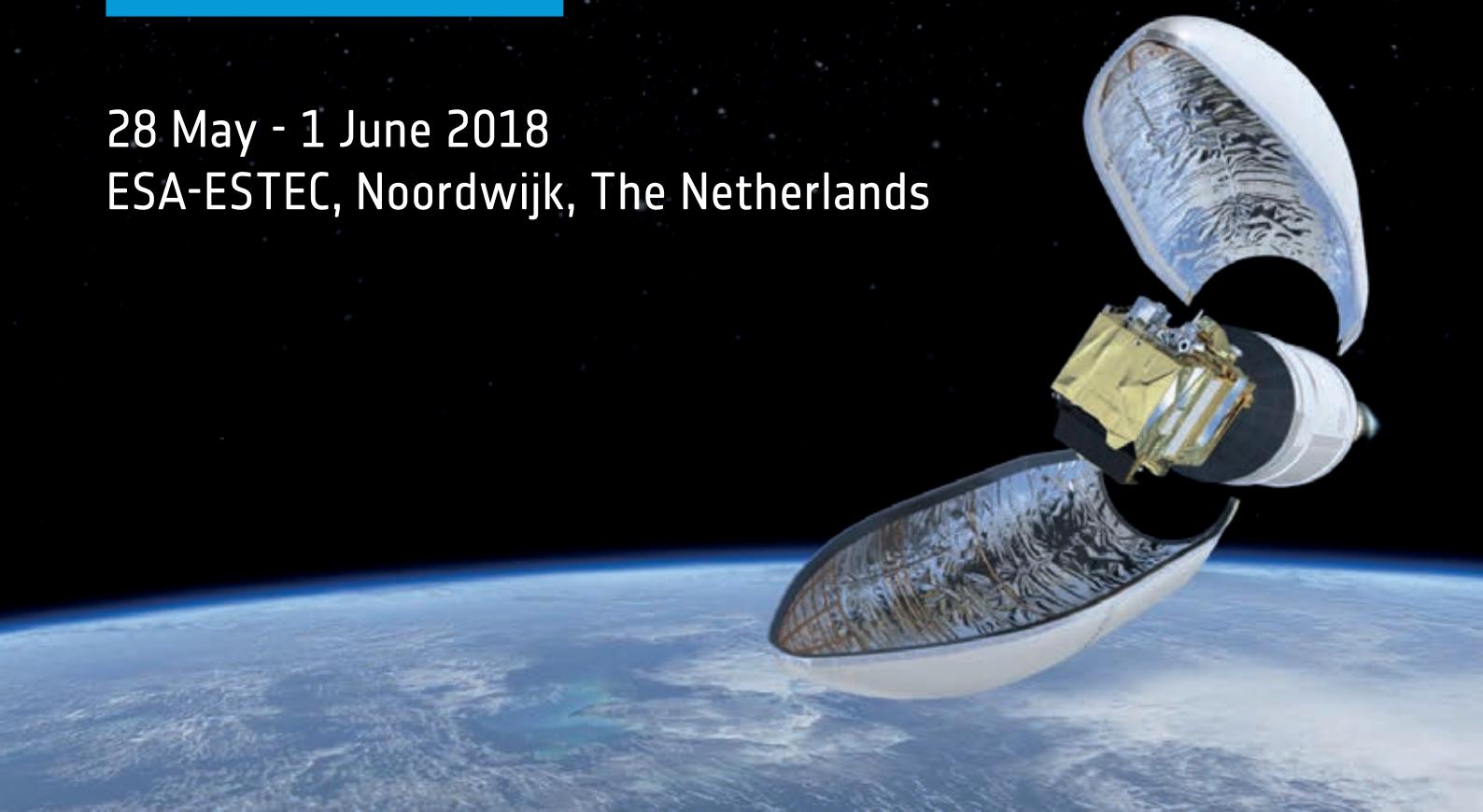


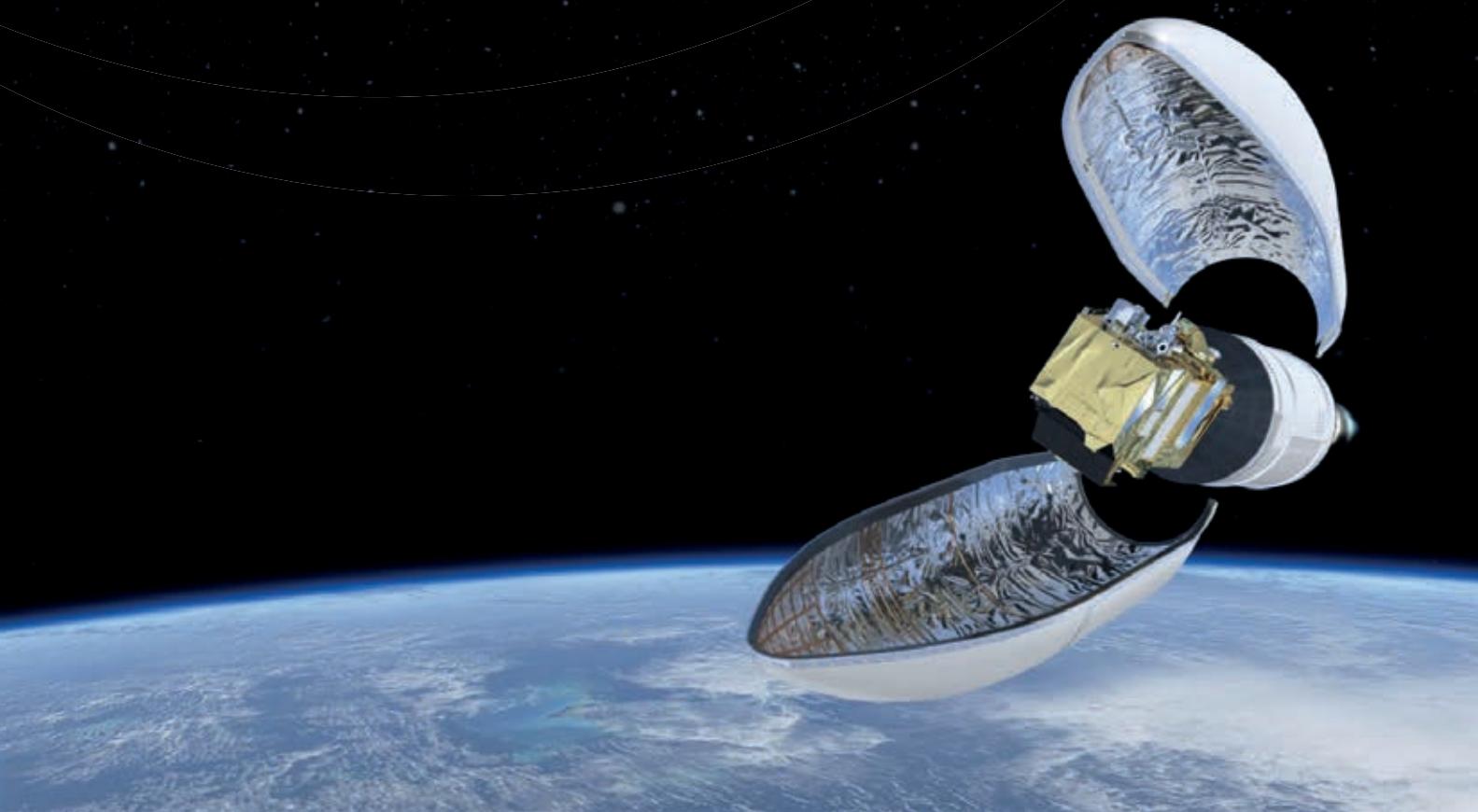


# 15<sup>TH</sup> EUROPEAN CONFERENCE ON SPACECRAFT STRUCTURES, MATERIALS AND ENVIRONMENTAL TESTING

28 May - 1 June 2018  
ESA-ESTEC, Noordwijk, The Netherlands



**15<sup>TH</sup> EUROPEAN  
CONFERENCE ON  
SPACECRAFT  
STRUCTURES,  
MATERIALS AND  
ENVIRONMENTAL  
TESTING**



# Day 1 - Monday 28 May 2018

09:00 - 13:00	REGISTRATION WINTERGARDEN	
13:00 - 14:00	WELCOME AND INTRODUCTION SESSION	
14:00 - 15:30	KEYNOTES	
	<b>Keynote 1</b> <b>Forty years of satellite and launcher mechanical experiences</b> <i>Prof. Dr. C. Stavrinidis, Technical Director, IABG</i>	
	<b>Keynote 2</b> <b>How to build more and more competitive and still robust Verification Plans for Structures</b> <i>Silvain Legrand, Mechanical Expert in the Technical Authority of the Competence Center Platform &amp; Integration, Thales Alenia Space</i>	
	<b>Keynote 3</b> <b>Mechanical System Testing on a Launcher Upper Stage</b> <i>Detlev Bütten, Team Leader "Dynamic &amp; Testing" Mechanical Engineering, ArianeGroup</i>	
15:30 - 16:00	BREAK	
	NEWTON 1	NEWTON 2
16:00 - 18:30	Spacecraft Design Architecture	Launcher Structures
	Chair: G. Kling - P. Martimort	Chair: P. Mourey - T. Ghidini
16:00 - 16:30	<b>S2TEP - A New Scalable Satellite Platform</b> <i>S. Reershemius<sup>1</sup>, T. Spröwitz, M. Jetzschmann, F. Dannemann <sup>1</sup>German Aerospace Center (DLR), Germany</i>	<b>Airbus DS Ariane 6 Structures</b> <i>A. Sanchez<sup>1</sup> <sup>1</sup>Airbus Defence &amp; Space Spain, Spain</i>
16:30 - 17:00	<b>MetOp-SG ICI Mechanical Architecture</b> <i>D. Sanchez-Pascuala Valencia<sup>1</sup> <sup>1</sup>Airbus Defence &amp; Space, Spain</i>	<b>New CFRP Basin Bulkhead Concept for Ariane 6: Shape Optimization</b> <i>M. G. Cosío<sup>1</sup>, J. Martín<sup>2</sup>, M. Angel Alén<sup>1</sup> <sup>1</sup>Citd Engineering &amp; Technologies, Spain, <sup>2</sup>Airbus D&amp;S, Spain</i>
17:00 - 17:30	<b>Novel CFRP Stable Structure for Sentinel 5 UVNS Optical Instrument</b> <i>S. Lucarelli<sup>1</sup>, M. Pellizzari<sup>1</sup>, D. Matias Boveda<sup>1</sup>, D. Ende<sup>1</sup>, G. Laduree<sup>2</sup> <sup>1</sup>Airbus Defence And Space, Germany, <sup>2</sup>ESA-ESTEC, The Netherlands</i>	<b>Development and Validation of a Composite Solid Rocket Motor Case</b> <i>M. Wolff <sup>1</sup>MT Aerospace AG, Germany</i>

REGISTRATION WINTERGARDEN		
BREAK		
EINSTEIN	ERASMUS	ESCAPE
<b>Acoustic Testing</b>	<b>Additive Manufacturing System Aspects</b>	
Chair: A. Kolaini - S. Scharfenberg	Chair: K. Pfaab - L. Pambaguiyan	
<b>Direct Field Acoustic Testing in Space Industry Facility</b> <i>C. Fabries<sup>1</sup>, B. Brevart<sup>1</sup>, N. Saggini<sup>2</sup>, A. Ciriello<sup>2</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> Thales Alenia Space, Italy	<b>Development and Test of a Three and an Half Space Applications Using Additive Manufacturing Technologies</b> <i>P. Bigot<sup>2</sup>, C. Borbouse<sup>4</sup>, A. Chiavarini<sup>3</sup>, E. Chouteau<sup>5</sup>, Y. Garin<sup>3</sup>, A. Heck<sup>4</sup>, L. Jacques<sup>2</sup>, I. Liemans<sup>6</sup>, F. Montredon<sup>5</sup>, L. Pambaguiyan<sup>1</sup>, JY. Plesseria<sup>2</sup></i> <sup>1</sup> ESA-ESTEC, Netherlands, <sup>2</sup> Centre Spatial de Liège, Belgium, <sup>3</sup> Lambda-X, <sup>4</sup> Safran AB, <sup>5</sup> Thales Alenia Space, France, <sup>6</sup> Thales Alenia Space, Belgium, <sup>7</sup> 3D-Systems	
<b>Pre-Test Analysis for Design of MIMO Control Strategies on DFAX</b> <i>M. Alvarez Blanco<sup>1,2</sup>, K. Janssens<sup>1</sup>, A. Carrella<sup>1</sup>, B. Peeters<sup>1</sup></i> <sup>1</sup> Siemens Industry Software NV, Belgium, <sup>2</sup> KU Leuven, Belgium	<b>Systems Engineering and Systems Architecting Approaches For Innovative Additive Manufactured Spacecraft Structures</b> <i>L. Pollice<sup>1</sup>, M. Gschweidl<sup>2</sup>, R. Usinger<sup>2</sup>, P. Gaudenzi<sup>1</sup></i> <sup>1</sup> Sapienza University Of Rome, Italy, <sup>2</sup> RUAG Space, Switzerland	
<b>Acoustic Test Definition for Satellite Antenna Feed Qualification</b> <i>B. Brevart<sup>1</sup>, B. Boury<sup>1</sup>, B. Butul<sup>1</sup>, M. Heim<sup>2</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> Centre National d'Etudes Spatiales, France	<b>Issues for Design, Modelling and Laser Beam Manufacturing of Structures Including Aluminium Micro-Lattices</b> <i>G. Pommatau<sup>1</sup>, M. Raynaud<sup>1</sup>, K. Cabannes<sup>1</sup>, F. Montredon<sup>1</sup>, K. Brethome<sup>3</sup>, B. Bonvoisin<sup>2</sup>, M. Komarek<sup>4</sup>, N. Sordello<sup>5</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> ESA-ESTEC, The Netherlands, <sup>3</sup> CNES, France, <sup>4</sup> LKE, Czech Republic, <sup>5</sup> Mecano ID, France	

17:30 - 18:00	<b>Structural Design Advantages of High Performance Radiators (HiPeR)</b> S. Carli <sup>1</sup> , A. Allgaier <sup>1</sup> , M. Altenburg <sup>1</sup> , A. Chiri <sup>1</sup> , B. Maria Guitard Lejarreta <sup>1</sup> , V. Hoefig <sup>1</sup> , S. Lucarelli <sup>1</sup> , A. Maas <sup>2</sup> , D. Sausen <sup>1</sup> , J. Truenkle <sup>1</sup> , S. Olga Vismara <sup>1</sup> , P. Weimer <sup>1</sup> , J. Wingens <sup>1</sup> , C. Wuehrer <sup>1</sup> <sup>1</sup> Airbus Defence & Space GmbH, Germany, <sup>2</sup> Airbus Defence & Space Netherlands B.V., The Netherlands	<b>Ariane 6 VINCI Thrust Frame</b> M. Brooker <sup>1</sup> , H. Meijer <sup>1</sup> , P. van Schie <sup>1</sup> <sup>1</sup> Airbus Defence & Space Netherlands, Netherlands
18:00 - 18:30		<b>CALLISTO Project – Structural Design and Sizing Challenges in the Frame of a Reusable First Stage Demonstration Vehicle</b> O. Diaz Lopez <sup>1</sup> <sup>1</sup> CNES, France
18:30 - 21:00	<b>COCKTAIL SPACE EXPO</b>	

## Day 2 - Tuesday 29 May

	NEWTON 1	NEWTON 2
08:30 - 10:30	<b>Micro-Vibration Characterisation</b>	<b>Deployable Structures - 1</b>
	Chair: P. Corberand - M. Wagner	Chair: C. Hühne - A. Ihle
08:30 - 09:00	<b>The Spacecraft Micro-Vibration Management and Control Problem: A Survey of Architectures, Isolation Technologies, Modeling, and Testing Approaches</b> C. Dennehy <sup>1</sup> , O. Alvarez-Salazar <sup>2</sup> <sup>1</sup> NASA Goddard Space Flight Center, United States, <sup>2</sup> Jet Propulsion Laboratory (JPL)/California Institute of Technology, United States	<b>An Overview of the Mechanisms and Deployables on the RemoveDebris ADR Mission</b> G. Aglietti <sup>1</sup> , J. Forshaw <sup>1</sup> , A. Viquerat <sup>2</sup> , B. Taylor <sup>1</sup> <sup>1</sup> Surrey Space Centre / University Of Surrey, United Kingdom, <sup>2</sup> Mechanical Engineering Dept / University Of Surrey, United Kingdom
09:00 - 09:30	<b>Recent Developments in Micro Vibrations Measurements</b> E. Cavro <sup>1</sup> , P.-E. Dupuis <sup>1</sup> , F. Vidal-Mata <sup>1</sup> , M. Privat <sup>2</sup> <sup>1</sup> Airbus Defence and Space, France, <sup>2</sup> CNES, France	<b>The ADEO Passive De-Orbit Subsystem Proto Flight Model: Reference Missions &amp; PFM Design</b> T. Sinn <sup>1</sup> , L. Tiedemann <sup>1</sup> , A. Riemer <sup>2</sup> , R. Hahn <sup>2</sup> , T. Sproewitz <sup>3</sup> , P. Seefeldt <sup>3</sup> , M. Sznajder <sup>3</sup> , S. Reershemius <sup>3</sup> , S. Meyer <sup>4</sup> , M. Zander <sup>4</sup> , K. Dietrich Bunte <sup>5</sup> , S. Weikert <sup>6</sup> , A. Wiegand <sup>6</sup> , T. Cardone <sup>7</sup> , D. Teti <sup>7</sup> <sup>1</sup> HPS GmbH, Germany, <sup>2</sup> HTS GmbH, Germany, <sup>3</sup> DLR German Aerospace Center - Institute of Space Systems, Germany, <sup>4</sup> DLR German Aerospace Center - Institute of Composite Structures and Adaptive Systems, Germany, <sup>5</sup> Etamax Space GmbH, Germany, <sup>6</sup> Astos Solutions GmbH, Germany, <sup>7</sup> ESA-ESTEC, The Netherlands

<p><b>Acoustic Test Predictions and Correlation for the SGE0 Satellite Platform PFM Model</b></p> <p><i>B. Martínez-Calvo<sup>1</sup>, N. Riva<sup>1</sup>, H. Segelke<sup>1</sup></i>  <sup>1</sup>OHB System AG, Germany</p>	<p><b>ALM Structures Design Process Development</b></p> <p><i>T. Benedicto Rinaudo<sup>1</sup>, R. Caujolle<sup>1</sup>, S. Roger<sup>1</sup></i>  <sup>1</sup>Airbus Defence &amp; Space, France</p>	
<p><b>Acoustic Test Data Analysis &amp; Signal Processing</b></p> <p><i>A. Kiley<sup>1</sup>, E. Collavo<sup>1</sup>, G. Labruyere<sup>2</sup>, G. Rodrigues<sup>2</sup>, M. Such<sup>2</sup></i>  <sup>1</sup>Airbus Defence &amp; Space, United Kingdom, <sup>2</sup>ESA-ESTEC, Holland</p>	<p><b>Development and Test of a Two-Mirror Telescope Using Additive Manufacturing Technology</b></p> <p><i>M. Thiel<sup>1</sup>, T. Sedlmaier<sup>1</sup>, J.-B. Volatier<sup>1</sup>, L. Pambagian<sup>2</sup>, S. Brinkers<sup>3</sup>, W. Crowcombe<sup>3</sup>, H. Oosterling<sup>3</sup>, F. van Kempen<sup>3</sup>, A. Hoogstrate<sup>3</sup>, P. Kerkhof<sup>3</sup>, C. Aumund-Kopp<sup>4</sup>, M. Mulser<sup>4</sup>, G. Hilfer<sup>5</sup>, S. Klein<sup>5</sup>, T. Hayo<sup>5</sup>, T. Domagala<sup>6</sup>, S. Kwast<sup>7</sup>, M. Eggens<sup>7</sup></i>  <sup>1</sup>OHB System AG, Germany, <sup>2</sup>ESA-ESTEC, Netherlands, <sup>3</sup>TNO, Netherlands, <sup>4</sup>Fraunhofer IFAM, Germany, <sup>5</sup>IABG, Germany, <sup>6</sup>Materialise, Germany, <sup>7</sup>SRON, Netherlands</p>	
<b>COCKTAIL SPACE EXPO</b>		

EINSTEIN	ERASMUS	ESCAPE
<b>Acoustic Analysis - 1</b>	<b>Additive Manufacturing - Design 1</b>	
Chair: N. Riva - I. Ngan	Chair: S. Legrand - A. Makaya	
<p><b>Investigating Diffusivity of Virtual Diffuse Field Acoustic Test (DFAT) Using Boundary Element Modeling and Wavenumber-Frequency Analysis</b></p> <p><i>B. Gardner<sup>1</sup>, A. Castel<sup>2</sup>, C. Musser<sup>1</sup>, A. Medeiros<sup>1</sup>, L. Alimonti<sup>1</sup></i>  <sup>1</sup>ESI Group, United States, <sup>2</sup>ESI NA, United States</p>	<p><b>Design of a Gyroscope Support Structure Manufactured by Selective Laser Melting</b></p> <p><i>T. Glaser<sup>1</sup>, O. Mierheim<sup>1</sup>, C. Hühne<sup>1</sup></i>  <sup>1</sup>German Aerospace Center, Germany</p>	
<p><b>Full Frequency Band Vibro-Acoustic Analysis of a Sandwich Composite Structure</b></p> <p><i>A. Caillet<sup>1</sup>, D.-O. Lee<sup>2</sup>, Y.-K. Lee<sup>2</sup>, A. Grillenbeck<sup>3</sup>, T. Lechelmayr<sup>3</sup></i>  <sup>1</sup>ESI GmbH, Germany, <sup>2</sup>ADD, Republic of Korea, <sup>3</sup>Space Centre IABGmbH, Germany</p>	<p><b>Topology Optimisation of a PCB Supporting Frame</b></p> <p><i>I. Liémans<sup>1</sup>, J. Garnier<sup>1</sup>, J. Polome<sup>1</sup>, D. Garray<sup>2</sup>, S. Vermeir<sup>2</sup></i>  <sup>1</sup>Thales Alenia Space in Belgium, Belgium, <sup>2</sup>Sirris, Belgium</p>	

09:30 - 10:00	<b>Test Facility for Microvibration Requirements Verification</b> Mr Santiago Pasalodos <sup>1</sup> , Mr Enrique del Castillo <sup>1</sup> , Mr Marcos Ubierna <sup>2</sup> , Mr Luis Pascual <sup>1</sup> , Mr Jose Ignacio Bueno <sup>2</sup> <sup>1</sup> SENER, Spain	<b>Gossamer Solar Array Concepts</b> T. Sproewitz <sup>1</sup> , P. Seefeldt <sup>1</sup> , C. Grimm <sup>1</sup> , R. Jahnke <sup>1</sup> , N. Reininghaus <sup>3</sup> , K. Sasaki <sup>1</sup> , M. Sznajder <sup>1</sup> , P. Spietz <sup>1</sup> , S. Reershemius <sup>1</sup> , H. Martens <sup>1</sup> , M. Hillebrandt <sup>2</sup> , S. Meyer <sup>2</sup> , N. Toth <sup>1</sup> , M. Vehse <sup>3</sup> , T. Wippermann <sup>1</sup> , M. Zander <sup>2</sup> , J.-T. Grundmann <sup>1</sup> <sup>1, 2, 3</sup> German Aerospace Center, Germany,
10:00 - 10:30	<b>A Methodology for Disturbance Characterisation of Families of Microvibration Sources</b> <u>S. De Lellis</u> <sup>1</sup> , A. Stabile <sup>1</sup> , G. S. Aglietti <sup>1</sup> , G. Richardson <sup>2</sup> <sup>1</sup> University of Surrey, United Kingdom, <sup>2</sup> Surrey Satellite Technology Limited (SSTL), UK	<b>An Engineering Investigation on Deployable Rigid Solar Array Inspired by Miura Origami</b> M. Li, Q. Cui, L. Zhang <sup>1</sup> Aerospace System Engineering Shanghai, China
<b>10:30 - 11:00</b>	<b>BREAK</b>	
	<b>NEWTON 1</b>	<b>NEWTON 2</b>
<b>11:00 - 13:00</b>	<b>Micro-Vibration Characterisation / Modelling / Isolation</b>	<b>Deployable Structures - 2</b>
	Chair: G. Aglietti - G. Smet	Chair: O. Mierheim - M. Such Taboada
11:00 - 11:30	<b>Investigation of Parameters Affecting the Microvibration Signature of Reaction Wheel Bearings</b> <u>C. Hodge</u> <sup>1</sup> , A. Stabile, G. Aglietti, A. Wade, G. Richardson <sup>1</sup> University Of Surrey, United Kingdom	<b>Scalability of Stable Deployable Trusswork Mast</b> C. Cardoso <sup>1</sup> , J. Loureiro <sup>1</sup> , I. Costa <sup>1</sup> , L. Moreira <sup>2</sup> , J. Rodrigues <sup>2</sup> , C. Ashcroft <sup>3</sup> , L. Puig <sup>3</sup> <sup>1</sup> HPS Lda, Portugal, <sup>2</sup> INEGI, Portugal, <sup>3</sup> ESA-ESTEC, Nederlands
11:30 - 12:00	<b>Microvibrations Modelling and Measurement on Sentinel 4 UVN Calibration Assembly Using Piezoelectric 6 Component Force Dynamometers</b> <u>J. Charet</u> <sup>1</sup> , J. Jacobs <sup>2</sup> <sup>1</sup> Kistler Instrument, Switzerland, <sup>2</sup> Centre Spatial de Liège, Belgium	<b>Conceptual Design of the Deployable Booms for the GOSOLAR-Satellite</b> M. Hillebrandt <sup>1</sup> , M. Zander <sup>1</sup> , C. Huehne <sup>1</sup> <sup>1</sup> Dlr - German Aerospace Center, Germany
12:00 - 12:30	<b>Mechatronic Design of an Active Micro-Vibration Isolation Platform Utilizing Magnetic Levitation</b> <u>Z. Gong</u> <sup>1</sup> , L. Ding <sup>1</sup> , H. Yue <sup>1</sup> , F. Yang <sup>1</sup> , R. Liu <sup>1</sup> , P. Xu <sup>1</sup> <sup>1</sup> Harbin Institute Of Technology, China	<b>Design and Inflatable Testing in Space for Gravity-Gradient Boom</b> <u>J. Wei</u> <sup>1</sup> <sup>1</sup> Harbin Institute of Technology, China
12:30 - 13:00	<b>Vibration Isolator for Cryogenic Machines</b> G. Carte <sup>1</sup> , <u>T. Demerville</u> <sup>2</sup> <sup>1</sup> Thales Alenia Space - France, France, <sup>2</sup> SMAC, France	<b>Exploring the Behavior of Self-Deployable Composite Booms Using High Definition Fiber Optic Sensing</b> N. A. Abdul Rahim <sup>1</sup> , K. Cox <sup>2</sup> , M. Davis <sup>1</sup> <sup>1</sup> Luna Innovations, United States, <sup>2</sup> Roccor, United States
<b>13:00 - 14:00</b>	<b>LUNCH</b>	

<p><b>Assessment on Acoustic Test Effectiveness in Reverberant Chamber via Analysis by Wave Number Transform on Normalized Cross Spectrum Density</b>  <u>D. Todaka</u><sup>1</sup>  <sup>1</sup>Japan Aerospace Exploration Agency, Japan</p>	<p><b>Bio-inspired Bracket in Additive Manufacturing: Umbilical Connector Bracket</b>  <i>G. Cosío M<sup>1</sup>, Vilanova J<sup>2</sup>, Lasagni F<sup>3</sup></i>  <sup>1</sup>Citd Engineering &amp; Technologies, Madrid, Spain, <sup>2</sup>Airbus D&amp;S, Madrid, Spain, <sup>3</sup>CATEC, Sevilla, Spain</p>	
<p><b>Acoustic Analysis of Structures with Local Nonlinear Behaviour</b>  <u>C. Puillet</u><sup>1</sup>  <sup>1</sup>CNES, France</p>	<p><b>Innovative Microsatellite Design Using Additive Manufacturing and Optimization Technics for its Structure</b>  <i>L. Ruiz De Ocenda<sup>1</sup>, A. Piquereau<sup>1</sup>, K. Pfaab<sup>1</sup>, B. Millord<sup>2</sup></i>  <sup>1</sup>Cnes, France, <sup>2</sup>SOGECLAIR Aerospace, France</p>	
<b>BREAK</b>		
<b>EINSTEIN</b>	<b>ERASMUS</b>	<b>ESCAPE</b>
<b>Acoustic Analysis -2</b>	<b>Additive Manufacturing - Process</b>	
Chair: Q. Shi - N. Eaton	Chair: G. Pommatau - L. Pambaguijan	
<p><b>Recovering Output from Vibro-Acoustic Analyses Containing Condensed Models</b>  <u>A. Vargalui</u><sup>1</sup>, I. Ngan<sup>1</sup>  <sup>1</sup>ESA-ESTEC, Netherlands</p>	<p><b>Impact of LBM Process Defects on Mechanical Properties of AS7G06 Aluminium Alloy</b>  <i>O. Quenard<sup>1</sup>, P. Guy<sup>1</sup>, S. Begoc<sup>2</sup></i>  <sup>1</sup>Icam, France, <sup>2</sup>CNES, France</p>	
<p><b>Efficient Matrix Randomisation Methodology for Reduced Spacecraft Models in Stochastic FEM-BEM Vibroacoustic Problems</b>  <u>V. Yотов</u><sup>1</sup>, M. Remedja<sup>1</sup>, G. Aglietti<sup>1</sup>,  G. Richardson<sup>2</sup>  <sup>1</sup>Surrey Space Centre, United Kingdom,  <sup>2</sup>Surrey Satellite Technology Ltd., United Kingdom</p>	<p><b>In-Situ Process Monitoring in Metal Powder Bed Fusion Processes by Means of Multi-Sensor Data Mining Methods</b>  <i>M. Grasso<sup>1</sup>, B. Maria Colosimo<sup>1</sup></i>  <sup>1</sup>Politecnico Di Milano, Italy</p>	
<p><b>Analysis of IXV Space Hardware Exposed to Acoustic Diffuse Random Field</b>  <u>S. Destefanis</u><sup>1</sup>, A. Talbot<sup>2</sup>  <sup>1</sup>Thales Alenia Space Italy, Italy, <sup>2</sup>Free Field Technologies, Belgium</p>	<p><b>Silicon Nitride Ceramic Development for Disruptive Space Applications</b>  <i>N. Rousselet<sup>1</sup>, S. Schweizer<sup>1</sup>, P. Grasset<sup>2</sup>, K. Brethome<sup>3</sup></i>  <sup>1</sup>DCeram, France, <sup>2</sup>Thales Alenia Space, France, <sup>3</sup>CNES, France</p>	
<p><b>Comparison of Stress Evaluation Approaches under Acoustic Loads</b>  <u>R. Olympia</u><sup>1</sup>, V. Mariathasan<sup>1</sup>  <sup>1</sup>Airbus Defence And Space, Germany</p>	<p><b>Realization of Smart Components with Embedded Electronics by using Fused Filament Fabrication</b>  <i>G. Graterol Nisi<sup>2</sup>, M. Eugeni<sup>1</sup>, V. Cardini<sup>1</sup>, S. Atek<sup>2</sup>, M. Pasquali<sup>1</sup>, P. Gaudenzi<sup>1</sup></i>  <sup>1</sup>Università Di Roma La Sapienza, Italy, <sup>2</sup>Smart Structures Solution, Italy</p>	
<b>LUNCH</b>		

	NEWTON 1	NEWTON 2
<b>14:00 - 16:00</b>	<b>Micro-Vibration Isolation / Control</b>	<b>Deployable Structures - 3</b>
	Chair: E. Standarovski - P. Gaudenzi	Chair: E. Pfeiffer - D. Ljubicic
14:00 - 14:30	<b>Characterisation of a Novel 2-Collinear-DoF Strut for Micro-Vibration Mitigation</b> <i>A. Stabile<sup>1</sup>, G. S. Aglietti<sup>1</sup>, G. Richardson<sup>2</sup>, G. Smet<sup>3</sup></i> <sup>1</sup> Surrey Space Centre, United Kingdom, <sup>2</sup> Surrey Satellite Technology Ltd, United Kingdom, <sup>3</sup> ESA-ESTEC, Netherlands	<b>Modular Deployable Structures Demonstrators</b> <i>J. Nieto<sup>1</sup>, J. Fayos<sup>1</sup>, Á. Pipó<sup>2</sup>, C. Montesano<sup>3</sup>, J. Santiago-Prowald<sup>4</sup></i> <sup>1</sup> Comet Ingeniería S.L., Spain, <sup>2</sup> Proxix Engineering, Spain, <sup>3</sup> Airbus D&S Space Systems España, Spain, <sup>4</sup> ESA-ESTEC, The Netherlands
14:30 - 15:00	<b>Vibration Damping of TALC in the Deployed Configuration: an Experimental Demonstration on a 1/10 Test Bench</b> <i>C. Collette<sup>1</sup>, A. Pece<sup>1</sup>, G. Durand<sup>2</sup>, S. Chesnè<sup>3</sup></i> <sup>1</sup> Université Libre De Bruxelles, Belgium, <sup>2</sup> CEA Saclay, France, <sup>3</sup> INSA-Lyon, France	<b>LAGARD: Breadboard Testing Towards an 11-Meter Deployable Stable Truss Structure</b> <i>D. E. Vlachos<sup>1,2</sup>, D. Lamprou<sup>1</sup>, A. I. Vavouliotis<sup>1,2</sup>, A. Kotzakolios<sup>1,2</sup>, F. Anagnostidis<sup>1</sup>, V. Baras<sup>1</sup>, E. C. Kaslis<sup>2</sup>, V. Kostopoulos<sup>2</sup>, L. Puig<sup>3</sup>, C. Ashcroft<sup>4</sup></i> <sup>1</sup> Adamant Composites Ltd, Greece, <sup>2</sup> Applied Mechanics Laboratory, Mech. Engineering and Aeronautics Dept., Greece, <sup>3</sup> European Space Research and Technology Centre (ESA-ESTEC), The Netherlands, <sup>4</sup> European Space Research and Technology Centre (ESA-ESTEC), The Netherlands
15:00 - 15:30	<b>Tuned Mass Dampers for Space Applications</b> <i>G. Carte<sup>1</sup>, T. Demerville<sup>2</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> SMAC, France	<b>Tensegrity Diaphanous Dome Demonstrator</b> <i>J. Fayos<sup>1</sup>, J. Nieto<sup>1</sup>, Á. Pipó<sup>2</sup>, J. Santiago-Prowald<sup>3</sup></i> <sup>1</sup> Comet Ingeniería S.L., Spain, <sup>2</sup> Proxix Engineering, Spain, <sup>3</sup> ESA-ESTEC, The Netherlands
15:30 - 16:00	<b>Method of Adjusting the Resonant Frequencies by Placing Vibration Isolator Locations to Avoid Harmful Resonances below Cut Off Frequency</b> <i>T.Kajikawa<sup>1</sup>, Y. Hyakusoku<sup>1</sup>, D. Todaka<sup>1</sup>, Q. Shi<sup>1</sup></i> <sup>1</sup> JAXA, Sengen, Japan	<b>Deployment Characteristics of the Composite Parabolic Reflector with Zero-Gravity Test</b> <i>J. Roh<sup>1</sup>, S. Ho Chae<sup>1</sup>, Y.-E. Oh<sup>1</sup>, S.-Y. Lee<sup>1</sup></i> <sup>1</sup> Korea Aerospace University, South Korea
<b>16:00 - 16:30</b>	<b>BREAK</b>	
<b>16:30 - 18:30</b>	<b>Vibration Isolation / Damping</b>	<b>Deployable Structures - 4</b>
	Chair: G. Carte - R. Knockaert	Chair: J. Nieto - M. Palladino
16:30 - 17:00	<b>Hybrid Isolatior for Space Applications</b> <i>Lafarga Nebot V<sup>1</sup>, Gadinho V<sup>1</sup>, Zhao G<sup>1</sup>, Rodrigues G<sup>2</sup>, Collette C<sup>1</sup></i> <sup>1</sup> Université Libre De Bruxelles, Belgium, <sup>2</sup> ATG-Europe B.V, The Netherlands	<b>Articulated Boom Development for Large Deployable Reflectors</b> <i>S. Endler<sup>1</sup>, D. Hartmann<sup>1</sup>, A. Riemer<sup>2</sup>, T. Sproewitz<sup>3</sup>, L. Carlos Moreira<sup>4</sup>, A. Ihle<sup>5</sup></i> <sup>1</sup> HPS GmbH, Germany, <sup>2</sup> HTS GmbH, Germany, <sup>3</sup> DLR, Germany, <sup>4</sup> INEGI, Portugal, <sup>5</sup> ESA-ESTEC, The Netherlands

EINSTEIN	ERASMUS	ESCAPE
<b>Static Testing</b>  Chair: J. Martin - R. Knockaert	<b>Additive Manufacturing - Material Characterisation 1</b>  Chair: F. Bruckner - A. Brandao	<b>WORKSHOP</b> <b>Direct Field Acoustic Testing</b> <b>14.00- 17.00</b>
<b>Static Load Test Qualification of a Geostationary Spacecraft Primary Structure</b> <i>A. Nippe<sup>1</sup>, G.Bianchi<sup>1</sup></i> <sup>1</sup> OHB System AG, Germany	<b>Study on the Effect of Temperature on the Mechanical Properties of Powder Bed Fused INCONEL 718</b> <i>M. Sprengel<sup>1</sup>, A. Baca<sup>2</sup>, J. Gumpinger<sup>2</sup>, A. Brandao<sup>2</sup>, T. Ghidini<sup>2</sup></i> <sup>1</sup> ESA-ECSAT, United Kingdom, <sup>2</sup> ESA-ESTEC, The Netherlands	
<b>Development and Qualification of the Primary Structure of Orion-MPCV European Service Module</b> <i>P. Palmieri<sup>1</sup>, L. Rutigliano<sup>1</sup>, S. Ottaviano<sup>1</sup>, D. Mioche<sup>2</sup>, G. Di Vita<sup>3</sup>, L. J. Ghosn<sup>4</sup>, T. L. Wallen<sup>5</sup></i> <sup>1</sup> Thales Alenia Space Italy, Italy, <sup>2</sup> Ariane Group, France, <sup>3</sup> ESA-ESTEC, The Netherlands, <sup>4</sup> NASA, USA, <sup>5</sup> Lockheed Martin Corporation (LMCO), USA	<b>Fatigue Properties and Material Characteristics of AM AISi10Mg: Effect of the Contour Parameter on the Microstructure, Density and Mechanical Properties</b> <i>E. Beevers<sup>1</sup>, A. Brandão<sup>1</sup>, J. Gumpinger<sup>1</sup>, M. Gschweil<sup>2</sup>, C. Seyfert<sup>3</sup>, T. Ghidini<sup>1</sup></i> <sup>1</sup> ESA-ESTEC, The Netherlands, <sup>2</sup> RUAG Schweiz AG, Switzerland, <sup>3</sup> EOS GmbH Electro Optical Systems, Germany	
<b>Test-Model Correlation of an AFP Full-Scale Demonstrator</b> <i>P. Mas<sup>1</sup></i> <sup>1</sup> Ariane Group, France	<b>Structural Damping of Additively Manufactured Structures - LEROS Engine Support</b> <i>M. Ferrari<sup>1</sup>, M. Gschweil<sup>1</sup>, P. Stiles<sup>2</sup>, N. Solway<sup>2</sup></i> <sup>1</sup> Ruag Schweiz Ag, Switzerland, <sup>2</sup> Nammo Westcott Ltd, United Kingdom	
<b>Damage Detection in Structures Using a Force Identification Algorithm Based on Transmissibility</b> <i>N. Maia<sup>1</sup>, M. Neves<sup>1</sup></i> <sup>1</sup> LAETA/IDMEC/Instituto Superior Tecnico, Portugal	<b>Material Characterization of Additively Manufactured PA12 and Design of Multifunctional Satellite Structures</b> <i>Hümbert S<sup>1</sup>, Springer P<sup>2</sup>, Lengowski M<sup>3</sup>, Sakraker Özmen I<sup>1</sup>, Gleixner L<sup>1</sup>, Arce E<sup>1</sup></i> <sup>1</sup> Deutsches Zentrum Für Luft- Und Raumfahrt e.V., Germany, <sup>2</sup> Fraunhofer-Institut für Produktionstechnik und Automatisierung IPA, Germany, <sup>3</sup> Universität Stuttgart Institut für Raumfahrtssysteme (IRS), Germany	
<b>BREAK</b>		
<b>Metrology for Spacecraft Testing</b>  Chair: D. Veal - G. Casarosa	<b>Additive Manufacturing - Industrialisation</b>  Chair: T. Glaser - B. Bonvoisin	
<b>The Future of Spacecraft Metrology</b> <i>JT Janssen<sup>1</sup></i> <sup>1</sup> National Physical Laboratory, United Kingdom	<b>Mechanical Properties of Surface Engineered Metallic Parts Prepared by Additive Manufacturing</b> <i>N. Stelzer<sup>1</sup>, M. Scheerer<sup>1</sup>, L. Baca<sup>1</sup>, Z. Simon<sup>1</sup>, T. Sebald<sup>2</sup>, H. Gschiel<sup>3</sup>, M. Hatzenbichler<sup>3</sup>, B. Bonvoisin<sup>4</sup></i> <sup>1</sup> Aerospace & Advanced Composites Gmbh, Austria, <sup>2</sup> Ariane Group GmbH, Germany, <sup>3</sup> FOTEC Forschungs- und Technologietransfer GmbH, Austria, <sup>4</sup> ESA-ESTEC, The Netherlands	

17:00 - 17:30	<b>Elastomer Snubbers Sizing Aiming to Reduce the Dynamic Displacements of a Large Deployable External Appendage</b> <i>C. Rouzée<sup>1</sup>, P. Camarasa<sup>1</sup></i> <sup>1</sup> Airbus Defence And Space, France	<b>The Tape Spring Hinge Deployment System of the Eu:CROPIS Solar Panels</b> <i>O. Mierheim<sup>1</sup>, T. Glaser<sup>1</sup>, C. Hühne<sup>1</sup>, C. Hobbie<sup>2</sup>, S. Kottmeyer<sup>2</sup></i> <sup>1</sup> German Aerospace Center DLR, Germany, <sup>2</sup> German Aerospace Center DLR, Germany
17:30 - 18:00	<b>Piezoelectric Shunt Damping of Space Telescopes Deformable Mirrors under Launching Loads</b> <i>D. Alaluf<sup>1,2</sup>, B. Mokrani<sup>3</sup>, K. Wang<sup>1</sup>, A. Preumont<sup>1</sup></i> <sup>1</sup> Active Structures Laboratory - Université Libre de Bruxelles, Belgium, <sup>2</sup> ESA-ESTEC, The Netherlands, <sup>3</sup> Department of Mechanical, Materials and Aerospace Engineering - University of Liverpool, United Kingdom	<b>Design and Analysis of Flexible Hinge Used for Unfolding Spacecraft Solar Panels</b> <i>J. Zhang, K. Yan, Z. Kou</i> <sup>1</sup> Taiyuan University Of Technology, China
18:00 - 18:30	<b>Determination of Optimum Vibration Isolator Properties for Payload Vibration Isolation</b> <i>B. Karaman<sup>1</sup>, G. ÖZGEN</i> <sup>1</sup> Roketsan, Turkey	<b>Development of a Deployable Composite Helical Antenna</b> <i>Q. Chen<sup>1</sup></i> <sup>1</sup> Shanghai Ys Information Technology Co., Ltd., China
18:30 - 20:30	<b>POSTER SESSION - 1</b>	

## Day 3 - Wednesday 30 May

	NEWTON 1	NEWTON 2
<b>8.30 - 10.30</b>	<b>Shocks - 1</b>	<b>Vibration Testing</b>
	<b>Chair: B. Brevart - S. Kiryenko</b>	<b>Chair: A. Kommer - T. Glaser</b>
08:30 - 09:00	<b>Improvement of Shock Specification and Representativeness for Tests at Subsystem Level</b> <i>E. Niemczyk<sup>1</sup>, J. D'Amico<sup>1</sup>, S. Behar-Lafenêtre<sup>1</sup>, E. Raynal<sup>2</sup>, C. Puillet<sup>2</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> Centre National des Etudes Spatiales, France	<b>Spacecraft Equipment Testing: Methodology to Reduce Overtesting</b> <i>C. Knight<sup>1</sup>, M. Remedia<sup>1</sup>, G. S. Aglietti<sup>1</sup></i> <sup>1</sup> Surrey Space Centre, United Kingdom
09:00 - 09:30	<b>Feedback from an Equipment Supplier on Shocks</b> <i>L. Jamot<sup>1</sup>, E. Carrié<sup>1</sup>, G. Marque<sup>1</sup></i> <sup>1</sup> Sodern, France	<b>An Extended Mass Operator Method Within James-Web-Telescope Vibration Tests</b> <i>M. Jentsch<sup>1</sup>, M. Schatz<sup>1</sup>, R. Olympio<sup>1</sup>, W. Konrad<sup>1</sup></i> <sup>1</sup> Airbus Defence And Space GmbH, Germany
09:30 - 10:00	<b>Improved Pyrotechnic Shock Scaling Method Based on Shock Response Spectrum and Statistical Energy Analysis</b> <i>J. Ho-Jin Hwang<sup>1</sup>, J. Fernandez<sup>1</sup>, G. Borello<sup>2</sup></i> <sup>1</sup> Jet Propulsion Laboratory, United States, <sup>2</sup> InterAC, France	<b>Estimation of Effective Mass and Interface Loads by a Complementary Vibration Test</b> <i>R. Arena<sup>1</sup>, M. Giuliano<sup>1</sup>, N. Girault<sup>1</sup>, N. Roy<sup>2</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> Top Modal, France

<p><b>Recent Developments in Traceable Thermometry at NPL</b>  <u>J. Pearce<sup>1</sup></u>, G. Sutton<sup>1</sup>, R. Simpson<sup>1</sup>,  G. Machin<sup>1</sup>  <sup>1</sup>National Physical Laboratory, United Kingdom</p>	<p><b>Off the shelf AM Cleats: the Solution to Local Thermoelastic Stresses and Optimized Load Distribution</b>  <u>L. Hernandez<sup>1</sup></u>, M. G. Cosío<sup>1</sup>  <sup>1</sup>Citd Engineering &amp; Technologies, Spain</p>	
<p><b>The Development of a Deployment Tracking System</b>  <u>M. Robroek<sup>1</sup></u>, M. Ellenbroek<sup>1</sup>  <sup>1</sup>Airbus Defence And Space Netherlands B.V., Netherlands</p>	<p><b>Probabilistic Assessment of Additive Manufacturing Parts in the Presence of Manufacturing Defects</b>  <u>S. Romano<sup>1</sup></u>, S. Beretta<sup>1</sup>, M. Gschweil<sup>2</sup>,  J. Gumpinger<sup>3</sup>, T. Ghidini<sup>3</sup>  <sup>1</sup>Politecnico Di Milano, Italy, <sup>2</sup>RUAG Space, Switzerland, <sup>3</sup>ESA-ESTEC, Netherlands</p>	
<p><b>Test Baseplate for MeteoSat Third Generation Telescope Optics: Novel and Versatile Design</b>  <u>N. Nava<sup>1</sup></u>, D. López<sup>1</sup>, C. Douville De Franssu<sup>2</sup>  <sup>1</sup>Lidax, Spain, <sup>2</sup>Thales SESO, France</p>	<p><b>Optimal Design of 3D Printed Panel Inserts</b>  <u>M. Schatz<sup>1</sup></u>, R. Schweikle<sup>1</sup>, M. Jentsch<sup>1</sup>,  W. Konrad<sup>1</sup>  <sup>1</sup>Airbus Defence And Space GmbH, Germany</p>	
<b>POSTER SESSION - 1</b>		

EINSTEIN	ERASMUS	ESCAPE
<b>Fatigue</b>	<b>Additive Manufacturing - Verification</b>	
<b>Chair: M. Vorel - G. Sinnema</b>	<b>Chair: D. Carponcin - R. Russel</b>	
<p><b>Numerical Analysis of Initial Crack Propagation under Vibration Fatigue</b>  <u>D. Di Malo<sup>1</sup></u>, F. Magi, I. Sever  <sup>1</sup>University Of Bristol, United Kingdom</p>	<p><b>Establishment of a Verification Methodology for Parts made by Additive Manufacturing</b>  <u>D. Carponcin<sup>1</sup></u>, G. Aridon<sup>1</sup>, G. Milbourn<sup>1</sup>,  B. Bonvoisin<sup>2</sup>, D. Monteiro<sup>2</sup>, L. Pambaguiyan<sup>2</sup>,  K. Cabannes<sup>3</sup>, F. Montredon<sup>3</sup>, M. Palm<sup>4</sup>,  S. Soller<sup>4</sup>, B. Dutton<sup>5</sup>, J. Dawes<sup>5</sup>  <sup>1</sup>Airbus, France / UK, <sup>2</sup>ESA-ESTEC, The Netherlands, <sup>3</sup>Thalès Alenia Space, France, <sup>4</sup>Ariane Group, Germany, <sup>5</sup>MTC, UK</p>	
<p><b>Practical Considerations on Residual Stresses and Impact on Dimensioning</b>  <u>M. Vorel<sup>1</sup></u>, M. Parmar<sup>1</sup>  <sup>1</sup>ArianeGroup, Germany</p>	<p><b>Non Conformance on Additive Manufactured Parts – How to Detect and How to React?</b>  <u>M. Gschweil<sup>1</sup></u>, M. Ferrari<sup>1</sup>  <sup>1</sup>Ruag Space, Switzerland</p>	
<p><b>Determination of Critical Energy Release Rates for Steel-CFRP Interfaces Considering Residual Thermal Stresses</b>  <u>E. Petersen<sup>1</sup></u>, C. Hühne<sup>1</sup>  <sup>1</sup>German Aerospace Center (DLR), Germany</p>	<p><b>Nonlinear Numerical Fracture Analysis of 3d Printed Samples</b>  <u>F. Cecchini<sup>1</sup></u>, M. Rinaldi<sup>1</sup>, G. Oronzo<sup>2</sup>,  F. Lumaca<sup>2</sup>, F. Nanni<sup>1</sup>  <sup>1</sup>University Of Rome Tor Vergata, Italy, <sup>2</sup>Thales Alenia Space Italia, Italy</p>	

10:00 - 10:30	<b>A Novel Computational Shock Prediction Approach Using Hybrid Data-Driven Methodology</b> <i>A. Derkevorkian<sup>1</sup>, A. Kolaini<sup>1</sup>, P. Brewick<sup>2</sup>, S. Masri<sup>2</sup>, J.-S. Pei<sup>3</sup></i> <sup>1</sup> <i>Jet Propulsion Laboratory/ California Institute of Technology, United States,</i> <sup>2</sup> <i>University of Southern California, United States, </i> <sup>3</sup> <i>University of Oklahoma, United States</i>	
<b>10:30 - 11:00</b>	<b>BREAK</b>	
	<b>NEWTON 1</b>	<b>NEWTON 2</b>
11:00 - 13:00	<b>Shocks - 2</b>  <b>Chair: P. Camarasa - C. Puillet</b>	<b>Vibration Testing and Shaker Control</b>  <b>Chair: C. Bisagni - G. Piret</b>
11:00 - 11:30	<b>Shock Release System Classification</b> <i>J. D'amico<sup>1</sup>, S. Behar Lafenêtre<sup>1</sup>, D. Dilhan<sup>3</sup>, E. Raynal<sup>3</sup>, R. Ullio<sup>2</sup>, S. Kiryenko<sup>4</sup></i> <sup>1</sup> <i>Thales Alenia Space, France, </i> <sup>2</sup> <i>Thales Alenia Space, Italy, </i> <sup>3</sup> <i>Centre National d'Études Spatiales, France, </i> <sup>4</sup> <i>ESA-ESTEC, Netherlands</i>	<b>Performance Assessment of 6-DOF Transient Test Experiments on HYDRA</b> <i>M. Appolloni<sup>1</sup></i> <sup>1</sup> <i>ESA-ESTEC, Netherlands</i>
11:30 - 12:00	<b>Shock Predictions</b> <i>N. Coral Gélis<sup>1</sup>, S. Orsingher<sup>1</sup>, E. Raynal<sup>2</sup>, C. Puillet<sup>2</sup></i> <sup>1</sup> <i>Mecano Id, France, </i> <sup>2</sup> <i>CNES, France</i>	<b>Spacecraft Transient Qualification Testing on HYDRA through the Injection of Six-DoF Base Acceleration Inputs Recorded from Coupled Loads Analysis</b> <i>S. Fransen<sup>1</sup>, H. Fischer<sup>1</sup>, M. Appolloni<sup>1</sup>, I. Ngan<sup>1</sup>, R. Knockaert<sup>1</sup>, G. Laduree<sup>1</sup>, R. Bureo Dacal<sup>1</sup></i> <sup>1</sup> <i>ESA-ESTEC, The Netherlands</i>
12:00 - 12:30	<b>New Experimental Test to Characterize Direct Bonding Shock Resistance</b> <i>M. Voisin<sup>1</sup>, A. Maurel-Pantel<sup>1</sup>, F. Lebon<sup>1</sup>, N. Cocheteau<sup>2</sup>, S. Begoc<sup>3</sup></i> <sup>1</sup> <i>LMA, France, </i> <sup>2</sup> <i>Thales SESO, France, </i> <sup>3</sup> <i>CNES, France</i>	<b>Multiple-Shaker Control for Single-Axis Environmental Vibration Testing</b> <i>G. Kleyman<sup>1</sup>, H. Bühmann<sup>2</sup></i> <sup>1</sup> <i>Institute of Vibration and Dynamic Research, Germany, </i> <sup>2</sup> <i>m+p international Mess- und Rechnertechnik GmbH, Germany</i>
12:30 - 13:00	<b>Shock Attenuator For Equipment (SAFE)</b> <i>P. Lamy<sup>1</sup>, T. Demerville<sup>1</sup>, Q. Bianco<sup>1</sup>, S. Kiryenko<sup>2</sup></i> <sup>1</sup> <i>SMAC, France, </i> <sup>2</sup> <i>ESA-ESTEC, The Netherlands</i>	<b>Fast Sine Sweep as an Alternative to Classical Sine Sweep for SC Qualification</b> <i>E. Cavro<sup>1</sup>, N. Roy, A. Girard, P.-E. Dupuis</i> <sup>1</sup> <i>Airbus Defence and Space, France</i>
<b>13:00 - 14:00</b>	<b>LUNCH</b>	

<p><b>New High Cycle Fatigue Test Facility of Adhesively Bonded Ceramic Electronic Components</b></p> <p><i>L. Ben Fekih<sup>1</sup>, O. Verlinden<sup>1</sup>, C. De Fruytier<sup>2</sup>, G. Kouroussis<sup>1</sup></i>  <sup>1</sup>University Of Mons, Belgium, <sup>2</sup>Thales Alenia Space, Belgium</p>	<p><b>Effect of Heat Treatment on the Microstructure and Mechanical Properties of Inconel 718 Alloy Produced by Selective Laser Melting</b></p> <p><i>A. Baca<sup>1</sup>, M. Sprengel<sup>2</sup>, J. Gumpinger<sup>1</sup>, A. Branda<sup>1</sup>, T. Ghidini<sup>1</sup></i>  <sup>1</sup>European Space Research and Technology Centre (ESA-ESTEC), Netherlands, <sup>2</sup>European Space Research and Technology Centre (ESA-ECSAT), United Kingdom</p>	<p><b>WORKSHOP</b>  <b>Metrology for the Future of Space Craft Testing</b>  <b>10.00-13.00</b></p>
<b>BREAK</b>		
<p><b>EINSTEIN</b></p>	<p><b>ERASMUS</b></p>	
<p><b>High Velocity Impact</b></p>	<p><b>Material Strength Testing</b></p>	
<p>Chair: C. Hühne - T. Cardone</p>	<p>Chair: K. Pfaab - N. Stelzer</p>	
<p><b>Mitigating the Effect of Space Small Debris on COPV in Space with Fiber Sensors and Self-Repairing Materials</b></p> <p><i>E. Haddad<sup>1</sup>, Y. Zhao<sup>2</sup>, C. Mert<sup>2</sup>, M. Basti<sup>2</sup>, K. Tagziria<sup>1</sup>, E. Wallach<sup>1</sup>, M. Mena<sup>1</sup>, A. Saffarpour<sup>1</sup>, C. Semprimoschnig<sup>3</sup>, U. Lafont<sup>3</sup>, I. McKenzie<sup>3</sup></i>  <sup>1</sup>MPB Communications Inc., Canada, <sup>2</sup>INRS-EMT, Canada, <sup>3</sup>ESA-ESTEC, The Netherlands</p>	<p><b>Mechanical Testing and Analysis of Hybrid Bonded Joints and Load Introduction Elements</b></p> <p><i>P. Richert<sup>1</sup>, J. Zimmermann<sup>1</sup>, T. A. Schervan<sup>1</sup>, K.-U. Schröder<sup>1</sup></i>  <sup>1</sup>Institute of Structural Mechanics and Lightweight Design, Germany</p>	
<p><b>Numerical Simulation of Composite Deployable Boom Response Subjected to Hypervelocity Impact (HVI)</b></p> <p><i>E. Giannaros<sup>1</sup>, A. Kotzakolios<sup>1</sup>, G. Sotiriadis<sup>1</sup>, S. Tsantzalis<sup>1</sup>, V. Kostopoulos<sup>1</sup>, G. Campoli<sup>2</sup></i>  <sup>1</sup>University of Patras, Greece, <sup>2</sup>ESA-ESTEC, The Netherlands</p>	<p><b>Verification of Failure Criteria for CFRP Composites under Cryogenic Thermo-Mechanical Loading</b></p> <p><i>J. Hohe<sup>1</sup>, S. Fliegener<sup>1</sup>, K.-P. Weiß<sup>2</sup>, S. Appel<sup>3</sup></i>  <sup>1</sup>Fraunhofer IWM, Germany, <sup>2</sup>Karlsruhe Institute of Technology KIT, Germany, <sup>3</sup>ESA-ESTEC, The Netherlands</p>	
<p><b>Analytical Prediction of High-Velocity Impact Resistance of Plane and Curved Thin Woven Fabric Composite Targets</b></p> <p><i>M. Pasquali<sup>1</sup>, P. Gaudenzi<sup>1</sup></i>  <sup>1</sup>Università Di Roma La Sapienza, Italy</p>	<p><b>Innovative Setup for Cryogenic Mechanical Testing of High Strength Metallic Alloys</b></p> <p><i>M. Tufano<sup>1</sup>, C. Zauner<sup>1</sup>, A. Morasch<sup>1</sup></i>  <sup>1</sup>KRP Mechatec GmbH, Germany</p>	
<p><b>Influence of Space Environment on Performance and Robustness of thin shell CFRP-Booms</b></p> <p><i>M. Eckhard Zander<sup>1</sup>, M. Sinapius<sup>2</sup>, C. Hühne<sup>1,2</sup></i>  <sup>1</sup>Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR) German Aerospace Center, Germany, <sup>2</sup>Technische Universität Braunschweig, Deutschland</p>	<p><b>Aerospace Adhesive Testing for Cryogenic Conditions with Small TAST Specimens</b></p> <p><i>K. Friedrich Reiling<sup>1</sup>, F. Altenwegner<sup>2</sup></i>  <sup>1</sup>University Of Applied Sciences Landshut, Germany, <sup>2</sup>SIKA Automotive GmbH, Germany</p>	
<b>LUNCH</b>		

	NEWTON 1	NEWTON 2
<b>14:00 - 16:00</b>	<b>Grid Structures</b>	<b>Thermo-Elastic - 1</b>
	Chair: L. Pavlov - C. Moratto	Chair: P. Corberand - J. D'amico
14:00 - 14:30	<b>Development Logic, Design, Manufacturing and Testing of Pre-Preg Lattice Structures for Satellite Central Cylinder Applications</b> <i>L. Pavlov<sup>1</sup>, B. Smeets<sup>1</sup>, T. Papenhuijzen<sup>1</sup>, M. Koot<sup>1</sup>, A. Doyle<sup>2</sup>, T. Flannagan<sup>2</sup></i> <sup>1</sup> ATG Innovation Ltd., Ireland, <sup>2</sup> ÉireComposites Teo, Ireland	<b>Reliability and Accuracy Improvement for Thermo-Elastic Analyses – Process and Tool Applied on the JUICE Mission</b> <i>M. Bourdeaud'hui<sup>1</sup>, S. Laborde<sup>1</sup>, J. Ponsy<sup>1</sup></i> <sup>1</sup> Airbus DS, France
14:30 - 15:00	<b>Design and Testing of Additively Manufactured Lattice Structures</b> <i>T. Lewis<sup>1</sup>, S. Kébreau<sup>1</sup></i> <sup>1</sup> Airbus Defence and Space GmbH, Germany	<b>Temperature Mapping for Structural Thermo-Elastic Analyses; Method Benchmarking and Application of the SINAS Method to an Optical Payload</b> <i>S. Simonian<sup>1</sup>, D. Dykstra<sup>1</sup></i> <sup>1</sup> ATG Europe, The Netherlands
15:00 - 15:30	<b>General Overview of Gridded Technology Design for Vega C Interstage 2/3</b> <i>A. Zallo<sup>1</sup>, M. Cioeta<sup>1</sup>, F. de Nicola<sup>2</sup>, G. Totaro<sup>2</sup>, G. Giusto<sup>2</sup>, P. Spena<sup>2</sup>, F. Di Caprio<sup>2</sup>, S. Mespoulet<sup>3</sup></i> <sup>1</sup> Avio s.p.a, Italy, <sup>2</sup> CIRA, Italy, <sup>3</sup> ESA-ESRIN, Italy	<b>Thermal Conductor Generation for Thermal and Thermo-Elastic Analysis Using a Finite Element Model and SINAS</b> <i>S. Simonian<sup>1</sup>, D. Dykstra<sup>1</sup></i> <sup>1</sup> ATG Europe, The Netherlands
15:30 - 16:00	<b>Highly Efficient CFRP Tubular Grid Structures for Satellites and Payloads: Design and Manufacturing Method</b> <i>G. Totaro<sup>1</sup>, F. De Nicola<sup>1</sup>, G. Giusto<sup>1</sup>, P. Spena<sup>1</sup>, S. Kiryenko<sup>2</sup></i> <sup>1</sup> Cira - Italian Aerospace Research Center, Italy, <sup>2</sup> ESA-ESTEC, The Netherlands	<b>Integration Of Thermo-Elastic Characteristics In Finite Element Method Reduced Models</b> <i>M. Trucchi<sup>1</sup>, V. Torrelli<sup>1</sup></i> <sup>1</sup> Assystem Technologies, France
<b>16:00 - 16:30</b>	<b>BREAK</b>	
	NEWTON 1	NEWTON 2
<b>16:30 - 19:00</b>	<b>Tanks &amp; COPV's</b>	<b>Thermo-Elastic - 2 - Testing</b>
	Chair: G. Sinnema - M. Voral	Chair: P.E. Dupuis - G. Casarosa
16:30 - 17:00	<b>Mechanical Properties and Microstructure of Next Generation Spacecraft Propellant Tanks Produced via Friction Stir Welding of Ti-6Al-4V</b> <i>M. Meisnar<sup>1</sup>, A. Norman<sup>2</sup>, K. Nor<sup>3</sup>, S. Dodds<sup>3</sup>, R. Freeman<sup>3</sup>, R. Bellarosa<sup>4</sup>, T. Ghidini<sup>2</sup></i> <sup>1</sup> ESA-ECSAT, United Kingdom, <sup>2</sup> European Space Agency, The Netherlands, <sup>3</sup> TWI, United Kingdom, <sup>4</sup> Airbus, United Kingdom	<b>Highly Accurate Thermo-Elastic Distortion Measurement Technique for Antenna Reflectors</b> <i>P.-E. Dupuis<sup>1</sup>, J.-C. Csont<sup>1</sup>, N. Chauvet<sup>1</sup>, B. Palacin<sup>2</sup></i> <sup>1</sup> Airbus Defence & Space, France, <sup>2</sup> CNES, France

EINSTEIN	ERASMUS	ESCAPE
<b>Fibre Optics - 1</b>	<b>Ceramics</b>	
Chair: J. Madrigal - I. McKenzie	Chair: S. Behar-Lafenêtre - M. Such Taboada	
<b>State-Of-The-Art of Research on Distributed Optical Fibre Sensors</b> <i>L. Thévenaz<sup>1</sup>, Z. Yang<sup>1</sup></i> <sup>1</sup> Ecole Polytechnique Fédérale de Lausanne, Switzerland	<b>Ceramic Structures Sizing and Verification Method Improvements - Investigation of Size Effects in Silicon Carbide</b> <i>D. Denaux<sup>1</sup>, D. Logut<sup>1</sup>, M. Such-Taboada<sup>2</sup></i> <sup>1</sup> Airbus, France, <sup>2</sup> ESA-ESTEC, The Netherlands	
<b>Fibre Optic Sensors for Spacecraft Structure In-flight Monitoring</b> <i>S. Abad<sup>1</sup>, J. Ortiz Martín<sup>2</sup>, F. S. Pinto<sup>1</sup>, F M. Araújo<sup>1</sup></i> <sup>1</sup> HBM FiberSensing, Portugal, <sup>2</sup> Airbus DS, Spain	<b>Derivation of Three-Parameter Weibull Distributions for Sizing and Verification of Ceramic Materials</b> <i>U. Barozzi<sup>1</sup>, S. Lucarelli<sup>1</sup>, M. Jentsch<sup>1</sup>, J. Steiner<sup>1</sup>, D. Denaux<sup>2</sup>, M. Such Taboada<sup>3</sup></i> <sup>1</sup> Airbus Defence and Space, Germany, <sup>2</sup> Airbus Defence and Space, France, <sup>3</sup> ESA-ESTEC, The Netherlands	
<b>Structural Health and Usage Monitoring with Fiber Optic Sensors of Unmanned Aircrafts</b> <i>M. Frovel<sup>1</sup>, A. Fernandez<sup>2</sup>, J. M. Pintado<sup>1</sup>, M. A. de la Torre<sup>1</sup>, Dr. Felix Terroba<sup>1</sup>, R. Lopez<sup>1</sup></i> <sup>1</sup> Inta, Spain, <sup>2</sup> ETSIAE, Spain	<b>Tailoring Ceramic Matrix Composites by Multiscale Simulation for Light Weight and Highly Stiff Space Structures</b> <i>J. Schmidt<sup>1</sup>, G. Seifert<sup>1</sup></i> <sup>1</sup> Fraunhofer ISC, Bayreuth, Germany	
<b>Monitoring the Propulsion System of PROBA-2 with Optical Fiber Sensors during 8 Years</b> <i>E. Haddad<sup>1</sup>, M. Mena<sup>1</sup>, R. V. Kruzelecky<sup>1</sup>, K. Tagziria<sup>1</sup>, E. Wallach<sup>1</sup>, F. Ricci<sup>2</sup>, I. McKenzie<sup>3</sup>, N. Karafolas<sup>3</sup>, F. Hannoteau<sup>4</sup></i> <sup>1</sup> MPB Communications Inc., Canada, <sup>2</sup> Xiphos Systems Corp, Canada, <sup>3</sup> Optoelectronics Section, European Space Agency, The Netherlands, <sup>4</sup> European Space Agency, Belgium	<b>Development of Strong Resistance Brazing Joints for Silicon Nitride Ceramic Parts for Space Applications</b> <i>L. Cornillon<sup>1</sup>, P. Grasset<sup>1</sup>, C. Devilliers<sup>1</sup>, S. Behar-Lafenêtre<sup>1</sup>, N. Louh<sup>1</sup>, A.-C. Bravo<sup>2</sup>, T. Campanella<sup>2</sup>, F. Gant<sup>3</sup></i> <sup>1</sup> Thales Alenia Space, France, <sup>2</sup> PMB, France, <sup>3</sup> CNES, France	
<b>BREAK</b>		
EINSTEIN	ERASMUS	ESCAPE
<b>Fibre Optics - 2</b>	<b>Composites</b>	
Chair: L. Thévenaz - N. Karafolas	Chair: M. Flanagan - U. Lafont	
<b>A Review of the State-Of-The-Art on the Capabilities of FBG Technologies for Space Sensing Applications</b> <i>J. M. Alvarez De Con<sup>1</sup></i> <sup>1</sup> Smart Fibres Limited, United Kingdom	<b>Demonstration of Thermoplastic Composites for Space Applications</b> <i>F. Preller<sup>1</sup>, C. Tschepe<sup>1</sup>, A. Doyle<sup>2</sup>, M. Ward<sup>2</sup>, B. Weafer<sup>2</sup>, A. Comer<sup>3</sup>, R. O'Higgins<sup>3</sup>, R. Schlitt<sup>4</sup>, S. Appel<sup>5</sup></i> <sup>1</sup> INVENT GmbH, Germany, <sup>2</sup> ÉireComposites Teo., Ireland, <sup>3</sup> University of Limerick, Ireland, <sup>4</sup> Engineering Services, Germany, <sup>5</sup> ESA-ESTEC, The Netherlands	

17:00 - 17:30	<b>A Comprehensive Tool to Simulate Composite Lay-Ups in Pressure Vessels</b> <i>L. Bizet<sup>1</sup>, K. Mathis<sup>1</sup>, P. Saffré<sup>2</sup>, D. Halm<sup>3</sup>, M. Gueguen<sup>3</sup>, P. Francescato<sup>2</sup></i> <sup>1</sup> CNES, France, <sup>2</sup> Symme, France, <sup>3</sup> Institut P, France	<b>Nanometer-order Displacement Measurement Using Built-in Interferometric Sensor for Dimensional Stability of Telescope Structure</b> <i>K. Kitamoto<sup>1</sup>, T. Kamiya<sup>1</sup>, T. Mizutani<sup>1</sup>, S. Yasuda<sup>1</sup>, R. Shimizu<sup>1</sup></i> <sup>1</sup> Japan Aerospace Exploration Agency, Japan
17:30 - 18:00	<b>A Smearing Technique for the 3D Solid Modeling of Composite Pressure Vessels</b> <i>H. Katajisto<sup>1</sup>, A. Ahvenainen<sup>2</sup>, P. Willems<sup>3</sup></i> <sup>1</sup> Altair Engineering Finland, Finland, <sup>2</sup> Aalto University, Finland, <sup>3</sup> Optimum CPV, Belgium	<b>Novel Technique for Thermal Deformation Test Utilizing Periodic Heating</b> <i>T. Miyazaki<sup>1</sup>, K. Ishimura<sup>2</sup>, Y. Satou<sup>2</sup>, T. Miyashita<sup>1</sup></i> <sup>1</sup> Department of Modern Mechanical Engineering / Waseda University, Japan, <sup>2</sup> Institute of Space and Astronautical Science / Japan Aerospace Exploration Agency, Japan
18:00 - 18:30	<b>Composite Overwrapped Pressure Vessel (COPV) Life Test</b> <i>R. Russell<sup>1</sup>, D. Dawicke<sup>2</sup>, J. Hochhalter<sup>3</sup></i> <sup>1</sup> NASA Kennedy Space Center, United States, <sup>2</sup> Analytical Services and Materials, Inc., United States, <sup>3</sup> NASA Langley Research Center, United States	
18:30 - 19:00		
19:00 - 22:00	<b>CONFERENCE DINNER ESTEC RESTAURANT</b>	

<p><b>Photonic Integrated Optical Combs for Structural Health Monitoring in Space</b></p> <p>S. O Duill<sup>1</sup>, F. Smyth<sup>1</sup>, D. Gutierrez<sup>1</sup>, J. Braddell<sup>1</sup> <sup>1</sup>Pilot Photonics, Ireland</p>	<p><b>New High Performance Thermoplastic Composite with Added Functionalities for 3D Printed Structure for Space Application</b></p> <p>M. Rinaldi<sup>1,2</sup>, L. Pigliaru<sup>3</sup>, F. Lamastra<sup>1,2</sup>, L. Ciccacci<sup>1</sup>, T. Ghidini<sup>3</sup>, F. Nanni<sup>1,2</sup> <sup>1</sup>University Of Rome Tor Vergata, Italy, <sup>2</sup>INSTM, Italy , <sup>3</sup>ESA-ESTEC, The Netherlands</p>	
<p><b>Integrated Photonic Devices for Low Footprint Fiber Sensing Space Applications</b></p> <p>T. Van Leest<sup>1</sup>, R. Evenblij<sup>1</sup>, P. Kat<sup>1</sup> <sup>1</sup>Technobis, Netherlands</p>	<p><b>Next-Generation Metal Matrix Composites for Space Launch Applications</b></p> <p>L. Rollings<sup>1</sup>, S. McDonald<sup>1</sup>, M. Roy<sup>2</sup>, P. Withers<sup>1</sup> <sup>1</sup>School of Materials, University Of Manchester, United Kingdom, <sup>2</sup>School of Mechanical, Aerospace and Civil Engineering, University of Manchester, United Kingdom</p>	
<p><b>Fiber Sensing for Space Applications</b></p> <p>S. Ibrahim<sup>1</sup>, R. McCue<sup>1</sup>, J. O'Dowd<sup>1</sup>, M. Farnan<sup>1</sup>, D. Karabacak<sup>2</sup> <sup>1</sup>FAZ Technology Ltd., Ireland, <sup>2</sup>Fugro Technology B.V., The Netherlands</p>	<p><b>Performance Enhancement of CF Composites under Fatigue and Static Loading by the Addition of Thermoplastic PPS Veil Additives</b></p> <p>A. Ramji<sup>1</sup>, Y. Xu<sup>1</sup>, M. Yasaee<sup>1</sup>, P. Irving<sup>1</sup> <sup>1</sup>Cranfield University, United Kingdom</p>	
<p><b>Regenerated Multicore Fibre Bragg Gratings for Structural Health Monitoring in Harsh Environments</b></p> <p>J. Madrigal<sup>1</sup>, D. Barrera<sup>1</sup>, P. Antonio Calderón<sup>1,2</sup>, S. Sales<sup>1,2</sup> <sup>1</sup>Photonics Research Labs, ITEAM research institute, Universitat Politècnica de València, Spain, <sup>2</sup>Calculation and Monitored Structures CALSENS SL, Universitat Politècnica de València, Spain</p>		
<p><b>CONFERENCE DINNER ESTEC RESTAURANT</b></p>		

# Day 4 - Thursday 31 May

	NEWTON 1	NEWTON 2
08:30 - 10:30	<b>Dynamic Coupling</b>  Chair: P. Palmieri - N. Roy	Antennas  Chair: L. Datashvili - G. Rodrigues
08:30 - 09:00	<b>Proba-3 Dynamic Coupling</b> <i>A. Salio<sup>1</sup></i> <i><sup>1</sup>Airbus Defence And Space, Spain</i>	<b>Recent Development at HPS GmbH for K-band and Q/V-band Reflector Antennas</b> <i>P. Projetti Zolla<sup>1</sup>, T. Sinn<sup>1</sup>, S. Hofer<sup>1</sup>, F. Triberti<sup>1</sup>, O. Reichmann<sup>1</sup>, L. Tiedemann<sup>1</sup>, E. K. Pfeiffer<sup>1</sup>, A. Ihle<sup>2</sup>, J. Santiago Prowald<sup>2</sup></i> <i><sup>1</sup>HPS GmbH, Germany, <sup>2</sup>ESA-ESTEC, The Netherlands</i>
09:00 - 09:30	<b>ACLAD – A New Philosophy for Launcher Coupled Load Analyses</b> <i>M. Trucchi<sup>1</sup></i> <i><sup>1</sup>Assystem Technologies, France</i>	<b>Streamlined Reflectors Antennas for Dual Deployments</b> <i>L. Sanchez Izquierdo<sup>1</sup>, O. Castro Matias<sup>1</sup>, J. L. Pardo Garcia<sup>1</sup></i> <i><sup>1</sup>Airbus, Spain</i>
09:30 - 10:00	<b>Use of Modal Effective Parameters to Improve Structural Optimisation</b> <i>X. Vaquer Araujo<sup>1</sup></i> <i><sup>1</sup>Airbus Defence And Space GmbH, Germany</i>	<b>Compact Feed and Subreflector Assembly</b> <i>J. L. Pardo-garcia<sup>1</sup>, S. McLaren<sup>2</sup>, B. Lopez-Zamora<sup>1</sup>, A. Yarza-Fuentes<sup>1</sup></i> <i><sup>1</sup>Airbus, Spain, <sup>2</sup>Airbus, England</i>
10:00 - 10:30	<b>Norton-Thevenin Receptance Coupling (NTRC) as a Payload Analysis Tool</b> <i>D. Kaufman<sup>1</sup>, S. Gordon, A. Majed</i> <i><sup>1</sup>NASA, United States, <sup>2</sup>Applied Structural Dynamics, United States</i>	<b>METOP SG - MWI Reflectors Evolution from MARFEQ-MADRAS Heritage</b> <i>A. Yarza-fuentes<sup>1</sup>, M. Esteban-Castaño<sup>1</sup>, P. Cortes-Gonzalez<sup>2</sup>, O. Castro-Matias<sup>1</sup>, S. Daddio<sup>3</sup></i> <i><sup>1</sup>Airbus, Spain, <sup>2</sup>CT Ingenieros, Spain, <sup>3</sup>ESA-ESTEC, The Netherlands</i>
10:30 - 11:00	<b>BREAK</b>	

EINSTEIN	ERASMUS	ESCAPE
<b>Thermal</b>	<b>Composites / Thermoplastics</b>	
<b>Chair:</b> S. Carli - R. Peyrou-Lauga	<b>Chair:</b> R. Usinger - M. Hillebrandt	
<b>Thermal Control of the Metop-SG ICI On Ground Calibration Targets</b> <i>A. Murk<sup>4</sup>, D. Döring<sup>1</sup>, D. Winter<sup>1</sup>, M. Bergadà<sup>2</sup>, B. Gimenez Bravo<sup>2</sup>, M. Kotiranta<sup>4</sup>, K. Pike<sup>3</sup>, R. Wylde<sup>3</sup></i> <sup>1</sup> IABG mbH, Germany, <sup>2</sup> Airbus DS Space Systems España, Spain, <sup>3</sup> Thomas Keating Ltd., United Kingdom, <sup>4</sup> University of Bern, Institute of Applied Physics, Switzerland	<b>Future of Thermoplastics in Space</b> <i>U. Lafont<sup>1</sup>, E. Amorim<sup>2</sup>, E. Laurent<sup>3</sup>, C. Semprimoschnig<sup>1</sup></i> <sup>1</sup> TEC-QEE - ESA-ESTEC, The Netherlands, <sup>2</sup> TEC-MSP - ESA-ESTEC, The Netherlands, <sup>3</sup> CNES, France	
<b>Evaluation of Heat Transfer at the Interface of Spacecraft Equipment</b> <i>S. Vandeveldé<sup>1</sup>, A. Dadié<sup>1</sup>, M. Sartor<sup>1</sup></i> <sup>1</sup> Institut Clément Ader, Université de Toulouse, UPS, INSA, ISAE-SUPAERO, MINES-ALBI, CNRS, France	<b>Automated Layup of Thermoplastic Composites for Space Applications</b> <i>K. Doyle<sup>1,2</sup>, J.-B. Deyts<sup>3</sup>, M. Blais<sup>3</sup>, A. Ayuso<sup>4</sup>, P. Lefébure<sup>5</sup>, A. Gilliot<sup>6</sup>, S. Appel<sup>7</sup>, A. Doyle<sup>1</sup></i> <sup>1</sup> EireComposites Teo, Ireland, <sup>2</sup> University of Limerick, Ireland, <sup>3</sup> Ariane Group, France, <sup>4</sup> Airbus DS, Spain, <sup>5</sup> Airbus Group Innovation, France, <sup>6</sup> Suprem SA, Switzerland, <sup>7</sup> ESA-ESTEC, The Netherlands	
<b>Validation of an Analytical Model Describing the Heat Flux Distribution in Load-Bearing CFRP Single-Lap Joints</b> <i>M. Lange<sup>1</sup>, V. Baturkin<sup>2</sup>, C. Hühne<sup>1</sup>, O. Mierheim<sup>1</sup></i> <sup>1</sup> DLR, Germany, <sup>2</sup> DLR, Germany	<b>Modelling and Experimental Investigation of Induction Welding of Thermoplastic Matrix Composites</b> <i>M. Flanagan<sup>1,2</sup>, B. Weaver<sup>1</sup>, K. Doyle<sup>1,4</sup>, A. Doyle<sup>1</sup>, T. Flanagan<sup>1</sup>, R. Canavan<sup>1</sup>, M. Bizeul<sup>1</sup>, M. Ward<sup>1</sup>, B. A.M<sup>5</sup>, Ó Brádaigh C.M<sup>6</sup>, J. Goggins<sup>2,7</sup>, Harrison N.M<sup>3,8</sup></i> <sup>1</sup> EireComposites, Ireland, <sup>2</sup> Civil Engineering, NUI Galway, Ireland, <sup>3</sup> Mechanical Engineering, NUI Galway, Ireland, <sup>4</sup> University of Limerick, Ireland, <sup>5</sup> Department of Engineering, University of Cambridge, UK, <sup>6</sup> Institute for Materials and Processes, School of Engineering, University of Edinburgh, UK, <sup>7</sup> Centre for Marine and Renewable Energy Ireland (MaREI), Ireland, <sup>8</sup> Advanced Manufacturing Research Centre (I-Form), Ireland	
<b>Wide Range Thermal Test Facility for JUICE Large Appendages, Design and Results</b> <i>C. Grodent<sup>1</sup>, T. Thibert<sup>1</sup>, S. Liebecq<sup>1</sup>, B. Marquet<sup>1</sup>, C. Lebranchu<sup>2</sup>, E. Bongers<sup>3</sup>, D. Ifrim<sup>4</sup>, C. Stanica<sup>4</sup>, I. Popa<sup>4</sup>, R. Mihalache<sup>4</sup>, D. Mihai<sup>4</sup>, A. Schnorhk<sup>5</sup></i> <sup>1</sup> Centre Spatial De Liège, Belgium, <sup>2</sup> ADS, France, <sup>3</sup> ADS, The Netherlands, <sup>4</sup> COMOTI, Romania, <sup>5</sup> ESA-ESTEC, The Netherlands	<b>Automated Tape Placement with In-Situ Consolidation of Thermoplastic Composites</b> <i>K. Doyle<sup>1,2</sup>, J. B. Deyts<sup>3</sup>, R. M. O'Higgins<sup>2</sup>, A. J. Comer<sup>2</sup>, S. Appel<sup>4</sup></i> <sup>1</sup> EireComposites Teo, Ireland, <sup>2</sup> University of Limerick, Ireland, <sup>3</sup> Ariane Group, France, <sup>4</sup> ESA-ESTEC, The Netherlands	<b>WORKSHOP</b> <b>Thermo-Mechanical Analysis and Verification</b> <b>10:00 - 13:00</b>
<b>BREAK</b>		

	NEWTON 1	NEWTON 2
<b>11:00 - 13:00</b>	<b>FEM / Damping Modelling</b>	<b>Joints &amp; Struts</b>
	<b>Chair:</b> - B. Fransen - J. Fayos	<b>Chair:</b> C. Tschepe - T. Rohr
11:00 - 11:30	<b>FEM Update for BepiColombo Using Multi-Objective Optimisation and Surrogate Models</b> <i>M. Ribera Vicent<sup>1</sup>, G. Aglietti<sup>2</sup>, M. Remedios<sup>2</sup>, A. Aizpuru Hofmann<sup>2</sup>, A. Kiley<sup>3</sup></i> <sup>1</sup> Imperial College London, United Kingdom, <sup>2</sup> Surrey Space Centre, University of Surrey, United Kingdom, <sup>3</sup> Airbus Defence and Space, United Kingdom	<b>Micro-Pins – The Next Step in Composite to Composite and Metal to Composite Joining</b> <i>N. Sarantinos<sup>1</sup>, V. Kostopoulos<sup>1</sup>, G. Di Vita<sup>2</sup>, G. Campoli<sup>2</sup>, L. Bricout<sup>2</sup></i> <sup>1</sup> AML - Applied Mechanics Laboratory, Greece, <sup>2</sup> ESA-ESTEC, The Netherlands
11:30 - 12:00	<b>Damping-Layout Design Approach with Frequency-Content Control</b> <i>M. Brumati<sup>1</sup>, J. Slavic<sup>2</sup>, M. Boltezar<sup>2</sup></i> <sup>1</sup> Synopta Gmbh, Switzerland, <sup>2</sup> University of Ljubljana, Slovenia	<b>Sentinel-1C&amp;D: Antenna Separation Mechanism for Safe De-Orbiting</b> <i>C. Lausch<sup>1</sup>, P. Pavia<sup>2</sup>, G. Laduree<sup>3</sup>, S. Diel<sup>1</sup>, M. von Alberti<sup>1</sup>, R. Baldassarri<sup>2</sup></i> <sup>1</sup> Airbus Defence & Space GmbH, Germany, <sup>2</sup> Thales Alenia Space, Italy, <sup>3</sup> ESA-ESTEC, The Netherlands
12:00 - 12:30	<b>Assessment of Grain Damping Models for Finite Element Analysis of Solid Rocket Motors</b> <i>F. Mastroddi<sup>1</sup>, C. Riso<sup>1</sup>, S. Fransen<sup>2</sup>, G. Coppotelli<sup>1</sup>, F. Trequattrini<sup>1</sup>, A. De Vivo<sup>3</sup></i> <sup>1</sup> Sapienza University Of Rome, Italy, <sup>2</sup> ESA-ESTEC, The Netherlands, <sup>3</sup> AVIO S.p.A., Italy	<b>SpaceStrut™, Development of a Full CFRP Strut Solution in the Frame of the Horizon2020 SME Instrument</b> <i>F. Ruess<sup>1</sup>, B. Braun<sup>1</sup>, F. von Dungern<sup>2</sup>, C. Tschepe<sup>2</sup>, S. Preussler<sup>3</sup></i> <sup>1</sup> Space Structures GmbH, Germany, <sup>2</sup> INVENT GmbH, Germany, <sup>3</sup> Hightex Verstärkungsstrukturen GmbH, Germany
12:30 - 13:00		<b>Qualification of a New Strut Joint Using an Inward-Spanning-Screw</b> <i>A. Netti<sup>1</sup>, R. Styner<sup>1</sup>, N. Gradwohl<sup>2</sup>, H.-P. Gröbelbauer<sup>2</sup></i> <sup>1</sup> Ruag Space, Switzerland, <sup>2</sup> University of Applied Sciences Northwestern Switzerland, Switzerland
<b>13:00 - 14:00</b>	<b>LUNCH</b>	
	NEWTON 1	NEWTON 2
<b>14:00 - 16:00</b>	<b>Tools for Verification &amp; Analysis</b>	<b>Insert Testing</b>
	<b>Chair:</b> J. Fatemi - D. Jaredson	<b>Chair:</b> S. Kögl - M. Such Taboada
14:00 - 14:30	<b>Time and Cost Reduction for Mechanical Analyses and Tests – Tool for Mechanical Design Validation, Justification, and Testing</b> <i>S. L. Sanchez<sup>1</sup>, J. Ponsy<sup>1</sup>, J.-F. Pinazza<sup>1</sup>, G. Zègre<sup>1</sup></i> <sup>1</sup> Airbus, France	<b>Fatigue Assessment on GreDom Potted Inserts</b> <i>A. Netti<sup>1</sup>, S. Kögl<sup>2</sup>, H.-P. Gröbelbauer<sup>3</sup></i> <sup>1</sup> Ruag Space, Switzerland, <sup>2</sup> KOEGL Space, Switzerland, <sup>3</sup> University of Applied Sciences Northwestern Switzerland, Switzerland

EINSTEIN	ERASMUS	ESCAPE
<b>Thermal Testing - 1</b>	<b>Advanced Materials - 1</b>	<b>WORKSHOP</b> <b>Thermo-Mechanical Analysis and Verification</b> 10:00 - 13:00
<b>Chair: A. Eisenmann - R. Messing</b>  <b>The Aeolus Spacecraft TVAC Full Qualification at Centre Spatial de Liège Premises: Description and Challenges.</b> <i>I. Domken, C. Grodent<sup>1</sup>, I. Tychon<sup>1</sup>, S. Liebecq<sup>1</sup>, V. Samain<sup>1</sup>, R. Wimmer<sup>2</sup>, M. Davidson<sup>2</sup>, J.-C. Barthes<sup>3</sup>, O. Lecrenier<sup>3</sup>, A. Elfving<sup>4</sup></i> <sup>1</sup> Centre Spatial De Liège, Belgium, <sup>2</sup> Airbus Defence and Space, United Kingdom, <sup>3</sup> Airbus Defence and Space, France, <sup>4</sup> ESA-ESTEC, The Netherlands	<b>Development and Design of Multifunctional Composite Structures for Satellite Applications</b> <i>S. Perfetto<sup>1</sup>, M. Schubert<sup>2</sup>, D. Mayer<sup>1</sup>, A. Dafnis<sup>2</sup>, Heiko Atzrodt<sup>1</sup></i> <sup>1</sup> Fraunhofer Institute for Structural Durability and System Reliability LBF, Germany, <sup>2</sup> Institute of Structural Mechanics and Lightweight Design, Germany	
<b>Development of a Single Purpose High Temperature Test Setup for the Solar Orbiter Heat Shield</b> <i>P. Jens Hein<sup>1</sup>, D. Lentz<sup>1</sup></i> <sup>1</sup> IABG mbH, Germany	<b>Emergence of Airware 2050 for Pressurized and Structural Components of Space Launchers and Crew Vehicles</b> <i>M. Niedzinski<sup>1</sup></i> <sup>1</sup> Constellium, United States	
<b>Thermal Tests of Solar Orbiter FDMs and Antennas Subsystems</b> <i>B. Provedo<sup>1</sup>, C. Borque<sup>1</sup>, A. Pereda<sup>1</sup></i> <sup>1</sup> SENER, Spain	<b>Novel Space Applications for CNT Composites, an Overview on the Activity NATAP</b> <i>L. Tiedemann<sup>1</sup>, P. Krzikalla<sup>1</sup>, E. K. Pfeiffer<sup>1</sup>, C. Pereira<sup>2</sup>, M. Martins<sup>3</sup>, J. de Wit<sup>4</sup>, S. Forero<sup>5</sup>, T. Vlček<sup>6</sup>, V. Liedtke<sup>7</sup>, A. Makaya<sup>8</sup></i> <sup>1</sup> HPS GmbH, Germany, <sup>2</sup> HPS Lda., Portugal, <sup>3</sup> INEGI, Portugal, <sup>4</sup> INVENT GmbH, Germany, <sup>5</sup> FutureCarbon GmbH, Germany, <sup>6</sup> TOSEDA s.r.o., Czech Republic, <sup>7</sup> Aerospace & Advanced Composites GmbH, Austria, <sup>8</sup> ESA-ESTEC, The Netherlands	
<b>A Successful TAS-I Strategy for Satellites Thermal Vacuum Test Management</b> <i>G. Bitetti<sup>1</sup>, P. Miccichè<sup>1</sup>, A. Ciriello<sup>1</sup></i> <sup>1</sup> Thales Alenia Space - CCPI-I AIT Center, Italy	<b>Manufacture and Test of C/C-SiC Sandwich Structures</b> <i>B. Heidenreich<sup>1</sup>, N. Bamsey<sup>2</sup>, Y. Shi<sup>1</sup>, D. Koch<sup>1</sup></i> <sup>1</sup> Deutsches Zentrum für Luft- und Raumfahrt e.V., German Aerospace Center, Institute of Structures and Design, Germany, <sup>2</sup> ESA-ESTEC, Materials Technology Section TEC-MSP, The Netherlands	
<b>LUNCH</b>		
EINSTEIN	ERASMUS	ESCAPE
<b>Thermal Vacuum Test Facilities</b>	<b>Advanced Materials - 2</b>	
<b>Chair: E. Bonnet - S. Roose</b>	<b>Chair: C. Edtmaier - C. Semprimoschnig</b>	
<b>In Situ Measurement of Sun Simulator Mirror Orientations and Validation of the Ray-Tracking Model</b> <i>S. Sablerolle<sup>1</sup>, R. Vink<sup>1</sup>, A. Filatov<sup>2</sup>, I. Nagorski<sup>2</sup>, M. Appolloni<sup>1</sup></i> <sup>1</sup> ESA-ESTEC, The Netherlands, <sup>2</sup> USHIO Europe BV, Netherlands	<b>Demonstration Cases of Spacecraft Elements with Nano-Enabled CFRPs</b> <i>A. Vavouliotis<sup>1,2</sup>, A. Baltopoulos<sup>1,2</sup>, V. Kostopoulos<sup>2</sup>, L. Pambaguian<sup>3</sup></i> <sup>1</sup> Adamant Composites Ltd, Greece, <sup>2</sup> Applied Mechanics Laboratory, Greece, <sup>3</sup> European Space Research and Technology Centre (ESA-ESTEC), The Netherlands	

14:30 - 15:00	<b>LAUMBS: a New Software for Launch Vehicle Design and Verification During Ascent and Payload Injection</b> <i>V. Rossi<sup>1</sup>, A. Wiegand<sup>1</sup>, S. Shäff<sup>1</sup>, M. Toso<sup>2</sup></i> <sup>1</sup> Astos Solution, Germany, <sup>2</sup> ESA-ESTEC, The Netherlands	<b>Development of a Semi-Automatic, Mobile Insert Pull-Test Machine</b> <i>J. Ess<sup>1</sup>, H. P. Gröbelbauer<sup>2</sup>, R. Usinger<sup>3</sup>, D. Nägel<sup>3</sup></i> <sup>1</sup> Infor AG, Switzerland, <sup>2</sup> University of Applied Sciences Northwestern Switzerland, Switzerland, <sup>3</sup> RUAG Space, Switzerland
15:00 - 15:30	<b>Determination of Critical Mechanical Load Cases: A Nonlinear Programming Approach</b> <i>K. Bodjona<sup>1</sup>, S. Wong<sup>1</sup></i> <sup>1</sup> Airbus Defence And Space Netherlands, The Netherlands	<b>Industrialisation of the APM Technology: OneWeb but not Only</b> <i>A. Di Carlo<sup>1</sup>, A. Isele<sup>1</sup>, Mr Dominik Naegeli<sup>1</sup></i> <sup>1</sup> Ruag Space, Switzerland
15:30 - 16:00	<b>SpaceBolt™: Fastener Verification Solution for the Space Industry</b> <i>N. Asmolovskiy<sup>1</sup>, F. Ruess<sup>1</sup>, B. Braun<sup>1</sup>, G. Campoli<sup>2</sup>, M. Such Taboada<sup>2</sup></i> <sup>1</sup> Space Structures GmbH, Germany, <sup>2</sup> ESA-ESTEC, The Netherlands	<b>Accelerated Fatigue Testing of Potted Inserts</b> <i>S. Kögl<sup>1</sup>, H.-P. Gröbelbauer<sup>2</sup>, N. Gradwohl<sup>2</sup>, R. Usinger<sup>3</sup>, Dominik Nägel<sup>3</sup></i> <sup>1</sup> KOEGL Space, Switzerland, <sup>2</sup> University of Applied Sciences Northwestern Switzerland, Switzerland, <sup>3</sup> RUAG Space, Switzerland
16:00 - 16:30	<b>BREAK</b>	
	<b>NEWTON 1</b>	<b>NEWTON 2</b>
16:30 - 18:30	<b>Load Derivation</b>	<b>Special Analyses</b>
	Chair: W. Konrad - B. Tang	Chair: P. Nali - M. Toso
16:30 - 17:00	<b>Prediction of the Mechanical Environments of the Load Critical Elements of the ATHENA Spacecraft</b> <i>S. Fransen<sup>1</sup>, C. Sanchez Herrera Cabanas<sup>1</sup>, M. Ayre<sup>1</sup>, I. Ferreira<sup>1</sup>, M. Bavadz<sup>1</sup>, E. Wille<sup>1</sup>, A. Stefanescu<sup>1</sup></i> <sup>1</sup> ESA-ESTEC, The Netherlands	<b>Towards Real-Time Highly Flexible Multibody System Simulations</b> <i>W. Long, P. Tiso</i> <sup>1</sup> ETH Zürich, Switzerland
17:00 - 17:30	<b>Local Refinement of Classical FEM Solutions Using Elements with Node-Dependent Kinematics</b> <i>E. Carrera<sup>1</sup>, M. Petrolo<sup>1</sup>, A. Pagani<sup>1</sup>, E. Zappino<sup>1</sup>, A. G. Fiordilino<sup>1</sup>, M. H. Nagaraj<sup>1</sup>, I. Kaleel<sup>1</sup></i> <sup>1</sup> Politecnico Di Torino, Italy	<b>Assessment of a Multi-Payload Insertion Problem by Means of MultiBody Dynamics</b> <i>M. Toso<sup>1</sup>, J. Demming<sup>2</sup>, V. Rossi<sup>3</sup></i> <sup>1</sup> ESA-ESTEC (ATG), The Netherlands, <sup>2</sup> ESA-ESTEC, The Netherlands, <sup>3</sup> ASTOS Solution, Germany

<p><b>Measurement and Simulation of Collimation Angle Phenomenon in LSS Sun Simulator</b></p> <p><i>A. Filatov<sup>1</sup>, S. Sablerolle<sup>2</sup>, R. Vink<sup>2</sup>, I. Nagorski<sup>1</sup>, M. Appolloni<sup>2</sup>, A. Cozzani<sup>2</sup></i></p> <p><sup>1</sup>Ushio Europe BV, The Netherlands, <sup>2</sup>ESA-ESTEC, The Netherlands</p>	<p><b>Design &amp; Development of the ExoMars Rover Bioseals</b></p> <p><i>P. Alvarez Fernandez, J. Jordan, D. Christou</i></p>	
<p><b>VTC1.5 Upgrade – Improvement of Spatial and Temporal Stability of High Flux Sun Simulator</b></p> <p><i>A. Filatov<sup>1</sup>, M. Appolloni<sup>2</sup>, I. Nagorski<sup>1</sup>, S. Sablerolle<sup>2</sup>, R. Vink<sup>2</sup></i></p> <p><sup>1</sup>Ushio Europe BV, The Netherlands, <sup>2</sup>ESA-ESTEC, The Netherlands</p>	<p><b>Evaluation of Lighter and More Efficient Radiation Protection Materials</b></p> <p><i>L. Tiedemann<sup>1</sup>, P. Krzikalla<sup>1</sup>, M. Leininger<sup>2</sup>, M. Steffens<sup>3</sup>, A. Tighe<sup>4</sup></i></p> <p><sup>1</sup>HPS GmbH, Germany, <sup>2</sup>OHB System AG, Germany, <sup>3</sup>Fraunhofer INT, Germany, <sup>4</sup>ESA-ESTEC, The Netherlands</p>	
<p><b>Modification of a High Temperature Industrial Furnace for Integration to a Space Simulation Test Facility</b></p> <p><i>D. Lenz<sup>1</sup>, P. Jens Hein<sup>1</sup></i></p> <p><sup>1</sup>IABG mbH, Germany</p>	<p><b>Friction Stir Welding of Fibre Reinforced Titanium Composites for Aerospace Structures</b></p> <p><i>N. Iqbal<sup>1</sup>, C. Blacker<sup>2</sup>, J. Martin<sup>1</sup>, K. Beamish<sup>3</sup>, A. Makaya<sup>4</sup></i></p> <p><sup>1</sup>Twi Ltd, United Kingdom, <sup>2</sup>TISICS Ltd, United Kingdom, <sup>4</sup>ESA-ESTEC, The Netherlands</p>	
<b>BREAK</b>		
<p><b>EINSTEIN</b></p>	<p><b>ERASMUS</b></p>	<p><b>ESCAPE</b></p>
<p><b>Thermal Protection Systems &amp; Parachute Testing</b></p>	<p><b>Joining Technology</b></p>	
<p>Chair: H. Ritter - L. Ferracina</p>	<p>Chair: L. Cornillon - S. Das</p>	
<p><b>ExoMars 2020 Mission Pilot Parachutes Dynamic Extraction Test: Overview on Test Development and Implementation</b></p> <p><i>A. Balossino<sup>1</sup>, L. Marconi<sup>1</sup>, F. Miglioretti<sup>1</sup>, V. Tamborra<sup>2</sup></i></p> <p><sup>1</sup>Arescosmo SpA, Italy, <sup>2</sup>Arescosmo SpA, Italy</p>	<p><b>ArianeGroup R&amp;T Activity on Composite Joining Technology</b></p> <p><i>M. Leroy<sup>1</sup></i></p> <p><sup>1</sup>ArianeGroup, France</p>	
<p><b>Mathematical Model of a Pneumatic Mortar for Parachute Deployment</b></p> <p><i>A. Balossino<sup>1</sup>, L. Marconi<sup>1</sup>, F. Miglioretti<sup>1</sup>, V. Tamborra<sup>2</sup></i></p> <p><sup>1</sup>Arescosmo SpA, Italy, <sup>2</sup>Arescosmo SpA, Italy</p>	<p><b>Glass to Metal Contact Optimization in an Optical Space Instrument Assembly</b></p> <p><i>C. Flores Diaz<sup>1</sup>, J. García Martínez<sup>2</sup>, J. Miguel Encinas Plaza<sup>1</sup>, J. Cabrero Gómez<sup>2</sup>, M. Colombo Bueno<sup>1</sup>, D. Escribano Lahera<sup>1</sup>, P. Gallego Sempere<sup>1</sup>, M<sup>a</sup> del Rosario Canchal Moreno<sup>1</sup>, M. Fernández Rodríguez<sup>1</sup></i></p> <p><sup>1</sup>Instituto Nacional de Técnica Aeroespacial (INTA), Spain, <sup>2</sup>ISDEF, Spain</p>	

17:30 - 18:00	<b>Consistent Dimensioning Interface Loads – An Alternative and Less Conservative Approach to Define Loads for Sub-Components</b> <i>R. Meitzner<sup>1</sup>, V. Ramanolla<sup>1</sup>, H. Kellermeier<sup>1</sup></i> <sup>1</sup> ArianeGroup, Germany	<b>The Cossmas Project, a Step Forward Towards the Digital Twin of Composite Launchers Structures</b> <i>F. Cugnon<sup>1</sup>, L. Ballere<sup>2</sup>, C. Martin<sup>2</sup>, C. Lequesne<sup>1</sup></i> <sup>1</sup> Samtech S.a., Belgium, <sup>2</sup> Ariane group, France
18:00 - 18:30	<b>Diversity of Random Vibration Load Definition from Early Phase to Testing Using the Example of Sentinel-2 S/C</b> <i>A. Kommer<sup>1</sup>, W. Konrad<sup>1</sup>, A. Karl<sup>1</sup></i> <sup>1</sup> Airbus Defence And Space GmbH, Germany	
<b>18:30 - 20:30</b>	<b>POSTER SESSION - 2</b>	

<p><b>Technology for Super Light-Weight Thermal Protection Systems for Space Applications</b></p> <p><i>M. Parco<sup>1</sup>, I. F Fagoaga<sup>1</sup>, I. Belan<sup>2</sup>, I. Neshpor<sup>2</sup>, G. Frolov<sup>2</sup>, I. Gusarova<sup>3</sup>, A. Potapov<sup>3</sup>, I. Derevianko<sup>3</sup>, I. Falchenko<sup>4</sup>, V. Yatsenko<sup>5</sup>, L. Silvestroni<sup>6</sup>, M. Kütemeyer<sup>7</sup>, T. Reimer<sup>7</sup></i></p> <p><sup>1</sup>Tecnalia, Spain, <sup>2</sup>Frantsevich Institute for Problems of Materials Science NAS of Ukraine, Ukraine, <sup>3</sup>Yuzhnoye State Design Office, Ukraine, <sup>4</sup>E.O.Paton Electric Welding Institute, Ukraine, <sup>5</sup>Space Research Institute of NASU-SSAU, Ukraine, <sup>6</sup>ISTEC/ CONSIGLIO NAZIONALE DELLE RICERCHE, Italy, <sup>7</sup>Institute of Structures and Design, DLR, Germany</p>	<p><b>Adhesive Free High Stability Optical Mount for Space Laser Applications: Design Optimization For Different CTE Materials Coupling</b></p> <p><i>P. Mosciarelli<sup>1</sup>, E. Di Carmine<sup>1</sup></i></p> <p><sup>1</sup>Leonardo, Italy</p>	
<p><b>Preliminary Analysis on Carbon-Phenolic Thermal Behavior Subjected to an Impinging Flame</b></p> <p><i>G. Dugast<sup>1</sup>, P. Tadini<sup>3</sup>, K. Chetehouna<sup>3</sup>, N. Gascoin<sup>3</sup>, M. Bouchez<sup>2</sup>, J.-L. Marceau<sup>2</sup>, R. Peiffer<sup>2</sup></i></p> <p>1MBDA France / Laboratoire PRISME, France, Bourges, France, 3Laboratoire Prisme, INSA CVL, Bourges, France</p>	<p><b>Development of Stable Sandwich Material Structures for Space Applications</b></p> <p><i>L. Cornillon<sup>1</sup>, O. Damiano<sup>1</sup>, S. Behar-Lafenêtre<sup>1</sup>, G. Brûche<sup>1</sup>, F.-R. Gourillon<sup>1</sup>, L. Chichignoud<sup>2</sup>, M. Ferraris<sup>3</sup>, S. De La Pierre<sup>3</sup>, V. Michaud<sup>4</sup>, M. Piccand<sup>4</sup>, J. Caron<sup>4</sup>, S. Peeterbroeck<sup>5</sup></i></p> <p><sup>1</sup>Thales Alenia Space, France, <sup>2</sup>NTPT, Switzerland, <sup>3</sup>Politecnico di Torino, Italy, <sup>4</sup>EPFL, Switzerland, <sup>5</sup>Materianova, Belgium</p>	

**POSTER SESSION - 2**

# Day 5 - Friday 1 June

	NEWTON 1	NEWTON 2
<b>08:30 - 10:30</b>	<b>Non-Linear Behavior - 1</b>	Buckling - 1
	Chair: M. Jacquesson - M. Ellenbroek	Chair: J. Wijker - E. Jansen
08:30 - 09:00	<b>Observation of Nonlinear Structural Responses in Vibration Testing</b> <i>A. R. Kolaini</i>	<b>Buckling Test of Composite Cylindrical Shells with Oval Imperfection Under Axial Compression</b> <i>A. Takano<sup>1</sup></i> <sup>1</sup> Kanagawa University, Japan
09:00 - 09:30	<b>Sentinel-1C&amp;D: Antenna Separation Mechanism Non-Linear Dynamic Behaviour</b> <i>G. Laduree<sup>1</sup>, D. Bibby<sup>1</sup>, M. Hofmann<sup>1</sup>, M. v. Alberti<sup>2</sup>, C. Lausch<sup>2</sup>, S. Diel<sup>2</sup>, R. Baldassarri<sup>3</sup></i> <sup>1</sup> ESA-ESTEC, The Netherlands, <sup>2</sup> Airbus DS GmbH, Germany, <sup>3</sup> Thales Alenia Space, Italy	<b>Experimental Investigation of Axially Compressed CFRP Thin-Walled Truncated Cones and Cylinders with Cutouts</b> <i>R. Khakimova<sup>1</sup>, R. Degenhardt<sup>2</sup>, D. Wilckens<sup>2</sup></i> <sup>1</sup> Invent Gmbh, Germany, <sup>2</sup> DLR, Germany
09:30 - 10:00	<b>Testing and Model Updating of Nonlinear Aerospace Structures</b> <i>T. Dosogne<sup>1</sup>, J.-P. Noël<sup>1,2</sup>, T. Detroux<sup>1,2</sup>, G. Kerschen<sup>1,2</sup></i> <sup>1</sup> University of Liège, Belgium, <sup>2</sup> NOLISYS sprl, Belgium	<b>Preliminary Study of the Local Buckling Behaviour of DLR's CFRP Booms Induced by the Stowing Process</b> <i>S. Meyer<sup>1</sup>, M. Hillebrandt<sup>1</sup>, C. Hühne<sup>1</sup></i> <sup>1</sup> DLR (German Aerospace Center), Germany
10:00 - 10:30	<b>Nonlinearities Detection and Identification: An Experimental and Practical Implementation</b> <i>M. Hofmann<sup>1</sup>, G. Ladurée<sup>1</sup>, R. Knockaert<sup>1</sup></i> <sup>1</sup> ESA-ESTEC, The Netherlands	<b>Buckling of Thin Cylindrical Shell Submitted to Local or Harmonic Mechanical Loads</b> <i>T. Benoît<sup>2</sup>, M. Jacquesson<sup>2</sup>, A. Limam<sup>2</sup>, H.V. Tran<sup>1</sup></i> <sup>1</sup> UDL, Université de Lyon, France, <sup>2</sup> CNES, France
<b>10:30 - 11:00</b>	<b>BREAK</b>	
	NEWTON 1	NEWTON 2
<b>11:00 -13:30</b>	<b>Non linear behaviour - 2</b>	<b>Verification approaches</b>
	Chair: A. Kolaini - G. Laduree	Chair: I. Ngan - C. Puillet
11:00 - 11:30	<b>Simulation of Non-Linear Damping Devices for Payload Comfort</b> <i>J. Marchesini<sup>1</sup>, L. Dastugue<sup>1</sup>, M. Lequoy<sup>1</sup>, R. Helfrich<sup>2</sup></i> <sup>1</sup> Intes France, France, <sup>2</sup> INTES GmbH, Germany	<b>Approach to Optimizing Environmental Test Condition Based on Modelling of Test Level and Loss Cost</b> <i>S. Shimazaki<sup>1</sup>, D. Takahashi<sup>1</sup>, Q. Shi<sup>1</sup></i> <sup>1</sup> Japan Aerospace Exploration Agency (JAXA), Japan

EINSTEIN	ERASMUS	ESCAPE
<b>Vibration Test Simulation &amp; Modelling</b>	<b>Flexures</b>	
Chair: G. Aglietti - M. Appolloni	Chair: G. Rodrigues - M. Richter	
<b>Virtual Shaker Testing: Actual Achievements in TAS and Future Prospects</b> <i>P. Nalì<sup>1</sup>, V. Di Pietro<sup>1</sup>, P. Ladisa<sup>1</sup>, G. Bitetti<sup>1</sup>, F. Lumaca<sup>1</sup>, A. Bettacchioli<sup>2</sup></i> <sup>1</sup> Thales Alenia Space, Italy, <sup>2</sup> Thales Alenia Space, France	<b>Optimization of Composite Tube Flexures</b> <i>M. Santer<sup>1</sup></i> <sup>1</sup> Imperial College London, United Kingdom	
<b>Post-Test Correlation Activity and 6-DOF Transient Test Validation by Means of Virtual Testing Approaches</b> <i>M. Remedja<sup>1</sup>, G. Aglietti<sup>1</sup>, M. Appolloni<sup>2</sup>, A. Cozzani<sup>2</sup>, A. Kiley<sup>3</sup></i> <sup>1</sup> University Of Surrey, United Kingdom, <sup>2</sup> ESA-ESTEC, The Netherlands, <sup>3</sup> Airbus Defence & Space, United Kingdom	<b>Reduction of Moments Induced on the Coiling Hub of a Boom Deployment Mechanism</b> <i>M. Richter<sup>1</sup>, M. Hillebrandt<sup>1</sup>, C. Huehne<sup>1</sup></i> <sup>1</sup> German Aerospace Centre (DLR e.V.), Germany	
<b>Integrated Solution for Virtual Testing</b> <i>F. D'ambrosio<sup>1</sup>, A. Carrella<sup>2</sup>, S. Hoffait<sup>3</sup>, G. Patanchon<sup>4</sup></i> <sup>1</sup> Siemens PLM, Belgium, <sup>2</sup> Siemens PLM, Belgium, <sup>3</sup> V²i, Belgium, <sup>4</sup> Ariane Group, France	<b>Experimental Verification of Novel Analytical Wrinkling Control Mechanism of Planar Membrane Reflector for Space Application</b> <i>S. Kumar<sup>1</sup>, S. H. Upadhyay<sup>1</sup></i> <sup>1</sup> Indian Institute of Technology Roorkee, India, <sup>2</sup> Indian Institute of Technology Roorkee, India	
<b>Experimental Data Driven Approach for Numerical Spacecraft Vibration Test Prediction</b> <i>S. Waimer<sup>1</sup>, B. Peeters<sup>1</sup>, M. Wagner<sup>2</sup>, P. Guillaume<sup>3</sup></i> <sup>1</sup> Siemens Industry Software NV, Belgium, <sup>2</sup> ESA-ESTEC, The Netherlands, <sup>3</sup> Vrije Universiteit Brussel, Belgium	<b>Dynamics Analysis for Spacecraft with a Tendon-Driven Continuum Manipulator</b> <i>H. Yao<sup>1</sup>, Z. Hu<sup>1</sup>, G. Wang<sup>2</sup>, M. Si<sup>1</sup>, Y. Zhang<sup>1</sup>, G. Zheng<sup>1</sup></i> <sup>1</sup> Tsinghua University, China, <sup>2</sup> Beijing Institute of Spacecraft System Engineering, China	
<b>BREAK</b>		
EINSTEIN	ERASMUS	ESCAPE
<b>Buckling - 2</b>		
Chair: P. Mourey - U. Block		
<b>Numerical Assessment of Existing Vibration Correlation Techniques</b> <i>F. Franzoni<sup>1</sup>, M. Arbelo<sup>2</sup>, R. Degenhardt<sup>1,3</sup></i> <sup>1</sup> DLR, Institute of Composite Structures and Adaptive Systems, Germany, <sup>2</sup> ITA, Aeronautics Institute of Technology, Brazil, <sup>3</sup> PFH, Private University of Applied Sciences Göttingen, Germany		

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11:30 - 12:00	<b>Identification and Simulation of Non-Linear Vibration Test</b> <i>A. Bettacchioli<sup>1</sup></i> <i><sup>1</sup>Thales Alenia Space, France</i>	<b>Influence of Materials and Processes on Mechanical and Design Margins of Telecom Satellite by Use of Stochastic Method and Software</b> <i>G. Pommatau<sup>1</sup>, S. Das<sup>2</sup>, J. D'add<sup>1</sup>, J. Dudon<sup>1</sup>, J. Buffe<sup>1</sup></i> <i><sup>1</sup>Thales Alenia Space, France, <sup>2</sup>ESA-ESTEC, The Netherlands</i>
12:00 - 12:30		
12:30 - 13:30	<b>CLOSING SESSION</b>	
13:30 - 14:30	<b>LUNCH</b>	
14:30 - 15:30	<b>TOURS</b> - Materials Lab - Testing	

<b>Mechanical Response of Variable and Constant Stiffness Cylindrical Shells of Launcher Structures</b> <i>E. Labans<sup>1</sup>, C. Bisagni<sup>1</sup></i> <sup>1</sup> Delft University of Technology, The Netherlands		
<b>Effect of Foam Thermal Insulation Layer on the Buckling of Thin-Walled Cylinders Under Axial Compression, Bending or Shear Load: Space Launchers Application</b> <i>M. Jacquesson<sup>2</sup>, A. Limam<sup>1</sup>, F. Lorioux<sup>3</sup>, F. Marteau<sup>2</sup>, H. Viet Tran<sup>1</sup></i> <sup>1</sup> University of Lyon, France, <sup>2</sup> CNES, France, <sup>3</sup> Airbus, France		
<b>LUNCH</b>		
TOURS - Materials Lab - Testing		

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# Poster presentations

## P1 Angular Vibration Test System and Its Application in Precise Pointing Mechanism Anti Base Disturbance Test

L. Tan<sup>1</sup>, G. Wang<sup>1</sup>, J. Luo<sup>1</sup>, Q. Meng<sup>1</sup>

<sup>1</sup>Beijing Institute of Space System Engineering, China

## P2 Sensors Digital Identification and Configuration Management

F. Canourgues<sup>2</sup>, P. Dupuis<sup>1</sup>, F. Haddad<sup>1</sup>, L. Perrin<sup>2</sup>

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## P3 Thermal Deformation Testing of Ultra-Stable Structures Down to Cryogenic Temperatures

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## P4 STM Development of Gregorian Type Top Floor Antenna Structure

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## P5 HUB Dispenser: Small-Sat Affordable Access to Space

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## P6 Influence of Materials and Processes on Mechanical and Design Margins of Telecom Satellite by Use of Stochastic Method and Software

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## P7 Study on Influence of Space Environment on Optical Solar Reflectors

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## P8 Environmental Test facilities of the DLR Institute of Space Systems

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## P9 Static and Dynamic Properties Research of a Deployable Space Mast for Membrane Solar Array

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## P10 Irradiation Facilities at DLR-Bremen

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**P11 Comparison Study of Different Methods of Accelerometer Calibration**

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**P12 Flattening Process Simulation and Parameter Study of Triangular Rollable and Collapsible (TRAC) Booms**

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**P13 New Large Slip Table and QUAD Expander**

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**P14 Accelerating the Industrialization of Additive Manufacturing through Process Simulation an Integrated End-to-End Process**

*O. Fergani*

**P15 Static and Dynamic Characterictics Of Spacecraft With Central Cylinders**

*M. Sahin<sup>1</sup>*

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**P16 Pyroshock Tests of Payload Repeater Equipment**

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