impactors (VIs) compatible with the orbital uncertainty distribution.

**VI:** region in parameter space leading to impacts along the same dynamical path.\*



### Challenges

1. Many VIs that must be separated.
2. Impact probabilities (IP) are usually small ( $\sim 10^{-7}$ ).
3. Nongravitational parameters must be handled automatically.
4. Pathological cases (Earth-like orbits, nonlinearities).

#### • Monte Carlo:

- |      |      |      |      |
|------|------|------|------|
| 1. ✗ | 2. ✗ | 3. ✓ | 4. ✓ |
|------|------|------|------|

#### • Line of Variations (LOV):\*

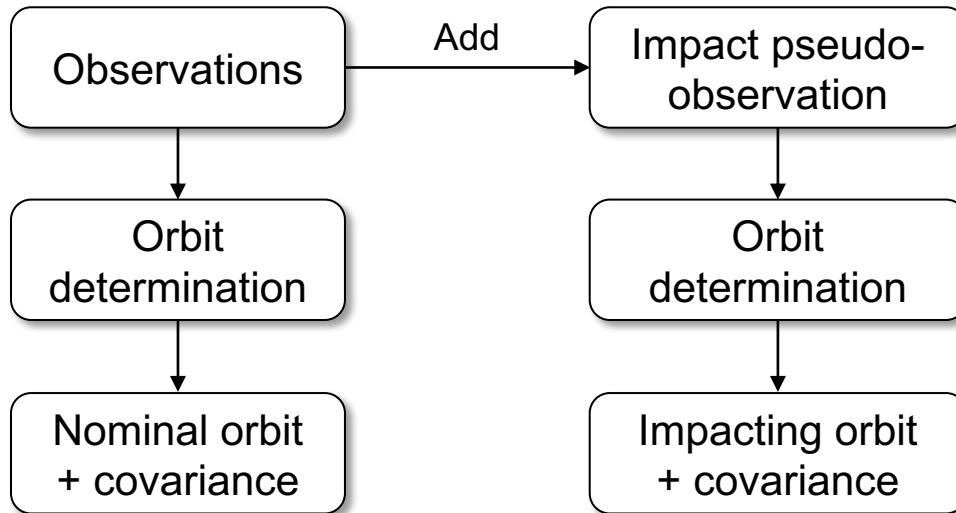
- |      |      |      |      |
|------|------|------|------|
| 1. ✓ | 2. ✓ | 3. ✗ | 4. ✗ |
|------|------|------|------|

Increased rate of NEA discoveries calls for a more robust method

\*Milani, A. et al. (2005): "Nonlinear impact monitoring: line of variation searches for impactors," Icarus, 173, 362-384

# Impact Pseudo-Observation I

Finding VIs using an orbit-determination program



## Impact pseudo-observation

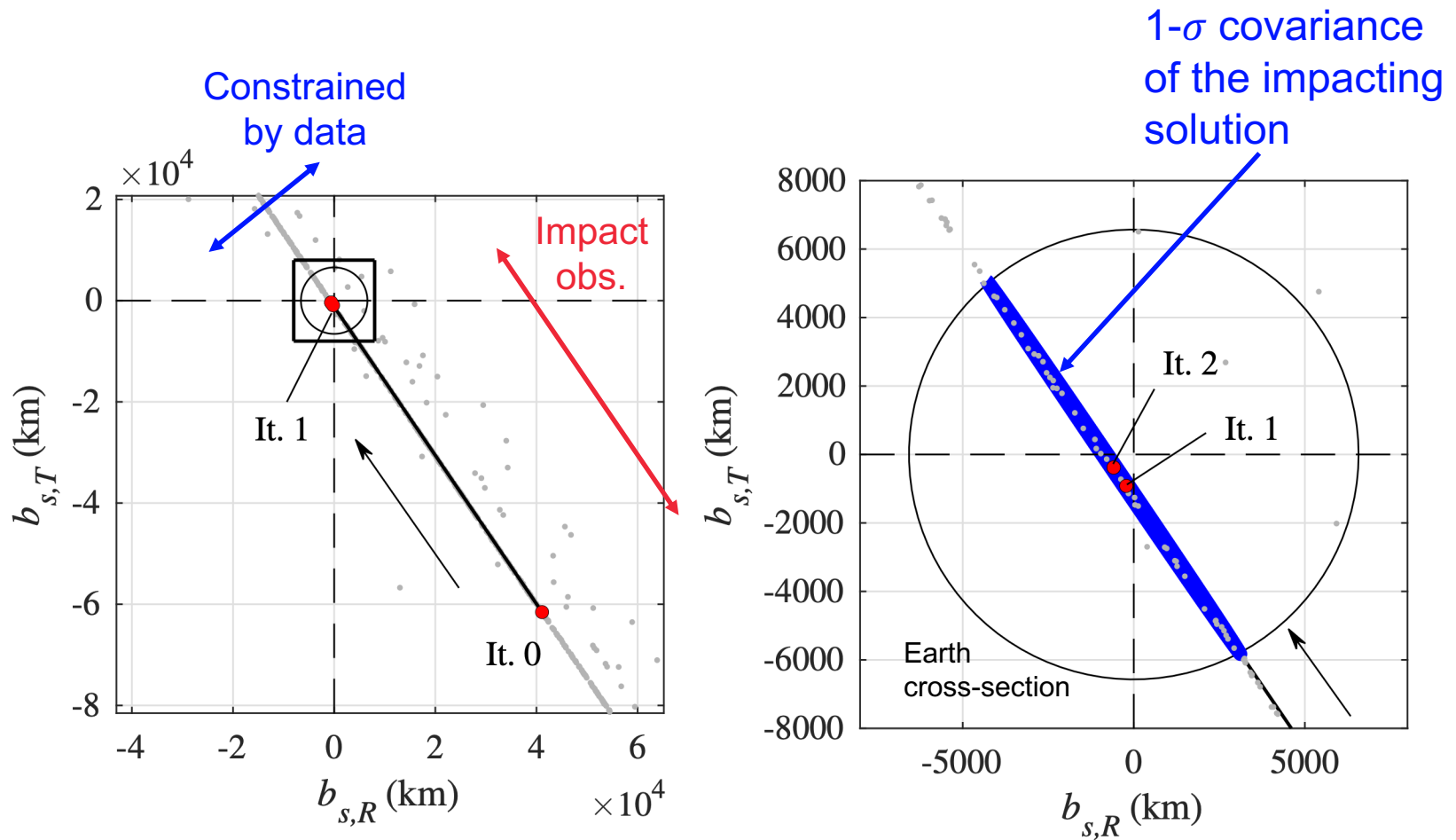
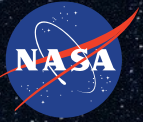
Add the impact condition as an observation:

- The residuals are the B-plane coordinates at the time of close approach.
- The uncertainty is a fraction of the Earth radius.

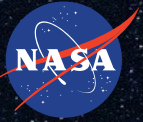
- No simplifying assumptions about the dynamics or the uncertainty distribution in parameter space.
- The covariance of the fit approximately models the VI in parameter space.
- Use the same operational OD program used to produce the nominal orbit.

# Impact Pseudo-Observation II

Finding VIs using an orbit-determination program







## Operation of Sentry-II

1. **Initial MC exploration:** detect close approaches for further investigation.
2. **Find VIs:** run OD filter extended with the impact pseudo-observation.
3. **Characterize the VIs:** use importance sampling to estimate the IP.

## Features and Comparison with Sentry (LOV based)

- Sentry-II has been running and mirroring Sentry for a few months.
- Sentry-II handles nongravitational parameters systematically.
- More robust in pathological cases.
- Median runtimes for 100-year exploration and IP down to  $10^{-7}$ :
  - Monte Carlo: 14 days (20,000 min)
  - Sentry: 20 min
  - Sentry-II: 40 min
- Sentry-II provides the nominal orbit and the uncertainty of each VI  
⇒ useful for negative observation campaigns.



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