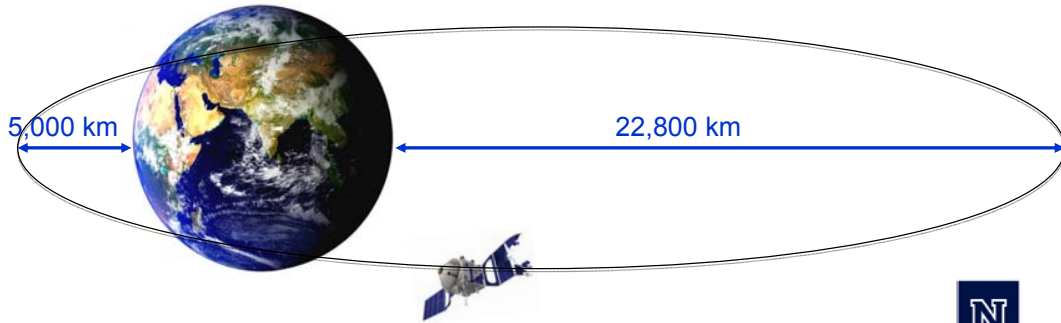


FOCOS Mission Concept – Fundamental physics with an Optical Clock Orbiting in Space

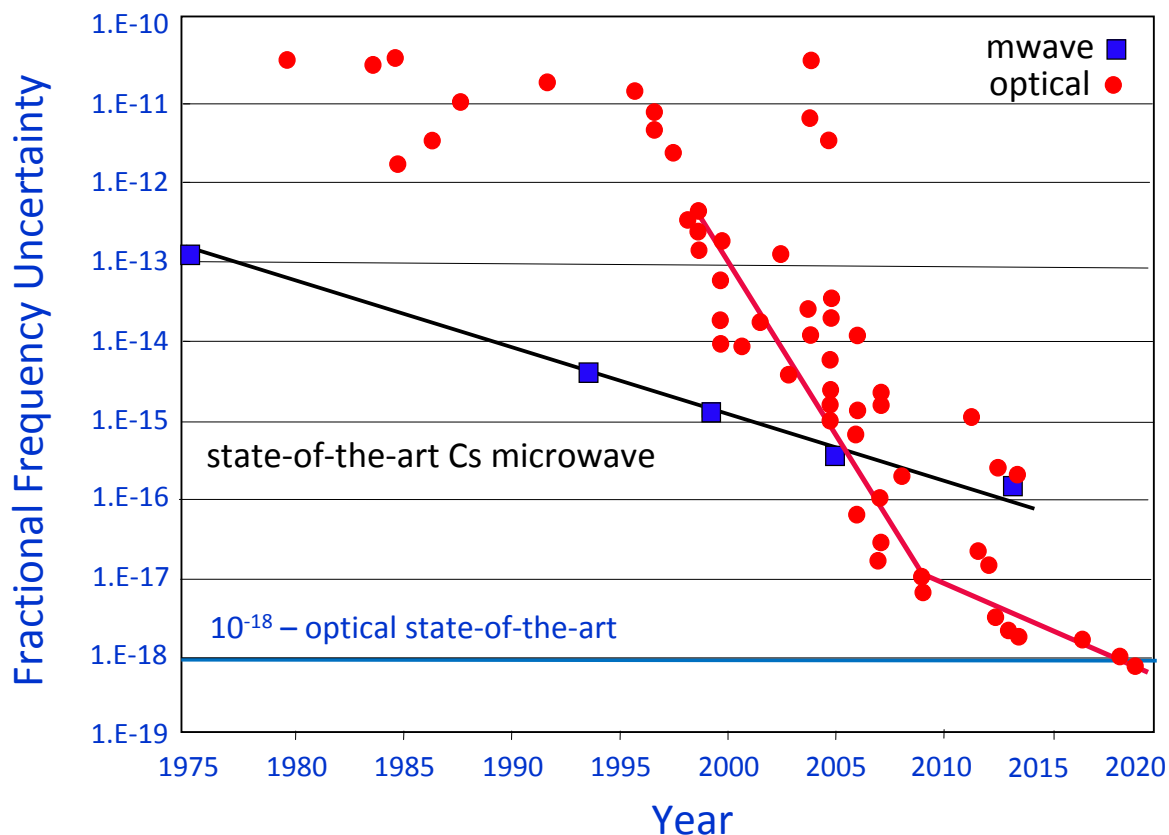


Andrei Derevianko, U. of Nev.-Reno
Kurt Gibble, Penn State U.
Leo Hollberg, Stanford U.
Nate Newbury, NIST
Marianna Safronova, U. of Del.
Laura Sinclair, NIST
Nan Yu, JPL
(Chris Oates, NIST)



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Improvement of the Uncertainty of atomic clocks

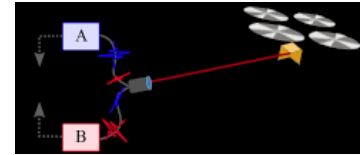


Impacts of Optical Frequency Clocks

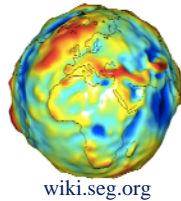


1. Metrology: Redefinition of the SI Second (within the next decade?)
2. Time Transfer: clock performance surpasses long-distance time transfer capabilities (10^{-18} vs. 10^{-16})

Fiber networks over 1,000 km, free-space over 100 km.



3. Geodesy: clock performance exceeds geodetic knowledge at higher altitudes.
 $10^{-18} \leftrightarrow 1$ cm



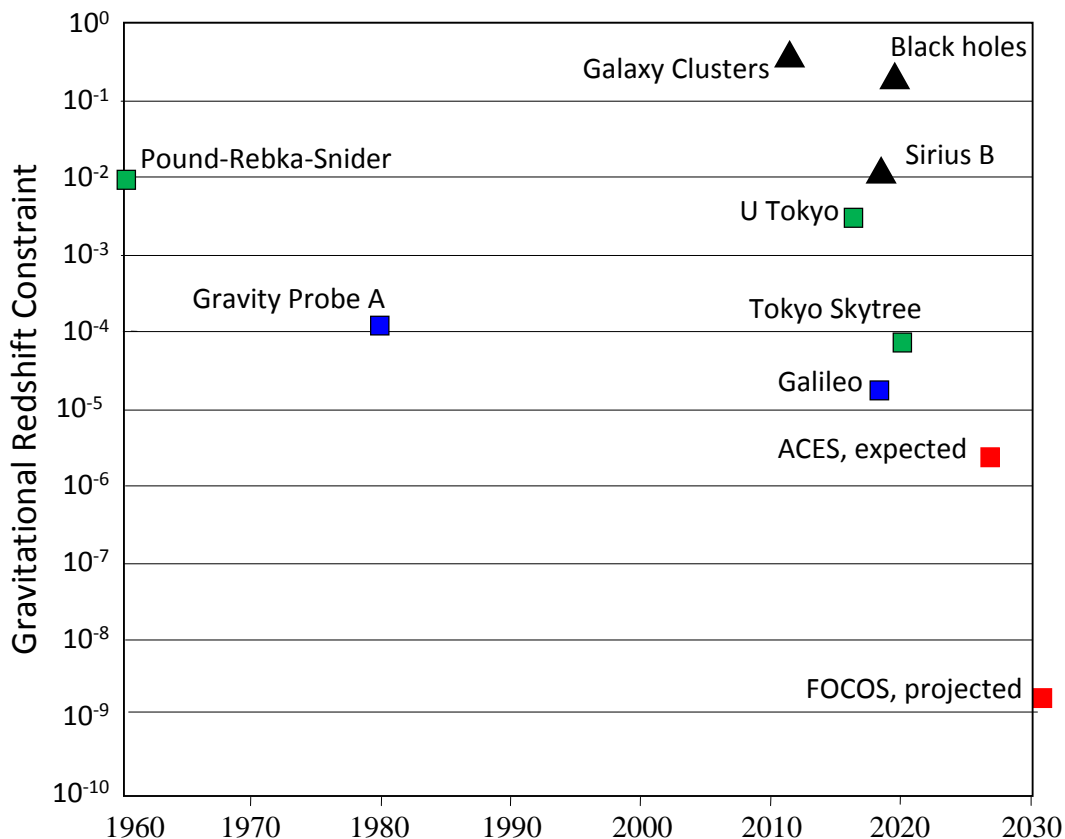
wiki.seg.org

Chronometric geodesy – clocks in space

4. New possibilities for next generation of fundamental physics tests

- Test General Relativity with much higher sensitivity
- Search for new physics

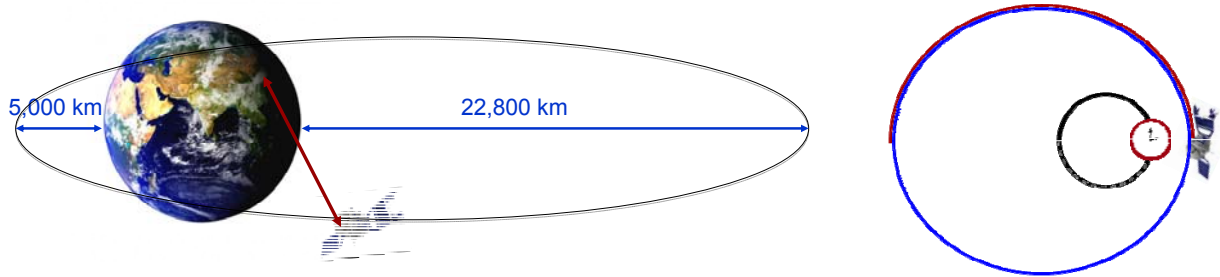
Tests of the Gravitational Redshift



FOCOS Mission concept



A high-performance optical clock in an elliptical orbit



Optical clock performance: $1 \times 10^{-16} \tau^{-1/2}$, 1×10^{-18} uncertainty

Optical link: $< 1 \times 10^{-16} \tau^{-1}$, $< 5 \times 10^{-19}$ bias

Range at periapsis known to 1mm (two-way Doppler link) and velocity to $1 \mu\text{m/s}$.

Orbit modulates gravitational potential: 2.4×10^{-10} variation.

Orbit observation times (2 hours apoapsis, 30 minutes periapsis) enable redshift uncertainty of 2.4×10^{-8} in $1\frac{1}{2}$ 8-hour orbits.

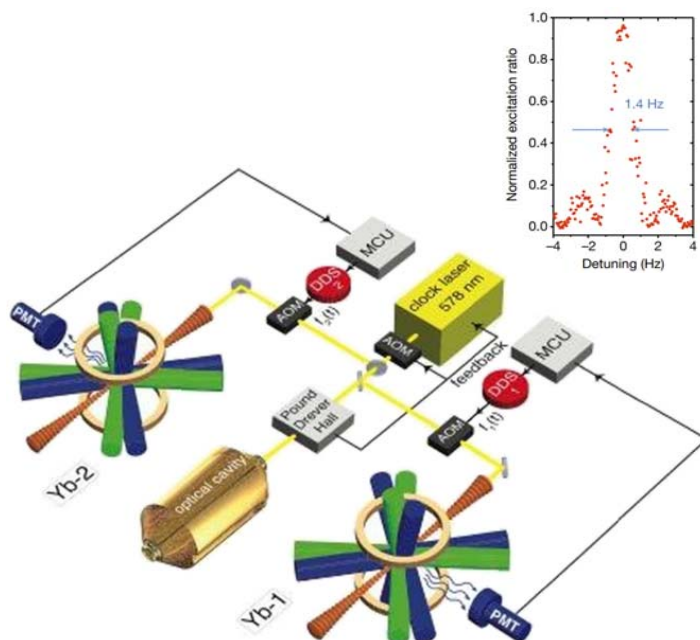
Average 100 orbits to reach 2.4 ppb.

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FOCOS Schematic



NIST Dual Yb Optical lattice clock



Two-way optical link

Talk after break by Fabrizio



FOCOS science goals

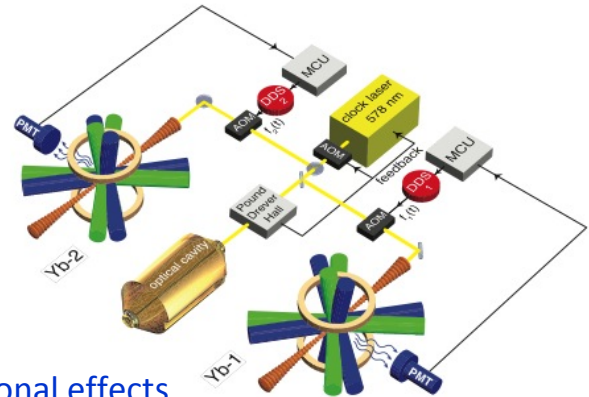


Fundamental physics:

1. Gravitational Redshift – improve the uncertainty by 30,000x to 2 ppb.
2. Local Lorentz Invariance (Kennedy-Thorndyke tests - SME coefficients)

Cavity-clock comparisons - orientation and velocity

Large expected gains in sensitivity.



3. Test higher-order relativistic and gravitational effects

Requires precision orbit determination

GRACE, etc. for the gravitational field

4. Post-Newtonian effects on the satellite orbit (through two-way link data)

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FOCOS Science Goals – International Clock Network



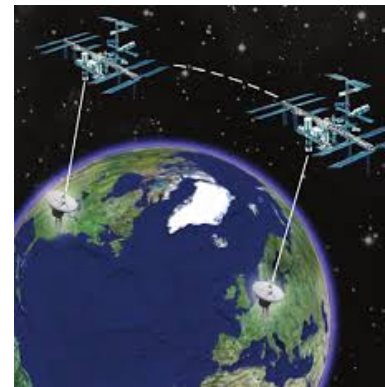
Improved network of earth clocks – a step towards a space network.

Fundamental physics:

5. Dark Matter searches in space and on Earth
– 100 to 1,000 × improvement
6. LPI tests – e.g., drifts of fundamental constants

Timing applications:

7. Worldwide timing: ns to ps level
8. Precision geodetic referencing at the mm-level
9. Space-time reference

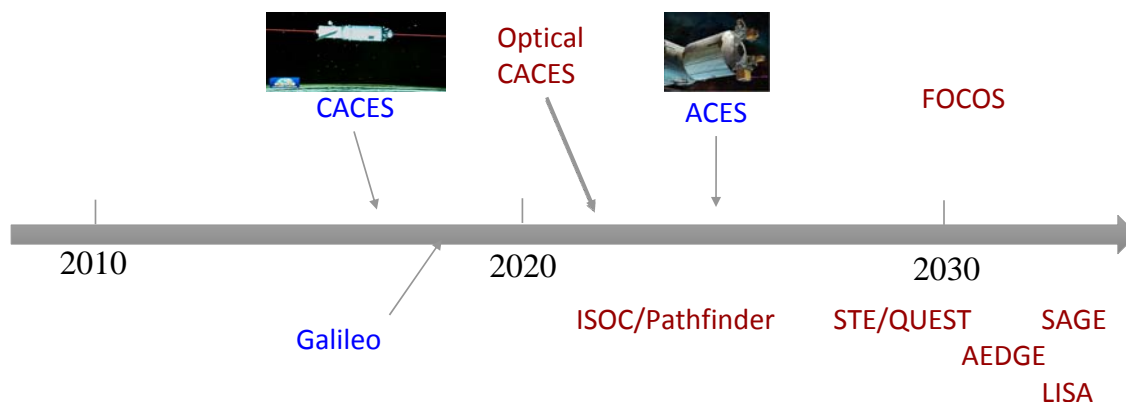


ACES

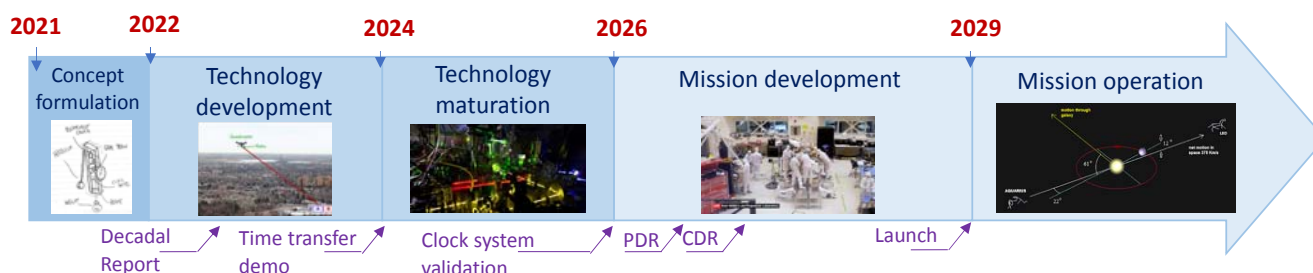
Pathfinder-style mission for future atom interferometry (Equivalence Principle), clock constellation in space, laser/atom-based gravity wave detection

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Timelines



International collaborations are sought: SWaP development, reference cavities, lasers and link hardware, theoretical contributions ...

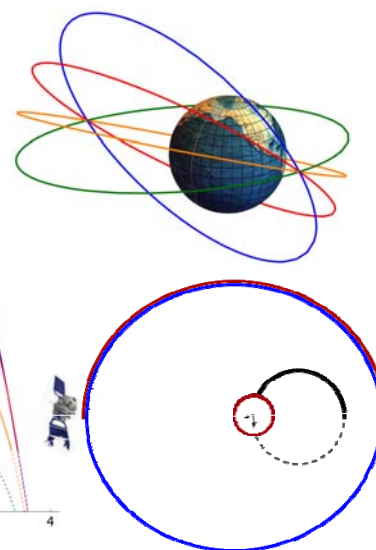
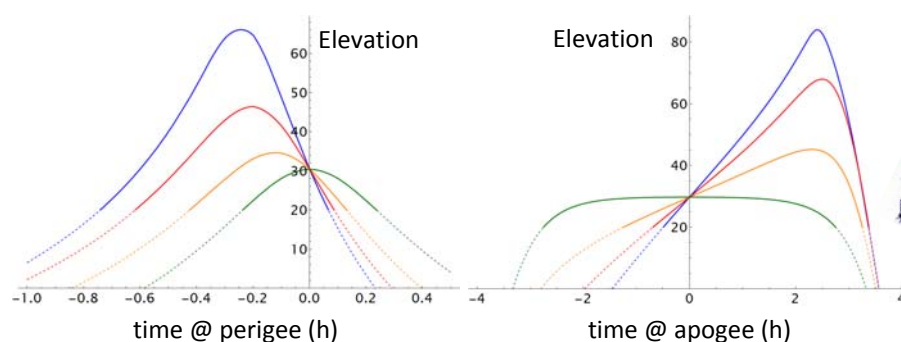


Choice of Orbit

An 8-hour orbit enables observation of periapsis and apoapsis within 12 hours, and reasonable ranges for the laser link.

Decreasing the perigee altitude gives a larger redshift variation, at the cost of observation time (and drag for very low perigees).

An orbit inclination of 9° gives equal perigee and apogee maximum elevations of 30° at 40° N for a 5,000 km perigee altitude.



Detailed analysis is needed to determine precise period, RAAN, perigee altitude ... to optimize worldwide visibility and redshift measurement.

FOCOS Size Weight and Power



The first cost estimate (\$451M*) for FOCOS conservatively used 500 W and 266 kg.

Actual power used by NIST portable Yb lattice clock is ~500 W (450 kg, 1.9 m³, A. Ludlow).

Minimal effort to reduce power, mass and size.

A recent exercise estimated 200 W for a lattice clock with one comb (+ 50 W telescope and link with a 2nd comb, N. Newbury).

Target for \$300M* NASA medium explorer mission class (Explorers Program) is 210 W, 227 kg, and 0.5 m³.

A modest investment in SWaP development is likely to meet those requirements.

\$5 to \$10M of development may reduce mission cost by >\$150M.*

(SWaP for a \$200M* mission: 170 W & 125 kg)

Reducing lattice clock performance does not yield large cost reductions.

*Cost information is for budgeting and planning and is intended for informational purposes only. It does not constitute a commitment on the part of JPL or Caltech.

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Wesley Brand